

The background of the entire slide is a space-themed illustration. On the left, a large, detailed Earth is shown in the foreground, with a smaller, reddish planet (Mars) visible behind it. A rocket is depicted in the distance, moving from left to right and leaving a bright blue trail of light. The sky is a deep, dark blue filled with numerous white stars. In the bottom right corner, the silhouette of a person's head and shoulders is shown in profile, looking towards the left. The bottom edge of the image shows a dark, silhouetted horizon line.

EXPLORESPACE TECH
TECHNOLOGY DRIVES EXPLORATION

**Technology, Innovation, and Engineering Committee Report
NASA Advisory Council Meeting**

Mr. Michael Johns | Committee Chairman | October 1, 2024

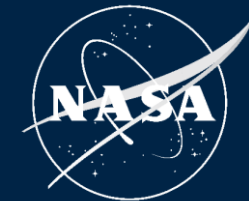
TI&E Committee Hybrid Meeting Attendees: Sept. 5, 2024


- Heshmat Aglan, Tuskegee University
- Mike Gazarik, University of Colorado (virtual)
- Michael Johns, Kratos SRE
- Rebecca Kramer Bottiglio, Yale University (virtual)
- Andrew Rush, Star Catcher
- Mitchell Walker, Georgia Institute of Technology

TI&E Committee Meeting Presentations: Sept. 5, 2024

- Welcome to NASA's Glenn Research Center
 - Wanda Peters, Acting Deputy Director, NASA Glenn
- Space Technology Mission Directorate (STMD) Update
 - Clayton Turner, Acting Associate Administrator, STMD
- 2024 Shortfalls Ranking Process and Results Overview
 - Alesyn Lowry, Director for Strategic Planning and Integration, STMD
 - Michelle Munk, Acting Chief Architect, STMD
- NASA Nuclear Systems Update
 - Anthony Calomino, Space Nuclear Technologies Lead, STMD
 - Lindsay Kaldon, Fission Surface Power Project Manager, NASA Glenn
 - Kurt Polzin, Chief Engineer for NASA's Space Nuclear Propulsion Project, NASA Marshall
- Cryogenic Fluid Management Portfolio Update
 - Lauren Ameen, Deputy Manager, Cryogenic Fluid Management Portfolio Project, NASA Glenn
- Commercial Lunar Payload Services Intuitive Machines-2 Technology Demonstrations Overview
 - Mark Thornblom, Deputy Program Manager, Technical Integration Game Changing Development (GCD) program, NASA Langley
- Early Career Initiative presentation on Mitigating Arc Inception via Transformational Array Instrumentation (MAI TAI)
 - Meghan Bush, Principal Investigator, MAI TAI, NASA Glenn

National Aeronautics and Space Administration




Glenn Research Center
Lewis Field



NASA Glenn

For the Benefit of All

Dr. Wanda Peters

Acting Deputy Center Director

National Aeronautics and
Space Administration



NASA Space Technology Update

NAC TI&E Meeting

Clayton Turner

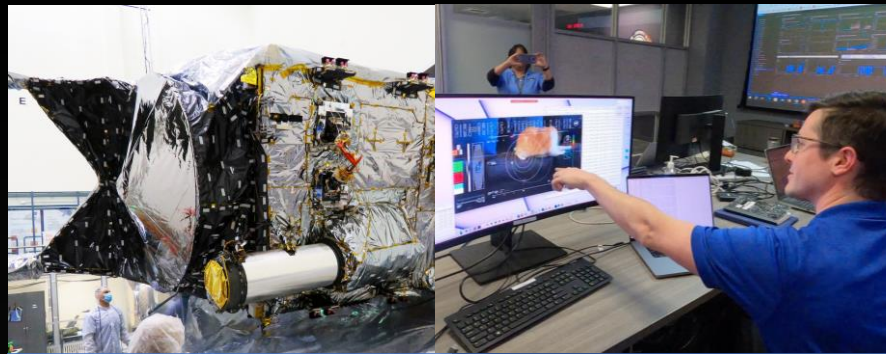
Associate Administrator (Acting),
NASA Space Technology Mission Directorate

September 5, 2024

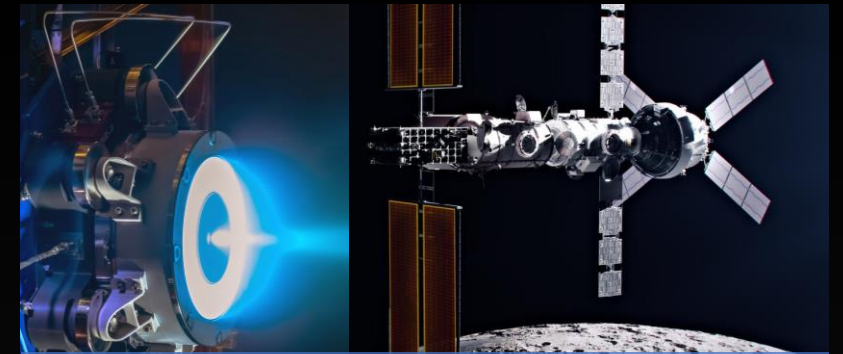
NASA's Space Technology Mission Directorate (STMD)

www.nasa.gov

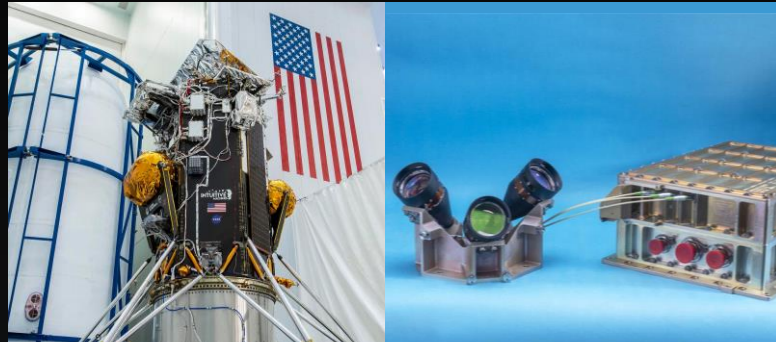
Recent Tech Highlights



Deep Space Optical Communications



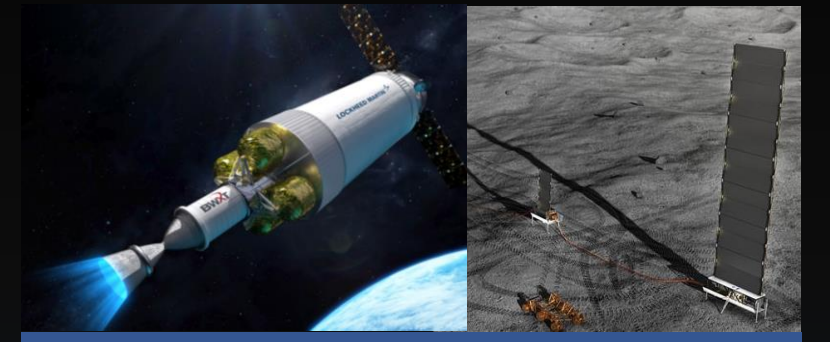
Solar Electric Propulsion



Intuitive Machines 1 CLPS Payloads



Cryogenic Fluid Management Demo



Space Nuclear Technologies



Starling Swarm Demo



Cooperative Robotic Scouts



Solar Sail Deployment



Flight Opportunities

National Aeronautics and
Space Administration



Civil Space Shortfall Ranking NAC TI&E

2024 Feedback Results

NASA's Space Technology Mission Directorate

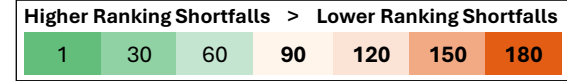
September 5th, 2024



Weighting Approach

- The integrated list utilized weights that reflect STMD's primary customers and their demand for future capabilities as well as other stakeholders' roles in partnering to provide solutions for such capabilities
- NASA inputs received $\frac{2}{3}$ of the overall weight and external scores received the remaining $\frac{1}{3}$
- STMD considered the following tenets of its investment strategy, in priority order, and developed the percentage weight for each stakeholder group
 1. Align with the Administration and the NASA Administrator's priorities that address the Blueprint Objectives (Artemis)
 2. Focus on investments that support science priorities identified in the Decadal Surveys
 3. Foster creation and growth of the space economy through partnering with industry and supporting small business innovation
 4. Engage NASA's workforce to deliver innovative solutions to the nation's toughest technology challenges
 5. Encourage transformative, cross-cutting technologies that benefit NASA as well as other government agencies
 6. Empower a broad community of innovators and academia through emphasis on early-stage investments

