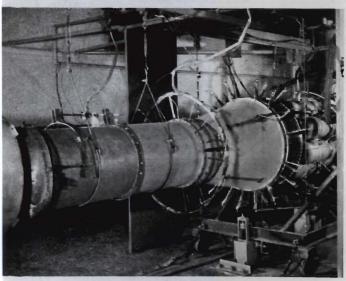


A Journey to Tranquility

NACA Wind Tunnel Test

See Story on Center Pages

COMPARING materials under a barrage of atoms to determine which are most resistant to atomic attack and more useful in missile and rocket manufacture. Atoms are produced by passing a gas at low pressures through a high voltage discharge. Fast vacuum pumps make the atoms hit a material test sample.



RED HOT tail pipe of a turbojet engine. Pressure and temperature readings are being taken.

EXPERIMENTAL MISSILE is installed on the launching racks of an F2H "Banshee" airplane by NACA technicians at the Lewis Laboratory. The missile will be flown from Cleveland to NACA's Pilotless Aircraft Research Station at Wallops Island, Va., where it will be launched over the Atlantic Ocean. Electronic tracking equipment will follow the missile's flight and record its performance at speeds greater than five times the speed of sound.

10 Years Afr

Supersonic Anniversary Puts Flight Laborat

High above the California desert a young Air Force officer from West Virginia made final adjustments in the cramped cockpit of his tiny, bullet-shaped airplane. Soon, he was hurtling towards the earth far below as his craft dropped away from the snug belly of the B-29 mother plane which had borne him aloft. The mancarrying bullet came to life as its four rocket engines spurted their full power. Now it was the mother plane's and the earth's turn to drop far below. The young man in his bullet shot into the upper reaches of the atmosphere.

Minutes later it was all over. The rocket engines had burned out. Plane and pilot safely glided to a fast power-off landing on the expanse of Rogers Dry Lake. This flight was as significant as that of the Wright Brothers at Kitty Hawk—man had flown faster than the speed of sound. Aviation had entered the supersonic age.

This adventure into the unknown was 10 years ago, October 14, 1947. The pilot was Capt. Chuck Yeager. His plane was the X-1, christened "Glamorous Glennis" after his wife. The X-1 was a triumph of the best engineering talent and teamwork of the Bell Aircraft Corp., Reaction Motor, Inc., Air Force and National Advisory

Committee for Aeronautics.

Today, many planes are capable of sustained flight well above that reached by "Glamorous Glennis" but the teamwork between American industry, the military services and the research organizations which made the X-1 performance possible is still functioning to bring forth aircraft of far greater performance.

Significantly on this 10th anniversary of the supersonic era, the attention of American aeronautics is focused on Cleveland this week while the NACA's Lewis Flight Propulsion Laboratory at Hopkins Airport holds its triennial inspection to acquaint the nation's aviation leadership with the trends in aircraft power plant research.

Among the almost 2,000 invited representatives from industrial, military, and scientific institutions will be members of NACA's "board of directors" including: Capt. Eddie Rickenbacker of Eastern Airlines; Fred Crawford of Thompson Products; Lt. Gen. Don Putt, the Air Force's development chief; Vice Adm. William Davis, air chief of the Navy, and Lt. Gen. Jimmy Doollitle, veteran airman and NACA's chairman.

Progress in harnessing the power of new high-energy fuels for air-breat



€ X-1

ory Here in Spotlight

engines and high-energy rocket propellants will be revealed. Other research areas to be reviewed include those dealing with hypersonic (more than five times the speed of sound) propulsion; high-temperature materials and other problems of aircraft nuclear propulsion; high-Mach number engines and aircraft jet noise reduction.

Research in aeronautics is never finished, and research is the business of NACA. Each major discovery or breakthrough opens a new frontier to be conquered and hatches a new brood of problems to be solved if America is to be first in aeronautics.

Every successful plane since that of the Wrights' has been built around its

(Continued on Page 26)

THE COVER

An experimental supersonic aircraft model is prepared by an NACA research scientist and a technician for operation in the test section of the I0x10-foot Supersonic Wind Tunnel at the NACA Lewis Laboratory. This facility permits testing of advanced design aircraft engines and their components at speeds of 1500 to 2500 mph at simulated altitudes up to 30 miles.

Color photos by EUGENE GICZY



TELEVISION CAMERA is focused by NACA technicians on a ramjet engine model through the schlieren optical windows of the 10x10-foot Supersonic Wind Tunnel's test section. Closed-circuit television enables scientists to view the ramjet, used for propelling missiles, while the wind tunnel is operating at speeds from 1,500 to 2,500 m.p.h.



10 Years After X-I

(Continued From Page 25)
power plant. Development of
the propulsion system must
come before the design of the
aircraft in which it is to be installed, otherwise the full potential of neither may ever be
realized. In these days of international competition for survival, few errors are permissible. This is equally true in the
case of missiles.

The go-power, or thrust, of turbojets, rampets, and rockets is developed by force from fire. Combustion creates super-hot gases which expand at high velocities. The velocity and pressure of these gases combine their effects at the jet engine's exhaust nozzle to produce thrust.

In essence, the work of NACA's Lewis Laboratory is to find new fuels to furnish more energy; find new materials which will withstand the greater temperatures and pressures; then of higher energy fuels find the best means to harness the higher temperatures and pressures.

An example of the problems facing scientists is weight. About 90% of the weight of a rocket-powered vehicle is its proposal. The choice of propel will, to a large extent, determine the size, weight, and

shape of the vehicle. In the present Vanguard project to launch a 20-pound earth satelite, the vehicle to carry it aloft is about 75 feet long. Perhaps a 25,000-pound payload would be required for a manned satellite.