Integrated Top 30 Shortfalls Compared to Stakeholder Group Rank



Not Ranked (NR)

Integrated Rank	Shortfall ID	Category
1	1618: Survive and operate through the lunar night	Thermal Management Systems
2	1596: High Power Energy Generation on Moon and Mars Surfaces	Power
3	1554: High Performance Onboard Computing to Enable Increasingly Complex Operations	Avionics
4	1557: Position, Navigation, and Timing (PNT) for In-Orbit and Surface Applications	Communication and Navigation
5	1545: Robotic Actuation, Subsystem Components, and System Architectures for Long-Duration and Extreme Environment Operation	Autonomous Systems and Robotics
6	1552: Extreme Environment Avionics	Avionics
7	1519: Environmental Monitoring for Habitation	Advanced Habitation Systems
8	709: Nuclear Electric Propulsion for Human Exploration	Propulsion: Nuclear
9	1304: Robust, High-Progress-Rate, and Long-Distance Autonomous Surface Mobility	Autonomous Systems & Robotics
10	1520: Fire Safety for Habitation	Advanced Habitation Systems
11	1531: Autonomous Guidance and Navigation for Deep Space Missions	Autonomous Systems & Robotics
12	1591: Power Management Systems for Long Duration Lunar and Martian Missions	Power
13	702: Nuclear Thermal Propulsion for Human Exploration	Propulsion: Nuclear
14	1559: Deep Space Autonomous Navigation	Communication and Navigation
15	1527: Radiation Countermeasures (Crew and Habitat)	Advanced Habitation Systems
16	1526: Radiation Monitoring and Modeling (Crew and Habitat)	Advanced Habitation Systems
17	879: In-space and On-surface, Long-duration Storage of Cryogenic Propellant	Cryogenic Fluid Management
18	1548: Sensing for Autonomous Robotic Operations in Challenging Environmental Conditions	Autonomous Systems & Robotics
19	1558: High-Rate Communications Across The Lunar Surface	Communication and Navigation
20	1626: Advanced Sensor Components: Imaging	Sensors and Instruments
21	792: In-space and On-surface Transfer of Cryogenic Fluids	Cryogenic Fluid Management
22	1569: High-Mass Mars Entry and Descent Systems	Entry Descent and Landing
23	1525: Food and Nutrition for Mars and Sustained Lunar	Advanced Habitation Systems
24	1571: Navigation Sensors for Precision Landing	Entry Descent and Landing
25	1573: Terrain Mapping Capabilities for Precision Landing and Hazard Avoidance	Entry Descent and Landing
26	1562: Advanced Algorithms and Computing for Precision Landing	Entry Descent and Landing
27	1597: Power for Non-Solar-Illuminated Small Systems	Power
28	1568: Entry Modeling and Simulation for EDL Missions	Entry Descent and Landing
29	1516: Water and Dormancy Management for Habitation	Advanced Habitation Systems
30	1524: Crew Medical Care for Mars and Sustained Lunar	Advanced Habitation Systems

Stakeholder Group Rank								
Academia	Small Industry	Large Industry	OGA	Other	NASA Centers	ESDMD	SMD	Other MDs
4	2	2	2	9	6	4	9	1
13	1	1	40	20	4	21	NR	16
80	28	21	27	13	3	34	1	56
9	11	15	29	67	10	28	NR	3
34	27	28	63	10	40	13	9	49
176	49	6	38	23	54	6	9	62
20	101	72	75	61	49	17	19	13
43	131	23	4	52	32	7	NR	7
27	42	30	121	91	34	25	25	66
23	24	78	12	12	12	29	55	14
47	67	24	3	89	42	64	23	15
40	12	10	52	24	68	35	NR	27
36	114	36	14	78	62	7	NR	11
62	129	27	5	120	38	64	23	10
5	23	22	6	2	5	63	NR	6
6	53	41	81	1	13	27	38	35
21	37	3	95	22	1	59	NR	2
42	17	26	90	16	44	14	26	57
25	73	29	77	162	20	5	NR	51
18	75	12	45	160	22	NR	18	68
17	29	4	51	26	2	62	NR	29
152	156	48	117	5	33	16	NR	12
8	32	116	41	45	30	11	NR	58
14	62	37	23	4	31	45	28	9
30	31	9	12	8	11	45	28	53
54	65	45	23	3	25	45	28	8
85	26	5	39	125	47	93	12	20
101	115	76	60	15	50	45	5	45
49	98	127	158	53	69	26	51	22
12	64	94	1	11	21	58	NR	17

ESDMD and SMD provided ranked lists (numbers shown above) in addition to shortfall scores (used for integrated list). ESDMD and SMD did not score all shortfalls. Unscored shortfalls were also not ranked.

National Aeronautics and
Space Administration



Nuclear Electric Propulsion



Space Technology Mission Directorate
Space Nuclear Propulsion
Dr. Kurt Polzin, SNP Chief Engineer | 9/5/2024



Challenges for Nuclear Electric Propulsion (NEP)

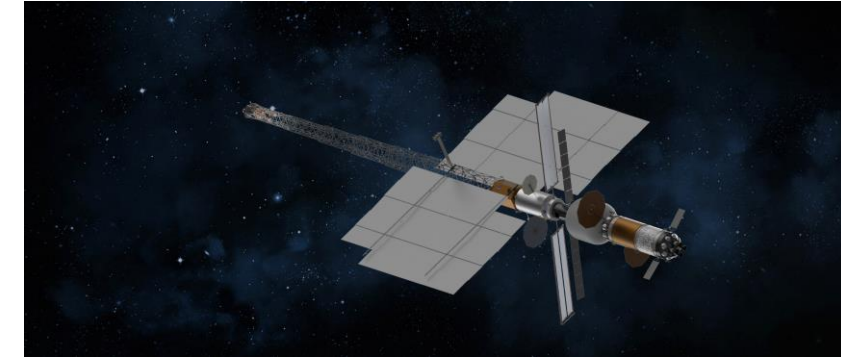
The need for coordinated, realistic technology maturation



For high-power NEP

NASA Engineering & Safety Center Findings¹

- The majority of critical technologies for NEP systems are relatively immature.
- TRLs in the literature are often overestimated and a proper baseline assessment to gauge resources required for advancement has been a consistent issue.
- The majority of critical technologies are at a relatively high level of advancement degree of difficulty ($AD2 > 4$) for maturation, strongly suggesting use of a dual development approach.
- Non-advocate reviews should occur at the start of a technology program and at all key milestones.



National Academies of Science, Engineering, and Medicine Findings²

- For NEP systems, the fundamental challenge is to scale up the operating power of each NEP subsystem and to develop an integrated NEP system suitable for the baseline mission. This requires, for example, scaling power and thermal management systems to power levels orders of magnitude higher than have been achieved to date.
- There has been, low, intermittent, and unfocused investments over the past several decades
- Regarding Hall thrusters and scaling to 100 kW_e thrusters: “... although ground testing of high-power Hall thrusters has revealed that interactions between the test facility, the thruster, and its conducting plasma plume can impact the performance and lifetime measurements in ways that are not fully understood as of this writing.”

¹Independent Assessment of the Technical Maturity of Nuclear Electric Propulsion (NEP) and Nuclear Thermal Propulsion (NTP) Systems, NASA Engineering & Safety Center, 2020

²Space Nuclear Propulsion for Human Mars Exploration, National Academies of Sciences, Engineering, and Medicine, The National Academies Press, Washington, D.C., 2021. DOI: [10.17226/25977](https://doi.org/10.17226/25977)

There are recognized challenges to developing an NEP system



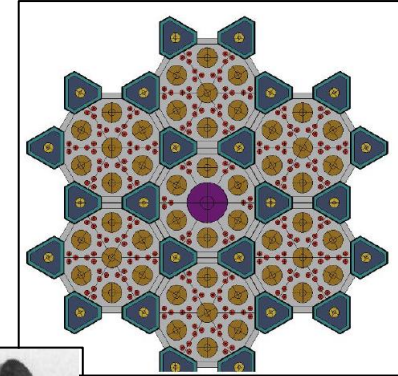
NEP Technology Maturation

Improve Both Low and High-Power Cases



National Academies of Science, Engineering, and Medicine Recommendation²

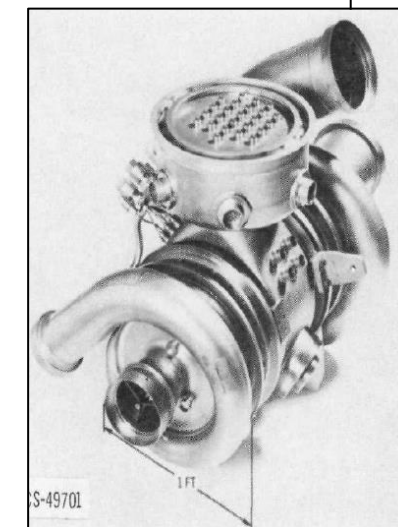
- Subscale in-space flight testing of NEP systems cannot address many of the risks and potential failure modes associated with the baseline mission NEP system. With sufficient M&S [modeling & simulation] and ground testing, **including modular subsystem tests at full scale and power**, flight qualification requirements can be met by the cargo missions that will precede the first crewed mission to Mars. Fully integrated ground testing may not be required.” [emphasis added]



- NASA tech maturation plan developed to guide investments

Key to plan is building and testing hardware **at power and scale**

- High temperature reactor components/heat extraction methods (SNP-funded effort)
 - High temperature Brayton (FSP-funded effort)
 - High temperature-capable electrical generator (SNP-funded effort)
 - Pushing limits on radiator temperatures (Two SNP-funded efforts through SBIR Phase III's)
- Leveraging other investments
 - Electric aircraft (power handling), additive manufacturing (complex structures/heat exchangers), terrestrial microreactors (HALEU-fueled power reactor technology), NTP high-temperature nuclear materials (SNP-funded)



Funding tech maturation activities & leveraging other activities (FSP and others) to make progress through hardware fabrication and demonstration



TI&E Committee Finding on NEP



Short Title of Finding:

Finding:

The Committee has been following the nuclear propulsion portfolio with great interest and has encouraged a balanced portfolio of investments between NTP and NEP. The committee notes the significant progress with NTP and the DRACO partnership and commends STMD for increasing funding of NEP, consistent with NAS recommendations, to address the Critical Technology Elements necessary for the NEP architecture.



CRYOGENIC FLUID MANAGEMENT PORTFOLIO PROJECT (CFMPP)

Project Update for the NASA Advisory Council (NAC)
Technology, Innovation and Engineering (T&E) Committee

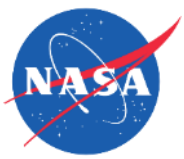
Lauren M. Ameen
Deputy Project Manager
NASA Glenn Research Center

September 5, 2024

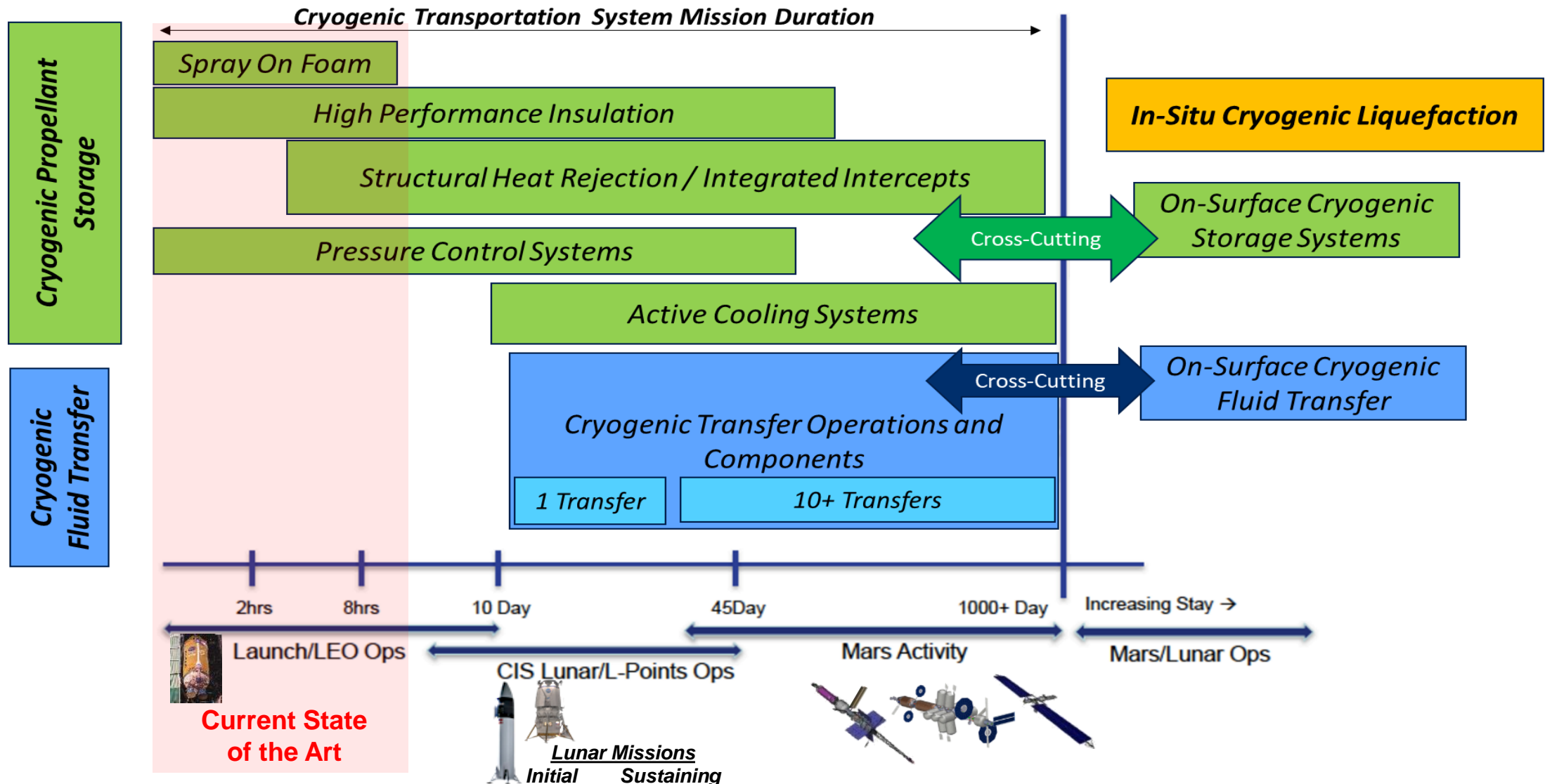


Photo Credit: NASA

- Interest for cryogenic propulsion systems is nearing an all-time high across both the government and industry
- Moon to Mars will push cryogenic mission durations from months to years
- Enabling CFM technologies at TRL ≥ 6 for Mars and Cis-lunar transportation, ISRU, and Surface Systems operations are required for the Moon to Mars Campaign
 - Advanced cryogenic propellant storage and transfer integrated systems, operable in microgravity environments, are required for **all** proposed mission architectures
- **Increasing cryogenic mission duration drives the need for more advanced CFM technology**
 - Current CFM capabilities support missions on the order of days



CFM Technology Capability Needs





Cryogenic Fluid Management Portfolio Project (CFMPP)



System Demonstration Complexity

Technologies Portfolio

Scope: Design, development, testing, and evaluation of critical-need cryogenic components enabling long-duration CFM storage and propellant transfer

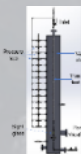
Major Activities:

Hydrogen low-leakage valves and cryo-couplers



Radio Frequency Mass Gauge (RFMG)

Next-generation FOSS (fiber optics sensing system)



Reduced Gravity Cryogenic Transfer (RGCT)