If a 75-foot vehicle is required to launch a 20-pound space satellite, the size of a vehicle to launch a 25,000-pound payload would be ungainly, to say the least. New higher energy propellants would result in obvious weight and size advantages.

Other problems to be solved include those caused by temperatures, hot enough to change the behavior of matter, induced on a missile's re-entry into the earth's atmosphere, and precise control of missile fuel consumption. The materials to withstand these temperatures and systems to handle the propellants with target-hitting precision are among the many goals of research.

That some of the problems of flight are being solved is evidenced by advances in plane and missile speeds since Capt. Yeager's flight in the X-1. By 1953 manned aircraft had reached 1,650 m.p.h., and last year the X-2 flew about 2,100 m.p.h. The missile speeds of 10 years ago

were only somewhat better than those of the World War II, 3,500 m.p.h. German V-2s. By 1949 missiles were in the 5,000 m.p.h. class. Last year, an NACA four-stage, rocket-powered research vehicle reached 6,864 m.p.h. Other rocket-powered research vehicles have since bettered this speed.

Contributions to the powerplant performance of these aircraft were made by skilled NACA scientists working at Cleveland with complex re-search facilities. The latest of these will be seen during the Lewis inspection. They are: The 10x10-foot supersonic wind tunnel, completed last year at a cost of 32 million dollars; the 8x6-foot supersonic wind tunnel. on which extensive modifications to extend its speed range were recently completed, and the rocket engine research facility, also completed this sum-

With both wind tunnels, scientists can study full-size plane and missile power plants operating at speeds past 2,500 m.p.h. at simulated altitudes up to 30 miles. The tunnels permit investigation of an engine's thermodynamic and aerodynamic performance, operating temperatures and stresses, control systems, and air inlet and exit performances. Among the contributions of the 10x10 during its first year's operations are supersonic performance data on

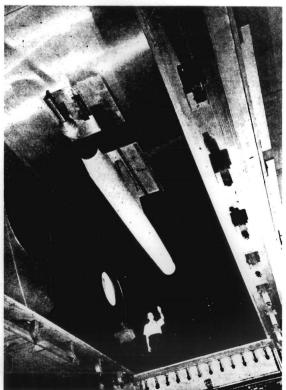
(Continued on Page 28)



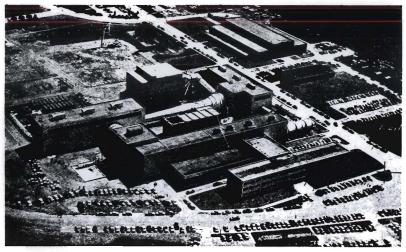
"SCRUBBER" is part of the silencing and exhaust gas disposal system of the new rocket engine research facility at Lewis. Water sprays within the scrubber remove rocket exhaust products. Like a giant automobile muffler, the scrubber also silences sounds of rocket engine operation.



***AM [ET TEST.** Another view inside the 10x10-foot wind tunner. Note the mirror-like polish of walls and floor. Men who set up the tests must remove shoes before entering tunnel, or else wrap shoes in towels.



AFT VIEW of a 16-inch ram jet engine in test position, seen through the floor opening of the wind tunnel.



AIR VIEW of the 10x10-foot supersonic wind tunnel. The air force through this tunnel makes a hurricane seem like a gentle summer breeze. It permits high altitude tests of full-scale engine and aircraft models at air speeds three and one-half times the speed of sound.

10 Years After X-I

(Continued From Page 27) the B-58 engines, various ramjet engines, and high-energy fuels.

The 10x10 covers an area of about two city blocks. The air which pushes at hurricane speeds through its test section is moved by two large compressors turned by seven electric motors producing 250,000 horse-power. Air enters the tunnel through an air dryer building which is 117 by 84 feet, 82 feet high Here moisture is removed by heat and chemicals at a rate equal to the capacity of about 12,000 home clothes dryers.

Walls of the test section are stainless steel plate, 10 feet wide, 78 feet long, and 1% inches thick. Hydraulic jacks can squeeze portions of these walls as much as two and one-half feet each to form a variable nozzle for different air speeds.

For the benefit of the neighbors, much effort has been expended to operate all Lewis research facilities as quietly as possible. A two-story structure silences the 10x10 tunnel noise as air is exhausted back to the atmosphere. A 24-foot valve also permits recirculation and reuse of the air within the tunnel.

The 8x6-foot tunnel is designed to produce the lower flight speeds in which all aircraft power plants must operate at times. For accurate data at transonic (at or near the speed of sound) speeds, 4,700 holes were bored in the four walls of the test section. The perforations let air "bleed" through

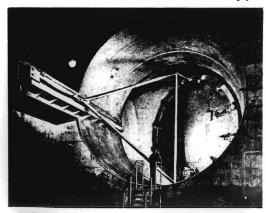
the walls. Transonic speeds could not be duplicated in wind tunnels before this "bleeding."

Scientists at the new rocket engine research facility will investigate, with practical-sized rocket engines, methods of utilizing high-energy propellants. Versatility of the equipment permits initial studies using low-cost fuels. There is a thrust stand where the rocket engine is securely mounted during tests; propellant supply and storage systems; an exhaust gas disposal system, and silencing equipment.

During operation, 50,000 gallons of water per second are sprayed into the exhaust duct to scrub combustion products from the rocket exhaust and to help silence normal operating noise. Closed circuit TV cameras, also a feature of the two large tunnels, let operators view activities from the control room.