Subsystems Portfolio

Scope: Design, development, and testing of complex systems of technologies to address technical challenges for specific CFM mission needs

Major Activities:

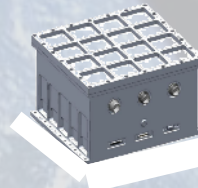


20W/20K Cryocooler

150W/90K Cryocooler

Cryocooler Electronics

Alternate Cryocooler Studies



Images by Creare LLC

Demonstrations Portfolio

Scope: Design, build, and test integrated flight and ground systems comprised of multiple CFM subsystems, enabling TRL 5+ maturation for many technologies

Major Activities:



Tipping Point Demo Contracts

Ground System Demonstrations



Two-Stage Cooling Demonstration

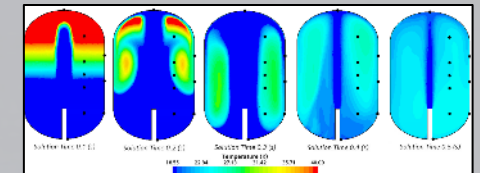
CryoFILL Liquefaction Demonstration



Modeling Portfolio

Scope: Develop, enhance, validate, and demonstrate Computational Fluid Dynamics (CFD) and Nodal tools to address capability gaps for predicting cryogenic fluid behavior in 1-G and microgravity environments for use as design tools for future NASA missions

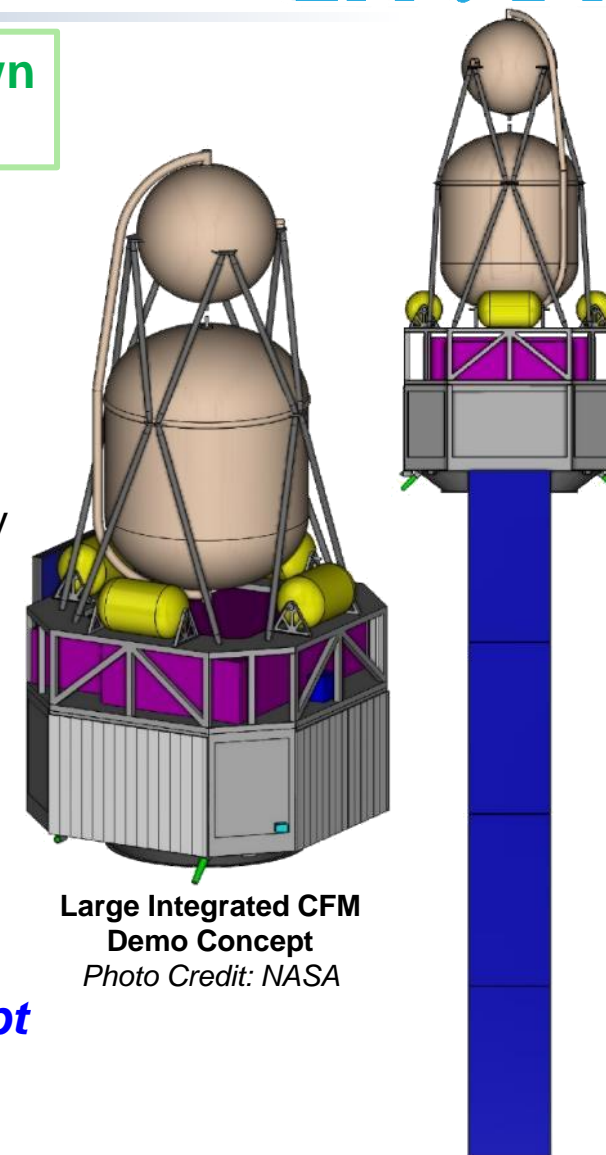
Testing and demo activities across the CFMP portfolio used within modeling tools to predict CFM behavior at a flight vehicle scale in a relevant microgravity environment



System Validation

Integrated system-level microgravity flight demo must be conducted to buy-down technical risk (increase TRL) for future long-duration cryogenic missions

- **CFMPP has been developing the gov't reference concept mission and requirements supporting remaining technology shortfall closure**
 - Mission pull need in the mid-2030s, with intent to build-upon cislunar CFM advancements
- **Demo concept defines appropriate scale and operational duration required to provide sufficient data enabling TRL 6+ maturation**
 - CFMPP has baselined LH₂ cryogen for demonstration → maximized cross-cutting technology gap closure to LO₂/LCH₄ systems
- **Concept Highlights:**
 - Large LH₂ Flight Demonstrator, Class D Mission
 - Mission lasts up to 8 months, providing flight data critical for design model validation
 - Integrated cryogenic demo mission phases will include:
 - Cryogenic active and passive storage operations
 - Cryogenic transfer system operations
- **CFMPP will continue Pre-Formulation of Government reference mission concept scope in FY25 while establishing budget baseline with STMD**



Large Integrated CFM
Demo Concept
Photo Credit: NASA

TI&E Committee Finding on CFM Flight Experiment



Short Title of Finding:

Finding: The Committee recognizes enabling CFM technologies at TRL ≥ 6 for Mars and Cis-lunar transportation, ISRU, and Surface Systems operations are required for the Moon to Mars Campaign. STMD has been advancing the CFM portfolio through tipping points and other STMD programs. The Committee encourages and supports STMD moving forward with the necessary flight demo(s) to mature this capability as soon as budget allows.



Commercial Lunar Payload Services Intuitive Machines-2 Technology Demonstrations Overview

**Technology, Innovation and
Engineering (TI&E) Committee**

Open Session

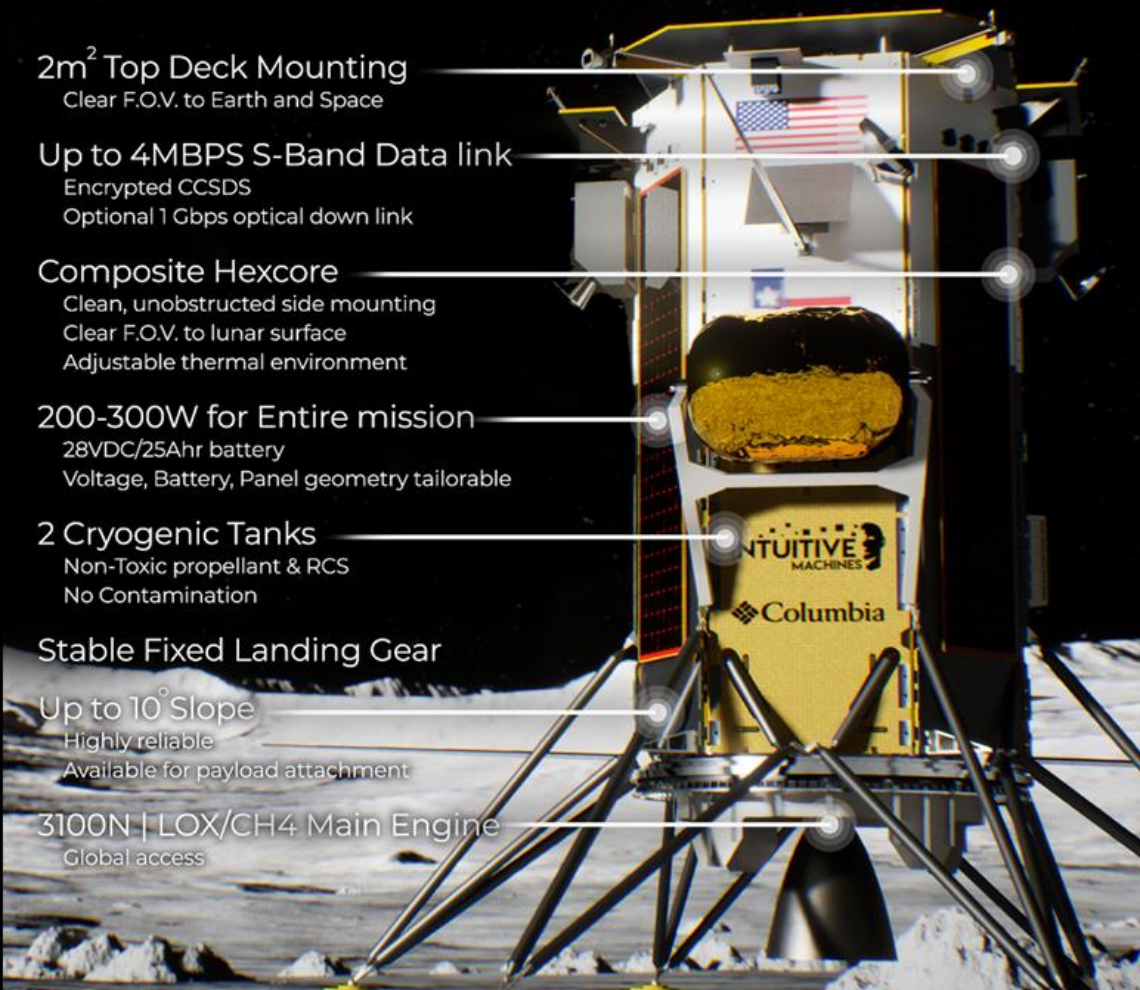
**Mark Thornblom
Game Changing Development
Space Technology Mission Directorate
Deputy Program Manager, Integration**

09.05.24

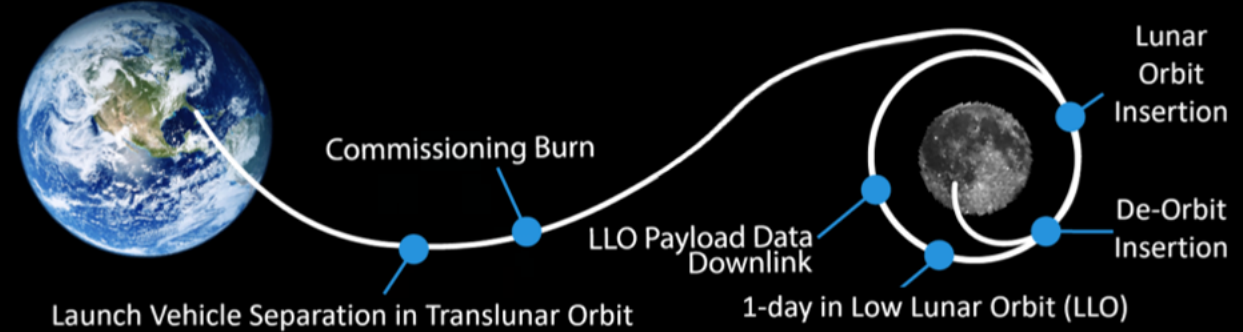
Intuitive Machine NOVA-C Mission Profile



IM NOVA-C Lander



- 2m² Top Deck Mounting**
Clear F.O.V. to Earth and Space
- Up to 4MBPS S-Band Data link**
Encrypted CCSDS
Optional 1 Gbps optical down link
- Composite Hexcore**
Clean, unobstructed side mounting
Clear F.O.V. to lunar surface
Adjustable thermal environment
- 200-300W for Entire mission**
28VDC/25Ahr battery
Voltage, Battery, Panel geometry tailorable
- 2 Cryogenic Tanks**
Non-Toxic propellant & RCS
No Contamination
- Stable Fixed Landing Gear**
- Up to 10° Slope**
Highly reliable
Available for payload attachment
- 3100N | LOX/CH₄ Main Engine**
Global access



NOVA-C Capabilities as a Service

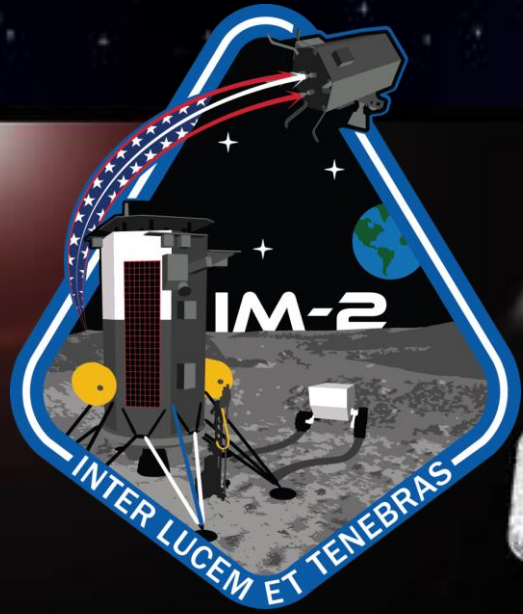
Landing Location	Nova-C	Nova-D	Nova-M*
Equatorial	130-250 kg	500-2500 kg	5,000-10,000 kg
South Pole	130-250 kg	500-2500 kg	5,000-10,000 kg
North Pole	130-250 kg	500-2500 kg	5,000-10,000 kg

	MAPP Rover	μNova Hopper	μNova-LX Hopper
Range (Max Distance from Lander)	2.5 km	25 km	25 km
Payload Mass	5 kg	1 kg	8 kg

Source: Intuitive Machines Lunar Access Services User's Guide- www.intuitivemachines.com