Lewis, on which construction started in 1941, is one of three major research centers operated by NACA. The others are the Langley Aeronautical Laboratory, near Hampton, Va., and the Ames Aeronautical Laboratory, near San Francisco. Smaller NACA research installations are located at Wallops Island, off the Virginia coast, and at Edwards, Calif.

NACA policy is established by the committee itself, a 17-man body whose membership is appointed by the President and serves without pay. There are 28 subordinate committees composed of about 400 specialists who also serve without pay.



SWINGING VALVE. 24 feet in diameter, is one of the key control elements of the Lewis 10x10 wind tunnel. In one position, it seals off the tunnel exhaust, making the tunnel a closed circuit, which is used for aerodynamic testing of models. In its other position, the valve acts as a seal across the tunnel.



miles an hour at altitudes of 20 miles will be tested in this \$2,500,000 base at Cleveland Hopkins Airport. HIGH-POWERED FUELS for rocket engines that will push missiles thousands of

Develop Fuels Here to Power Planes 4 Times Speed of Sound

sound.

Akron, Ohio

October 7, 1957

Fuels Perfected

eveland Laboratory

2,600 MPH Plane

an hour, or to New York in a little over 15 minutes.

mit aritiners to travel at four times the speed of sound.

This 2600-mph speed would 2000 of the nation's top air. chairman of the National Adallow a piane to streak from craft experts gathered at the visory Committee on Aeronautics; Eddie Rickenbacker, AKRON BEACON JOURNAL

Edward R. Sharp, director

CLEVELAND—Scientists here have perfected fuels that

will permit airlines to travel at four times the speed of This 2,600 mph speed would permit a plane to streak from Cleveland Hopkins Airport to San Francisco in less than an hour, or to New York in a little over 15 minutes.

The spectacular development was disclosed today as wraps off a new \$2,500,000 2,000 of the nation's top air rocket engine research factoraft experts gathered at the lity. National Advisory Committee on Aeronautics laboratories at Cleveland-Hopkins Airport for the installation's "triennial in-inspection."

Creveland-nopming Samuel in Inspection.

The 2,600 MPH speed for passenger planes made possible the intercontinental possible the intercontinental passenger planes made possible the intercontinental passenger planes made possible the intercontinental passenger planes made possible the intercontinental passenger planes and passenger planes and possible the intercontinental passenger planes and passenger planes and the plane and tuels of tomorrow, took the fuels of tomorrow (that's the official title of the NACA in stallation here) experts said the new liquid fuels with a high boron content are now expensive and scarce.

But, they said, new plants under construction will soon make these fuels available for wider use and flight tests, as well as laboratory experiments.

By CHARLES TRACY
Press Aviation Editor
Cleveland scientists have perfected fuels that will permit airliners to travel at four
The spectacular develop.
These present included the specific of the spectacular develop.
The spectacular develop.
These present included the specific of the spectacular develop.

board chairman of Eastern Airlines; Fred Crawford, board chairman of Thompson Products, and John Victory, NACA's national secretary.

of the 100-million-dollar collection of wind tunnels and laboratories devoted to development of the planes and fuels of tomorrow, took the wraps off a new \$2,500,000 rocket engine research facil-

NACA Missile Achieves Speed

A 2,000 Mph Missile! **New Fuel Gets Credit**

from the fuel research now under way.

NACA scientists told visitors whe laboratory has been studying the effectiveness of various borane fuels and recently has tried them out the engines.

Officials asid the results have been encouraging, but some difficult problems remain. These in Compounds can be quite toxic.

2. Combustion deposits inside the engines, which cut down on performance.

Laboratory of ficials as and used results when the provided the engines of the property research now underway may result in greater gains and surpass the rocket developments of the past few years. They said their study of the various boron fuels have been "encouraging" but a dmitted some difficult problems remain such as danger to personnel because boron compounds can be quite toxic and combustion deposits inside the engines which cut down on performance.

2. Composition deposits missies the engines, which cut down on performance. MACA officials explained that while the petroleum fuels now the petroleum



An aeronautical research scientist at the Lewis Flight Propulsion Laboratory displays a model of a theoretical Mach 4 turbojet engine based on a composite of advanced ideas from component research and cycle analysis.

> WASHINGTON STAR Washington, D.C. October 7, 1957

resist have centered upon because of its late on the proposed of the first semipublic inspection. Visiting experts were shown how the facility is equipped to test fuels in full-sized rocket engines roaring at top speed. Other disclosures by NACA scientists: HIGH ENERGY FUELS tested here can increase the range of supersonic planes and missiles by 40%. This means that the United States has perfected fuels that make possible the intercontinental ballistic missile or a rocket to the moon. RESEARCH HAS PRODUCED plans for a new fushed permit passenger planes to travel at supersonic speeds. The 2600-miph speed for images and more as the range of speed of sund of the later step has now the full permit passenger planes made to the mount of the later step has now the full permit passenger planes and means that the united States has perfected fuels that make possible the intercontinental ballistic missile or a rocket to the moon. RESEARCH HAS PRODUCED plans for a new fushed permit passenger planes and the conventional cigar shape. The new laboratory, completed last expended for the triemal inspection of the later step has already the proposal provided by conventional provided by co

CLEVELAND (#)—A ramjet has tried them out in full-test missile burning one of the compound, has flown more than three times the speed of sound, or faster than 2,000 mph.

The was disclosed today at The triangle largestic of the

This was disclosed today at the triennial inspection of the Lewis Flight Propulsion Labo-compounds can be quite toxic. ratory of the National Advisory Committee for Aeronautics the engines, which cut down on (NACA), the top government performance.

NACA of ficials explained the work of the compound of

nder way.

NACA scientists told visitors
the laboratory has been studying the effectiveness of various boron fuels and recently

agency in basic air research.