IM-2 NASA Payloads Overview

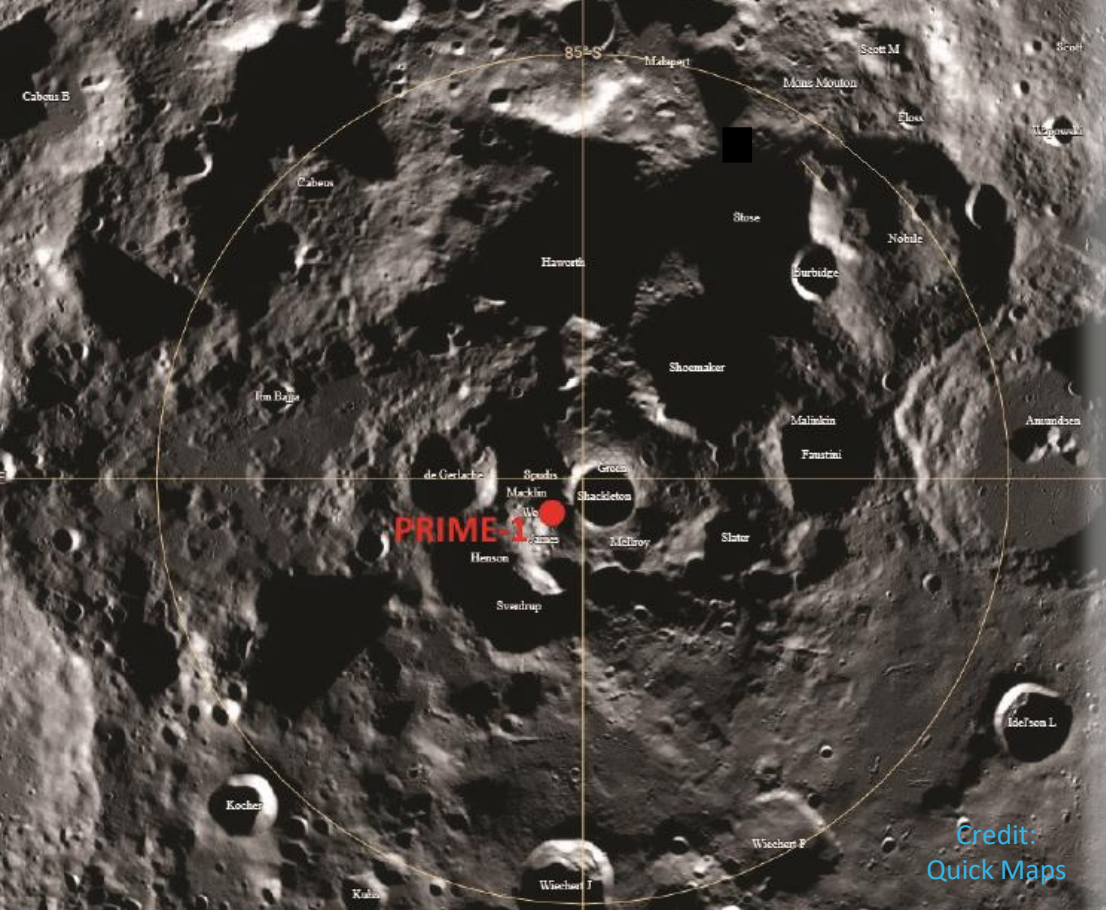


PRIME-1

**NOKIA
BELL
LABS**



IM-2 Mission planned for FY25 Quarter 1

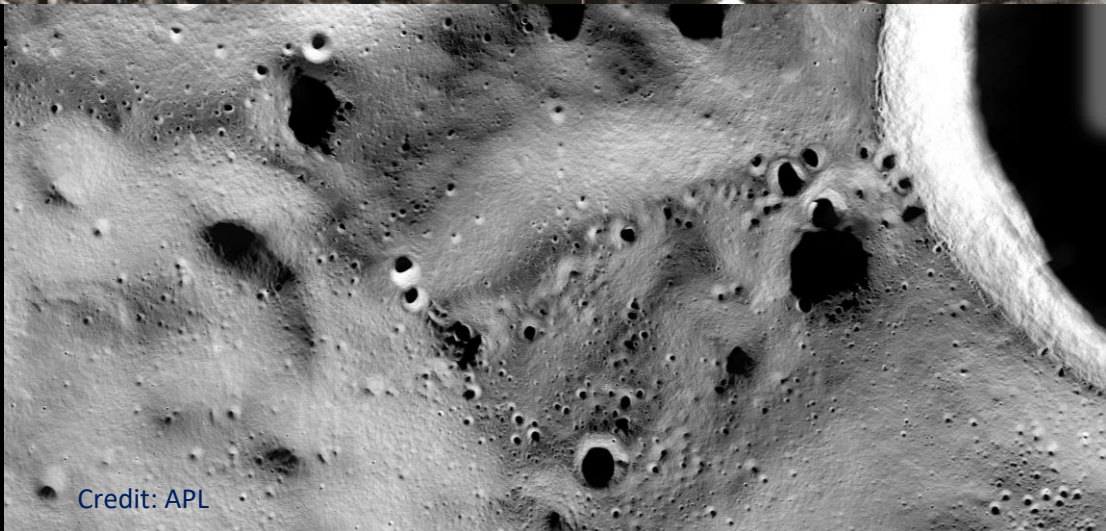


Credit: Quick Maps

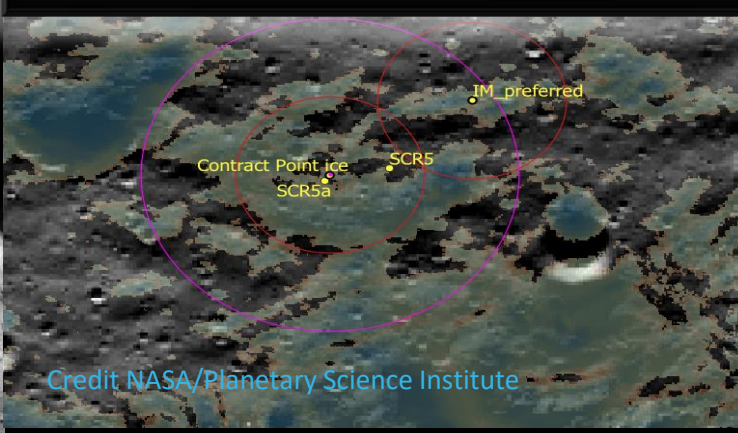


Click image to play video

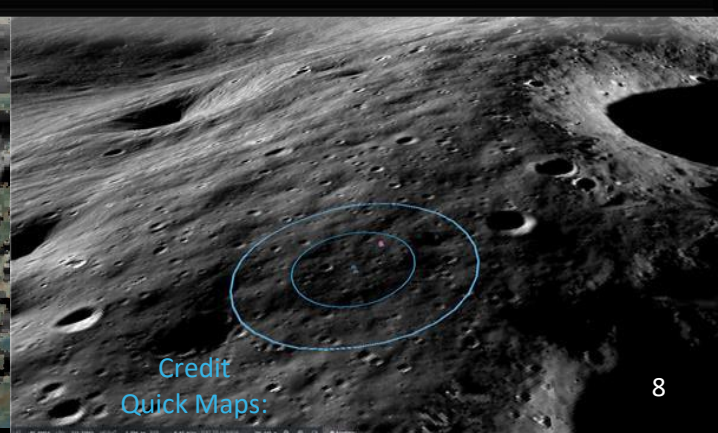
Credit: APL:



Credit: APL



Credit NASA/Planetary Science Institute



Credit Quick Maps:

PRIME-1

PRIME-1 is the combination of two high-TRL instruments; Mass Spectrometer observing lunar operations (MSolo) and The Regolith and Ice Drill for Exploring New Terrain (TRIDENT). PRIME-1 will drill at a south pole lunar landing location where orbiting assets indicate the potential to find water ice. PRIME-1's MSolo instrument will analyze for volatiles present in lunar regolith down across a one-meter vertical profile. Demonstration of the PRIME-1 system will buy down engineering risk to the VIPER mission

<https://twitter.com/NASAMoon/status/1471889928762109953>

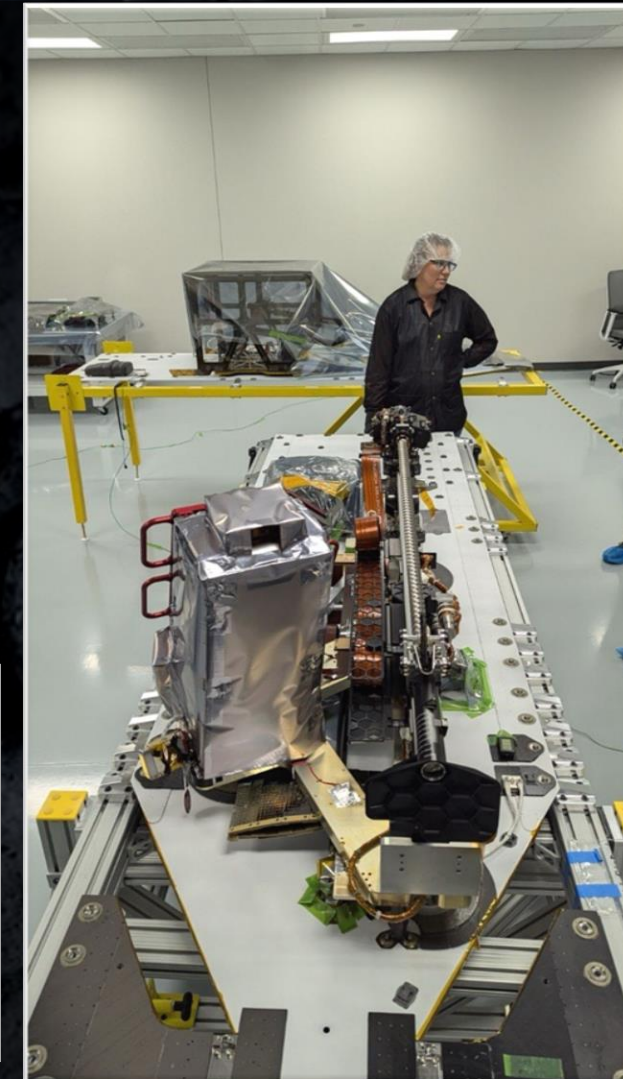


MSolo Cross-Beam Ion Source



TRIDENT Drill Tip

<p>Objective #1</p>	<p>TRIDENT will drill into the lunar subsurface up to a meter deep and deliver regolith cuttings to the surface for water and other volatiles evaluation by MSolo</p>
<p>Objective #2</p>	<p>MSolo will measure within its field of view, the composition of gases emanating near the TRIDENT drill before, during and after drilling activities</p>



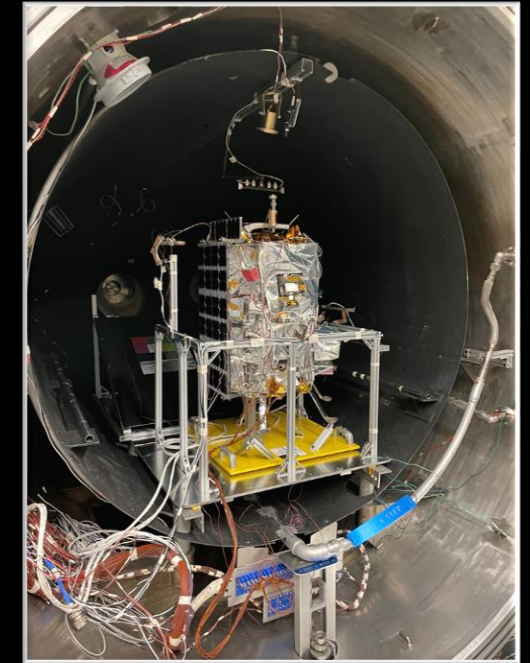
PRIME-1 Hardware Integrated to NOVA-C Spacecraft panel ready for installation

Lunar Deployable Hopper Tipping Point



The lunar deployable hopper project will develop and demonstrate a small robotic hopper, deployed as a secondary payload from the IM-2 Nova-C lander, that can provide access to extreme environments and locations of interest on the lunar surface. μ Nova Hopper is designed to hop into and out of permanently shadowed regions (PSRs), providing a first look into undiscovered regions that may provide the critical science needed to sustain human presence on the Moon. This demonstration drives a commercial venture for IM and helps establish a space economy ecosystem economy.

<https://www.youtube.com/watch?v=Ckcrn1btv8M>



μ Nova tested in a TVAC Chamber



μ Nova Hopper vehicle on test stand

Objective #1	Enable robotic access to extreme lunar environments for scientific exploration
Objective #2	Enable regional exploration of wider areas than small rovers or other mobility platforms can cover

The overall goal of the LTE-TP project is to revolutionize lunar surface communications by leveraging advances in terrestrial communications technology to both improve the quality of communications (lower power, better range, higher bandwidth) while simultaneously reducing cost and providing an on-going commercial communication solution for the lunar economy (both surface and orbital). Future uses of the LTE network will be utilized for both human and robotic missions.

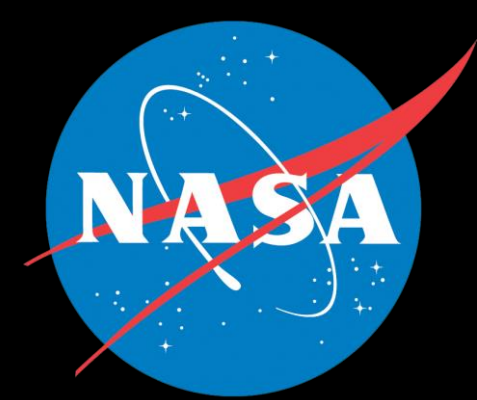


MAPP rover testing with LTE

Objective #1	Adapting Nokia's LTE technology for the lunar demonstration.
Objective #2	Integrate and test Nokia's LTE technology on Intuitive Machines (IM)s Nova-C lander/rover.
Objective #3	Demonstrate Nokia's LTE technology on the lunar surface at short and long range
Objective #4	Provide a post mission lunar propagation model based on the lunar surface test results



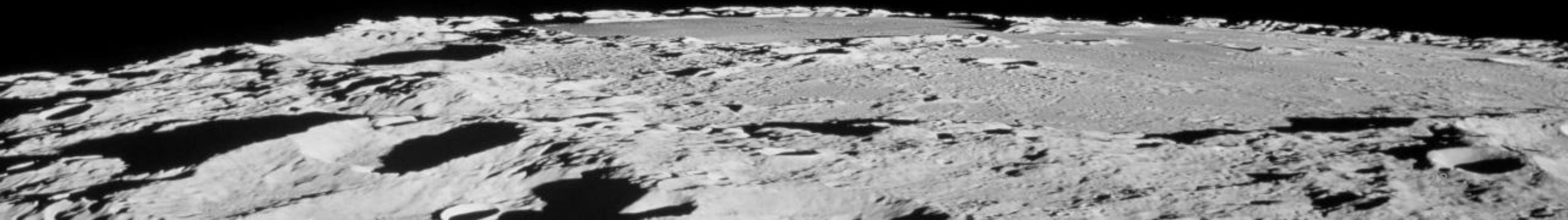
Flight Hardware Integration with IM-2 lander



MAI TAI

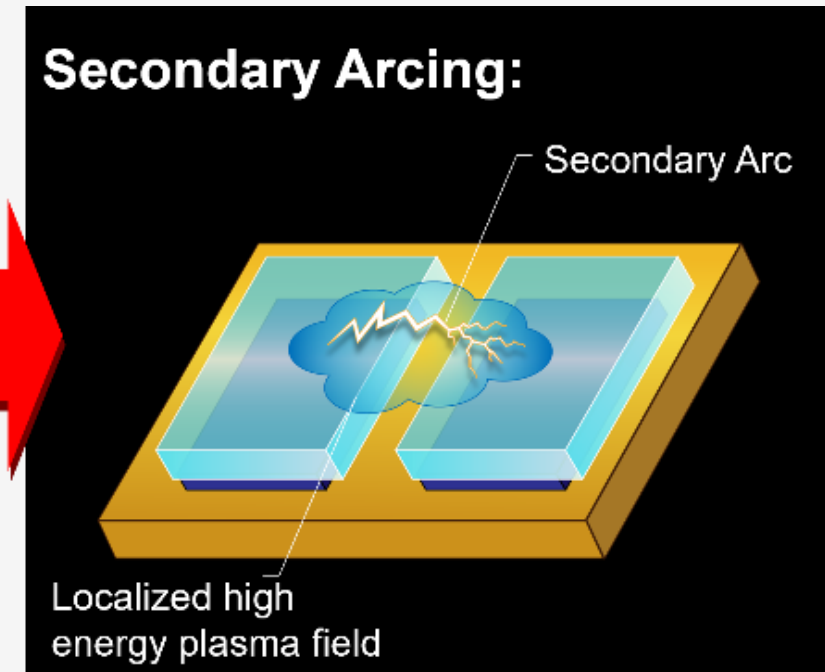
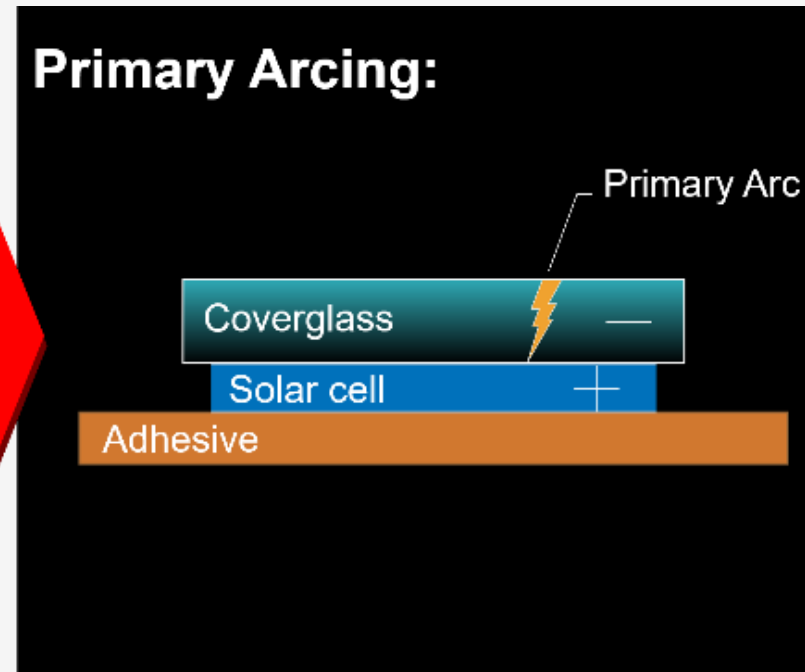
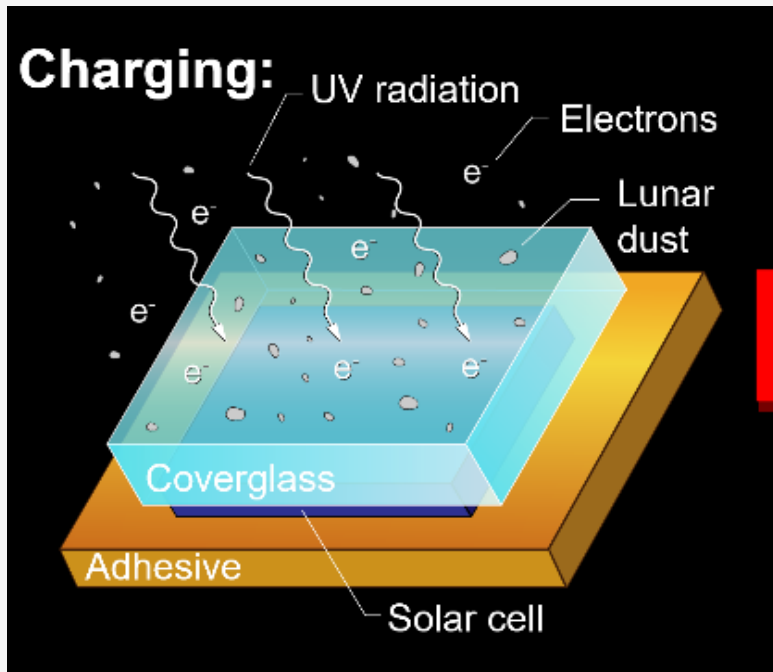
Mitigating Arc Inception via Transformational Array Instrumentation