NACA SHOWED publicly for the first time a 2½-million-dollar rocket engine research last August.

The agency said that the spectaceular rocket development of the past few years may be surpassed by even greater gains from the fuels research now under way.

Boron-Fueled Missile Tops 2,000 M.P.H.

CLEVELAND, Oct. 7 (**P).—A rampet test missile burning one of the new high-energy fules, a boron compound, has flown at more than three times the speed of sound, or faster than 2,000 miles per hour.

This was disclosed today at the triemnal inspection of the Lewis Flight Propulsion Ladvisory Committee for Aeronautics, the top Government agency

tics, the top Government agency in basic air research.

The committee showed pub-licly for the first time a \$21_million rocket engine research facility which was completed last August.

The agency said the specta-cular rocket developments of the last few years may be sur-passed by even greater gains from the fuel research now under way.

under way.

The laboratory's scientists told visitors they have been studying the effectiveness of various bronn fuels and recently have tried them out in full-scale ramiet and turbojet engines. Officials said the results have been encouraging, but some difficult problems remain.

These include:

1. Danews to provide the second of the s

These include:

1. Danger to personnel. Boron compounds can be quite toxic.

2. Combustion deposits inside the engines, which cut down on performance.

Officials explained that while the petroleum fuels now burned in jet aircraft are relatively cheap, plentiful and safe, their energy content is too low to

energy content is too low to satisfy the needs of military planes and missiles.

DEVICE TO POWER
SPACE SHIP SHOWN

U.S. Scientists DisplayModel of Engine Designed for Use Beyond Atmosphere

By RICHARD WITHIN Second by New York Time.
CLEVELAND, Oct. 7—Government scientists demonstrated they are seeking to develop for light in outer space.
The engines, known as long gun, would proutee only the content of the seeking to develop for light in outer space.
The engines had propelled where there was no longer any and the seeking to develop for light in outer space.
The engines had propelled where there was no longer any and the seeking to develop for light in outer space.
The engines had propelled where there was no longer any the seeking to develop for light in outer space.
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The engines had propelled where there was no longer any through the Lewis Fairly speciments and the seeking to the seeking to develop for light in outer space.

The engines had propelled where there was no longer any through the Lewis Fairly speciments and the seeking to the seeking to develop for light for light in the seeking to the seeking to develop for light for light in the seeking to the seeking to develop for light in the seeking to the seeking to develop for light in the seeking to the seeking t

Government's top aviation re-search organization.

High-Speed Engines Planned
A tour through the Lewis fa-cilities includes these other
A tour through the Lewis fa-cilities includes these other
AN A. C. A scientists have concluded from recent research
that it should be possible to develop turbo-jet engines for speeds up to four times the speeds up to four times the speeds up to four times the speeds and the speed might be feasible not only for military craft but for commercial air-craft but for commercial air-mile-an-hour jet airliners in mile-an-hour jet airliners in mile-an-hour jet airliners in service in Russia and soon to enter service here are turbo-jets.

The problem of suppressing jet-engine noise to piston-engine hovels has not yet been developed. The rest of the job could be accomplished through a complete the piston-engine hovels here have not yet been perfected, according to the ATCA. As a grave question whether the highly publicized, high-energy boren fuels will ever be used. The fuels at present, have serious drawbacks in heavy deposits in the engine. Provisions for Power The "ion gun" engines for maneuvering satellites or space strongulation in Provisions for Power The "ion gun" engines for maneuvering satellites or space strongulation in Provision for Power The provision for Power The provision for Power of the provision in Provision for Power The strongulation in Provision for Power Development of the provision for the provision for Power Power The strongulation in Provision for Power Power The strongulation in Provision for Power Power The strongulation for Power Power The Power Power Power The P

INDIANAPOLIS NEWS

Goes 2,000 MPH CLEVELAND (AP)—A ram-jet test missile burning one

of the new high-energy fuels, a boron compound, has flown faster than 2,000 mph. The flight at more than

three times the speed of sound was revealed yesterday by the National Advisory Committee for Aeronautics, the top government agency in basic air

WASHINGTON POST Washington, D.C. October 8, 1957

Space-Ship Engine Forerunner Exhibited Cleveland this week.

THE SUN Springfield, Ohio October 8, 1957

Scientists Predict Turbojet Aircraft many problems yet to be licked many problems. Set to be licked with the problems and the

Indianapolis, Indiana on Stream U.S. Test Missile May Power Space Flight

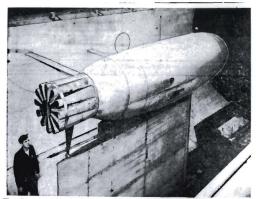
WASHINGTON STAR

October 8, 1957

Washington, D. C.

CLEVELAND, Oct. 8 (P) The National Advisory mittee for Aeronautics bombarded a small pinwheel inside a glass vacuum-jar with an invisible stream of ions today and made it whirl briskly.

and made it whird briskly. The same ion-stream principle, the committee said, some day may be harmessed to push 10-ton space ships monouvard at many thousands of miles an hour once they committee the same of the same ships and the same ships



Here is an experimental jet engine noise suppressor developed at the Lewis Flight Propulsion Laboratory here. It cuts jet noise from about 120 decibels, almost deafening, to 108, which is loud but not deafening. Scientists hope eventually to

Forerunner Exhibited By John G. Norra CLEVILAND. Oc. Target and the present of the present of

Agency Shows New Unit.

Tells of Rocket Gains

CLEVELAND.—(AP)—A ranjet test missile burning one of the new high-energy fuels, a boron compound, has flown more than three times the speed of sound, or faster than 2,000 m.p.h.

This was disclosed at the triennial inspection that Propulsion Laboratory of the National Advisory Committee for Aero-fully to the first time a 12,000 m.p.h. included it is a speed of 2 of 100 m.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both the first time a 12,000 m.p.h. included the use of both times the special study to the first time a 12,000 m.p.h. included the use of both times the special study to the first time a 12,000 m.p.h. included the use of both times the special study to the first time a 12,000 m.p.h. included the use of both times times the special study to the special study to the study the meets times to see the special study to the study times and the proposition times the special study to the study times the study end to the study times and times times times to see the study times the seating end to the study end to the study end to the study end times times times to see the study end times times times to see t

search facility which was completed last August.

The agency said that the spectacular rock
developments of the past few years may be
surpassed by even greater gains from fuel research currently under way.

N.A.C.A. scientists loid visitors the laboratory has been studying the effectiveness of
over the control of the control of the control
them out in full scale rample and turbojut enentess.

Top Federal Air Rescarch agency Shows New Unit, elatively cheap, plentiful and safe, their energy content to too low to satisfy the needs of military planes and missiles.

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The agency said that the spectacular rocket developments of the past few years may be surpassed by even greater gains from fuel research currently under way.

Lory has been studying the effectiveness of a real case of a redically new fuel type unster consists of a redically new fuel type and sometiment of the consists of a redically new fuel type as the fuel type and sometiment of the consists of a redically new fuel type and sometiment of the consists of a redically new fuel type as the fuel type of a redically new fuel type as the fuel type of a redically new fuel type as the fuel type of a redically new fuel type as the fuel type of a redically new fuel type unster consists of the consists of the section of the later of a redically new fuel type as the fuel of a redically new fuel type as the fuel of a redically new fuel type of a redically new fuel type as the fuel of a redically new fuel typ

2,600 M.P.H. Jet Predicted by U.S. CLEVELAND, Oct. 7—APJ—The Federal Gov mement is to percentage scientist to down for sainty the needs of percentage and the second percent

2,000 mph on New Jet Fuel | Silencer Cuts Down

of the corrugations, and to mix this air with the exhaust

mix this air with the exhaust fumes leaving the plane. Sanders said this device could cut the engine's noise by 13 decibels, reducing it to only 1/16 that of the conven-tional engine.

scoop outside air through vents forming the outer side

NEW YORK HERALD TRIBUNE October 8, 1957

whether would ever be will make manned alteract observed for U.S. defense.

Abe Silverstin associated here to give for warfare.

Abe silverstin associated will make manned alteract obsolete for warfare.

Visitors at the triennial inspection at Lewis lab were shown a model of a high-altitude turbole airplane that one could be made to fly; sometiments of the country of the country

and fuel into the burner chamber. "Fluorine - supported flames may be 2000 degrees to 3000 degrees (Fahrenheit) hotter than oxygen-jet fuel flames. which reach 5000 degrees (Fahrenheit)."

U. S. Reveals a Model Of Space-Ship Rocket

By Ansel E. Talbert

Millary and Aviation Editor

CLEVELAND. Oct. 7—A

intherto secret working model

of an ionic rocket — a new

of an ionic rocket. — an experimental according to

rature manned space stips and earth satellites — was shown to

tousy for the first time by the

National Advisory Committee

The committee, then ation of a secret than 2,500 miles an

on Aeronauties. The committee than the property of the committee of the committee

The officials said that if a stream of ions could be accelerated to high velocities by use of electric or magnetic fields, their energy could give thrust which might propel satellites in flights beyond the Earth's atmosphere.

The scientists disclosed that

The scientists disclosed that in addition to dizzying speeds and space flight they also are working on more mundane headaches.

headaches.

They said one problem which might arise in connection with the jet airliners scheduled to begin service in 1959 was that their engines on the ground sometimes tend to act as mammoth vacuum cleaners.

cleaners.

The engines might pick up pebbles, small tools and debris with occasional catastrophic results," the scientists said.

NACA is working on methods to prevent these results and has met with some success, they explained.

Newport News, Virginia October 8, 1957

IONIC ROCKET POWER PLANNED TIMES - HEROLD Model Of Manned Space



NACA engineer holds model of Mach 4 turbojet engine.

By ANSEL E. TALBERT Herald Tribune News Service Herald Tribane News Service
CLEVELAND — A hitherto servet working model of an ionic
rocket — a new propulsive sysships and earth satellites — was
shown yesterday for the first
time by the National Advisory
Committee for Aeronautics.
The committee also revealed
that its current research projects
above that ship — flying aircraft
show that ships — flying aircraft
bow that ships — flying aircraft
show that ships — flying
times the speed of sound. The
NACA is headed by Dr. James

Ship Revealed By NACA
ANSEL E. TALBERT
Id Tribune News Service
Id Tribune News

S TIMES HERALD

SNY Virginia

7 8, 1957

PLANNED MACAA
A Doollith, former Air Force
leed By NACA
A Doollith, former Air Force
leed By Nacca, and in the state of the state of

outer space.

The NACA up to now has not been closely connected with the current earth satellite or Vanguard project of the United States, which has just been beaten to completion by the Soviet Union. The U.S. satellite probated on existing fuels and chemical rocket engines, while the NACA's prime job is one of advanced research.

Pave Way Here for Space Travel

By CHARLES TRACY, Aviation Editor

Vital research in the U. S. race to send American airmen orbiting through space ahead of the Russians is going on in NACA's Air Lab at Cleveland Hopkins Air-





ATOMIC PLANE'S SHIELDING, made of lead and paraffin, is installed in test model by NACA nuclear scientist, James Blue, for NACA visitors Capt. Russell Langlois (center), of the Air Force research staff in Washington, and Air Commodore R. C. Cameron, Royal Canadian Air Force liaison officer in Washington. of the Air Force research staff in Washington, and Air Commeron, Royal Canadian Air Force liaison officer in Washington.