Meghan Bush, NASA GRC, Photovoltaics and Electrochemical Systems Branch
NAC Technology, Innovation, and Engineering Meeting
September 5, 2024



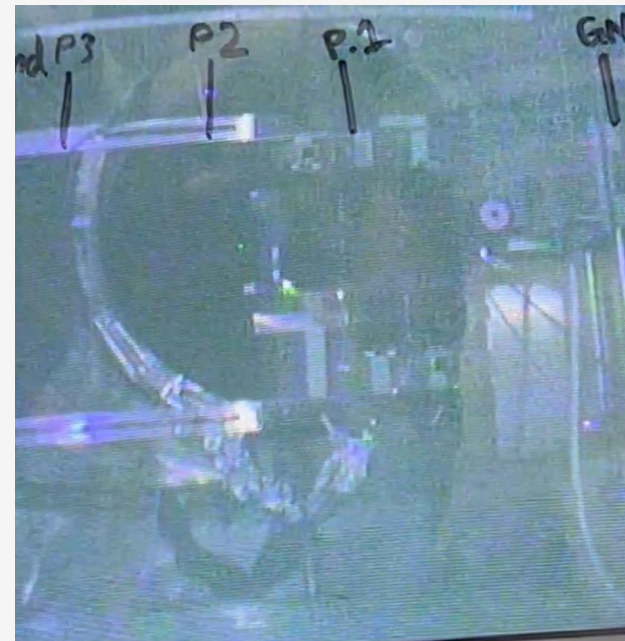
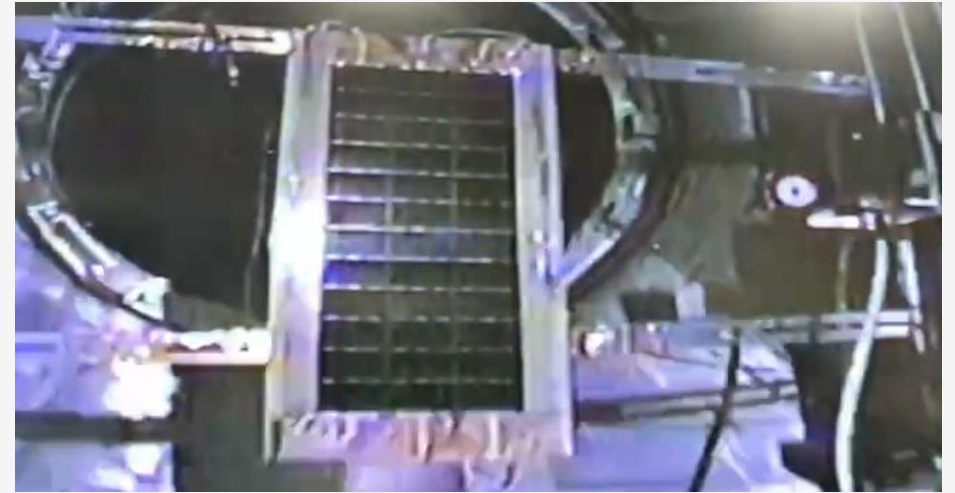
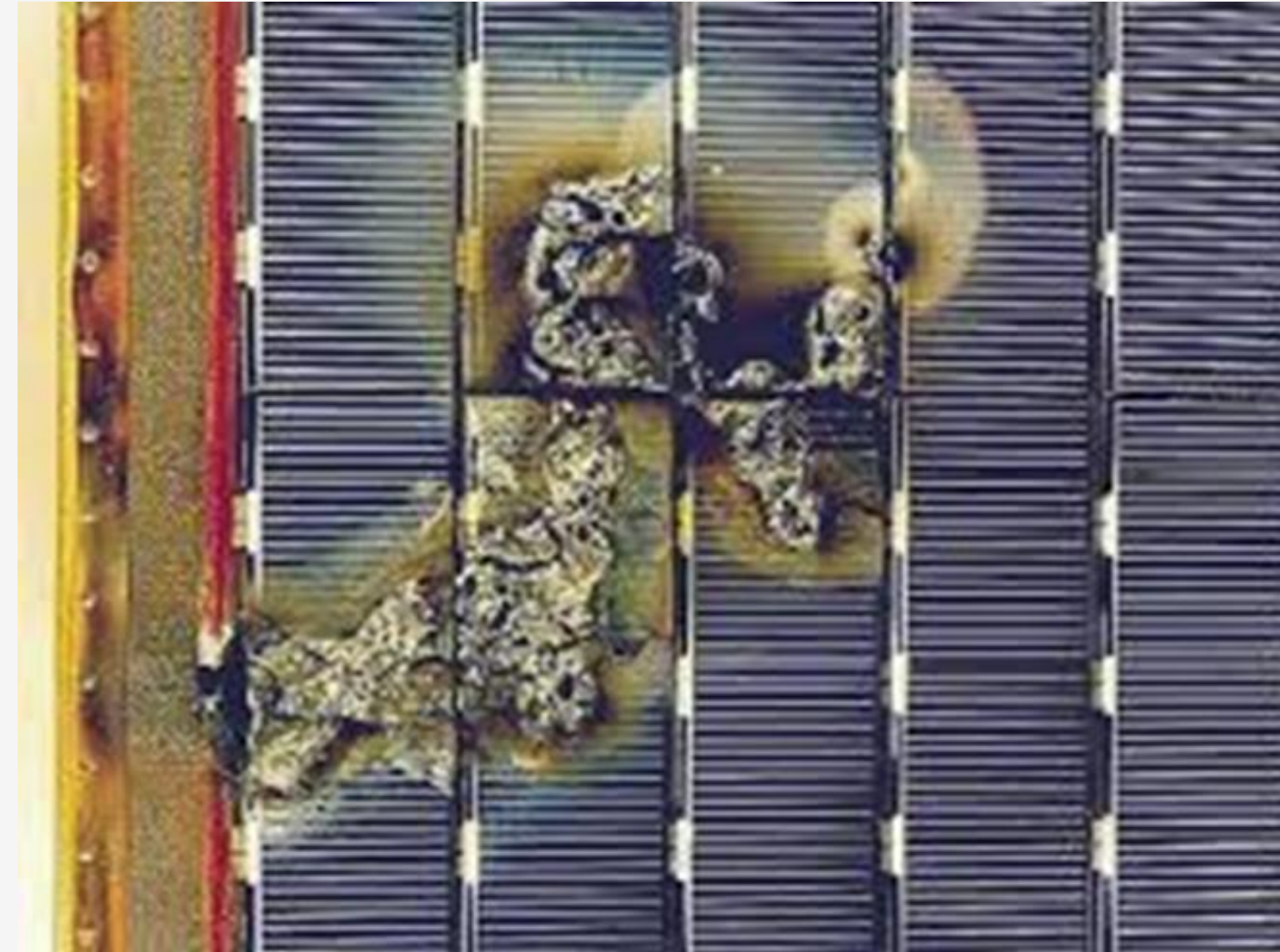


Spacecraft Charging





Secondary Arcing



slow motion
← real time