U. S. Reveals a Model Of Space-Ship Rocket

By Ansel E. Talbert

Military and Aviation Editor

CLEVELAND, Oct. 7—A

hitherjo secret working model

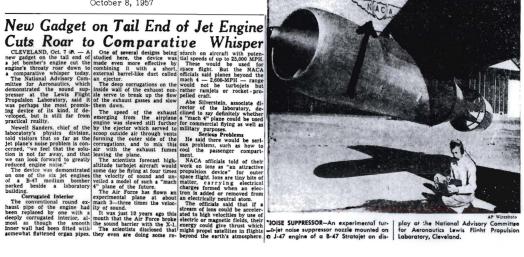
of an ionic rocket — a new
propulsive system intended for
future manned space ships an earth satellites — was shown
today for the first time by the
National Advisory Committee
The committee, the nations
The position of the nations
The propelled by combined
the three major research establishments of the N.A. C.A.
Can be built to transport human which now has \$100,000.004
The N.A. C.A. is headed our worth of research 'establishments of the N.A. C.A.
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THE JOURNAL Wilmington, Delaware October 8, 1957

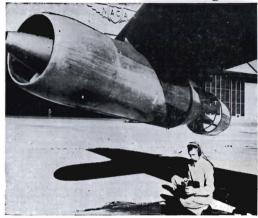
New Gadget on Tail End of Jet Engine Cuts Roar to Comparative Whisper

SATELLITE MODELS with wings, designed to carry a man 1050 miles up into outer space, are displayed to NACA visitors C. C. Weismann (left), and Lieut. C. W. Stoddard, from the Office of Naval Research. Washington, by Rocket Engineer Edmund Jonash. The winged models are man-carriers while the slender model held by Lieut. Stoddard is a copy of the Vanguard satellite to be launched next spring.



ROAR BECOMES WHISPER

Jet Noise War Shows Progress



An experimental turbo-jet noise suppressor nozzle mounted on a J-47 engine of a B-47 Stratojet was on display yesterday at the Notional Advisory Committee for Aeronautics Lewis Fight Propulsion Laboratory at Cleveland, O.

Difficulties ore encountered in the de-sign of such nozzles, to keep drag, weight and engine performance penal-ties at a minimum while at the same time accomplishing the desired noise reduction. (AP Wirephoto)

a comparative whisper yesterday.

The National Ad visory
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Propulsion Loboratory of Claveland, O. reduction. (AP Wirephoto)

Claveland (AP)—A ne we nonstarted on one of the six engine was slowed still furgrated to the six engine of a plet engines of a B47 medium propulsion. The comparative whisper yesterday.

The National Advisory to the engine that been replaced by one with been replaced by one with been replaced by one with a been replaced by one with a been replaced by one with a comparative whisper yes to the engine had been replaced by one with a comparative whisper with the been replaced by one with a comparative whisper with the been replaced by one with a comparative whisper special with a some what is flattened superseasor at the Lewis Flight from the size of the comparation of the comparative whisper with the propulsion. Laboratory, said it with a some what is flattened with a some what is flattened to the comparative while the size of the comparation of the comparative while the propulsion. Laboratory said its way be a size of the comparation of the comparative while the propulsion of the comparation of the comparation

One of several designs being studied here, the device was made even more effec-tive by combining it with a short, external barrel-like duct called an ejector.

The deep corrugations on the inside wall of the exhaust nozzle serve to break up the flow of the exhaust gasses and slow them down.

THE DEVICE WAS dem-

Cottober 8, 1957

THE SPECIALLY fitted engine was run up first, and emitted a modest amount of noise. Then one of the other General Electric J-47 engines was operated without the equipment to demonstrate the difference.

Sanders said this device.

Sanders said this device could cut the engine's noise by 13 decibels. He said this meant that the noise power of the specially fitted engine was only 1-16th that of the conventional en-

gine.

The physicist explained that
the greater part of the noise The physicist explained that the greater part of the noise of a jet engine comes not from the flying of its compressor blades or the rapid combus-tion of fuel but from the mix-ing of the exhaust jet with he atmosphere.

he atmosphere.

He said the hot gas and ah
roll up into irregular swirh
producing fluctuating pres
sures that radiate sound

Aviation Dally_

NACA FACILITY AUGKS ON MACH . TUREWEY

The National Advisory Committee for Aeronautics announced its Levis Flight Pr pulsion Laboratory in Cleveland is now working on the preliminary development of a Nach 4 turbojet engine capable of driving an aircraft 2,600 mph at high altitudes.

Occasion of the NACA announcement was the triennial inspection of the \$100,000,000 Levis Research facility -- one of the three major NACA facilities. Levis conducts

research in all advanced areas of aircraft and missile propulsion, including turbojets, ramjets and rocket engines, fuels and oxidizers, structural materials and oper-

Reporters attending the inspection were told that a Mach 4 turbojet engine might have only a three-stage compressor, compared to the 12 to 15 stages in present engines, and that it would have much a smaller combustor section than present powerplants, leading to considerable compactness.

Such a high-speed engine would require variable air inlets and exhaust nozzles

Such a nign-speed engine would require variable air finets and exhaust nozies in order to operate over the broad range of speeds from take off to Mach 4. Without such variable geometry, MACA said, the engine would probably be unable to Tly faster than Mach 1.4 and the takeoff itself would be "marginal."

"Although it appears that aerodynamics and combustion principles of a Mach 4

"Although it appears that aerodynamics and combustion principles of a Mach 4 engine are being worked out, practical construction will be difficult because of the high temperature environment in which engine components must operate," RACA said.

"The compressor temperature of a Mach 4 turbojet is so high that its construction will require the use of alloy currently being used for turbine blades."

Bigh Ran Air Temperatures Troublesome

MACA scientists explained that the high ram temperatures encountered by a Mach 4 engine, about 1,2007, will be principally responsible for the more difficult materials involved in such an engine. While the strength and temperature of compressor blades

involved in such an engine. While the strength and temperature of compressor blades must be increased, the turbine blades themselves might have even greater strength and temperature resistance. Engine bearings and seals also present difficult problems because of the high level of ram air temperatures.

MACA also disclosed that it has conducted successful test flights at its Wallops Island, Va. research station with experimental ramjet vehicles burning a high-energy boron-compound fuel. One vehicle attained a velocity "greater than a Mach number of " MACA said.

The Levis Laboratory is giving considerable attention to the problem of hyper sonic flight -- at speeds of more than Mach 5, or 3,300 mph. At this speed, NACA said, air temperatures reach 2,000° F. on engine surfaces, while at Mach 7 they reach 4,000° F.

Much research is now directed at the problem of cooling engines operating in these speed ranges, including the possibility of using fuel as a coolant.

The Levis laboratory also showed off a \$2.5 million rocket engine research facility completed in August. It will permit tests of rocket engines up to 20,000 lbs. in thrust and will be particularly valuable in the study of higher energy rocket pro-pellants like fluorine and liquid hydrogen.

Other areas under study at Levis include:

1 - Jet noise - experiments are simed at the redesigning the exhaust nozzles of jet engines to reduce the velocity of the exhaust jet and thus cut noise. One promising area is the possibility of designing engines with lower jet velocity in combination with special nozzle shapes to minimize exhaust turbulence.

2 - Operating problems - jet engines create vortexes at the inlet which behave the redesigning area to be supported by the control of the control

2 - Operating problems - jet engines create vortexes at the inlet which behave like gigantic vacuum cleaners and suck up all manner of dirt, tools and other foreign matter. MACA is studying various methods of avoiding engine damage of this type.

3 - Materials - MACA wants to increase turbojet engine operating temperatures from the present 1,650° F. to more than 2,000° F. within the next few years. To achieve this, new, more heat-resistant materials are necessary. Levis is now looking into a variety of possibilities for getting greater heat-resistance, including the use of ceramics, new alloying methods and new kinds of alloys which resist oxidation-

San Jose Evening News Tuesday, Oct. 8, 1957 15

Model Turbojet Engine



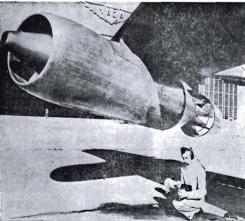
CLEVELAND, Oct, 8.—An engineer holds a model of a theoretical turbojet engine designed for speeds in the Mach 4 (four times the speed of sound) range, Scientists at Levis Flight Propulsion Laboratory here said they are confident of working furbojet engine can be built despite "formidable obstacles."

Ion Bombardment May Serve As Power Source For Future Space Ships

CLEVELAND. Oct. 8.—(AP)—

randa of rules an hour once the first factor of the National Advisory Committed to first factor of the National Advisory Committed to first factor of the National Advisory Committed to first factor of the National Properties of the Nation

In order to simulate the searing speed for an earth satellite to temperatures which will be experienced by aircraft and power speed would be required to break persons of the properties of the



Goal—Noise Reduction

An experimental turbo-jet noise suppression nozzle mounted on a J47 engine of a B47 Stratojet was on dis-play today at the National Advisory Committee for Aeronautics' Lewis Flight Propulsion Laboratory at

Cleveland, Ohio. Difficulties are encountered in the design of such noz-zles to keep drag, weight and engine penalties at a minimum while at the same time accomplishing the desired noise reduction. (AP Wirephoto)



ATOM EFFECTS—During trienniel inspec-tion of Cleveland NACA, cloud chamber re-search shows how radiation affects aircraft.



LAB SKY—Simulating high atmosphere is this rocket-missile experiment. Dr. E. R. Sharp, a native of Hampton, headed Lewis research.



WIND TUNNEL—Emphasis at Lewis is on air propulsion. Here, scientists prepare an aircraft model to test in 10 by 10 2,500 mpb tunnel.



NOISE MUFFLER.—Many Langley NACA of-ficials are at Lewis inspection and will see model of turbojet noise suppressor, here on engine.

THE TIMES-HERALD

SAN FRANCISCO CHRONICLE Wednesday Oct 9 1957

THURSDAY, OCTOBER 10, 1857

PAGE 13

NACA Lectures Concerned With New Engines

New engines with high strength at extreme temperatures generat-ed in hypersonic flight was con-of the major points being discussed Levis Laboration to the National Advisory Committee for Aeronau-tics in Cleveland, O.

ties in Cleveland, O.

The inspection, similar to the one held last October at Langley Air Force Base for the NACA, has considered to the control of the con

been created in the Lewis latora-tory.

Mich research on the whole problem of heat resistance in ma-problem of heat resistance in ma-the Lewis spokemen said in their demonstration. Although results have been encouraging, considerable more work needs to be done before the highly advanced materials for use in engines or structural aircraft and missile parts will be available.

New Gadget Cuts Down Jet's Roar

CLEVELAND, Oct. 8 (AP) A new gadget on the tail end of a jet bomber's engine cut the engine's throaty roar down to a comparative whisper this week.

The National Advisory Com-The National Advisory Com-mittee for Aeronautics, which demonstrated the sound sup-pressor at the Lewis flight-propulsion laboratory, said it was perhaps the most prom-ising device of its kind.

The device was demonstrated on one of the six jet engines of a B-47 medium bomber parked beside a laboratory building.

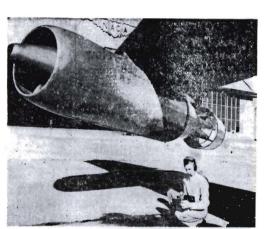
The conventional round ex-The conventional round ex-haust pipe of the engine had been replaced by one with a deeply corrugated interior, almost as though the smooth inner wall had been fitted with somewhat flattened or-gan pipes.

gan pipes.

One of several designs being studied here, the device was made even more effective by combining it with a short external barrel-like duct called an ejector.

The deep corrugations on the inside wall of the exhaust nozzle serve to break up the flow of the exhaust gasses and slow them down.

Sanders said this device cut the engine's noise to only one sixteenth that of the con-ventional engine.



The gadget mounted on the J-47 engine of this B-47 Stratojet cut down

the typical jet roar, but experimenters said the device is not yet practical.



FUEL RESEARCH—Major Lewis Lab ef-fort of NACA in past decade has been re-search in high energy fuels for all aircraft.

CLEVELAND PRESS October 9, 1957

Predict Manned Planes at 25,000 MPH

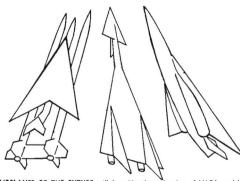
By CHARLES TRACY
Aviation Editor
Evolution of the manned
strplane from the fading turbojet to the 25,000-mph rocket
was foretold today by scientists at the NACA Air Lab,
Cleveland Hopkins Airport.
They saw the end of the
development of the tur-bojet
engine within a few years, addressing 2000 of the nation's
aeronautical experts, here this
week to inspect Lewis laboratories.

The present turbojet fight-er plane has nearly reached the peak of its development today, the aviation scientists agreed.

Partiaments during the

agreed.
Refinements during the next few years will speed it up 2600 mph but that's about the limit for pure jet engines.

Ramjets Next
Ramjets will follow the turbojets and speed planes from 2600 to 4600 mph.



AIRPLANES OF THE FUTURE will look like these sketches of NACA models. The ramjet (left) will succeed the turbojet (right), then comes the hypersonic rocket (center) crusing at 25,000 mph.

bojets and speed planes from Getting its power and its mame from air rammed into its front end, the ramjet has a moving parts, become some powerful the faster it goes.

But the ramjet doesn't start forest. But the ramjet doesn't start forest. But the ramjet doesn't start forest. But the ramjet doesn't start in air regions of the earth.

NACA experts suggest twin forest forest forest will prove the SDO mph. It must fly with air regions of the earth.

NACA experts suggest twin forest fores

Continuing up the corridor drops back into the earth's of continuous flight, scientists atmosphere. The plane and

KING-SIZE **PROPULSION**

Earlier this month, in Cleveland, Earlier this month, in Cleveland, Ohio, the National Advisory Committee for Aeronautics played host to a group of the Nation's leading scientists, educators and writers at their Lewis Laboratory.

One of three NACA labs which are opened for a select audience on a rotating triennial basis, Lewis Lab is concerned directly with experi-mental propulsion systems—turbo-jet, ramjet, rocket, and nuclear. Operational altitudes of 150 miles

and speeds in excess of 25,000 m.p.h. were among future possibilities discussed during the tour. Strangely enough, contrary to the majority of enough, contrary to the majority of present day opinion concerning a declining need for aircrews, these figures were advanced for manued space satellites.

Among top attention-getters was a detailed discussion of high energy aircraft fuels ranging from boron hydrogen compounds, reputed to possess nearly double the thrust capabilities of present fuels, to ionic drive. Power sources included solar energy and nuclear reaction.

Closer to the realm of present day working hardware was the talk cen-tering around turbojet engines. Previous conclusions had placed their top speed capabilities at Mach 2.5. Lewis scientists now express confidence that a Mach 4 (2,600 m.p.h. at altitude) powerplant could successfully be constructed and placed in operation.

The Lewis Flight Propulsion Lab, named for the late Dr. George Lewis, first director of NACA, was established in 1941 adjacent to Hopkins Airport. It boasts a staff of 2,700 professional and technical employand is headed by Dr. Edward R. Sharp.



A small wheel is jet-operated by m



High energy fuels are burned in combustion test apparatus at Lewis.



retical Mach 4 turbojet, based on a composite of advanced ideas.

CLEVELAND PLAIN DEALER October 13, 1957



Today's Siftings Shape Tomorrow's Realities

2,500 Tour NACA On 4-Day Inspection

By NORMA HIGGINS

James "Jimmy" Doolittle and Edward V. "Capt. Eddie" Rickenbacker were among the 2,500 persons who toured the Lewis Flight Propulsion Laboratory (NACA) during the triennial four-day inspection last week.

Eddie" Rickenbacker were among the 2,500 persons who toured the Lewis Flight Propulsion Laboratory (NACA) during the triennial four-day inspection last week.

During a coffee break in the morning tour Deolittle commented that the Russian's earth circle morning tour Deolittle commented that the Russian's earth circle morning tour Deolittle commented that the Russian's earth circle morning tour complements of the properse. "he said because American scientists were not first, but most because it will "shake us loose from our complementarily of NACA scientists and the growth and the stone of the ston

inoise indicators) both with and without suppression equipment. The importance of the results of these tests are awaited by commercial airlines as well as government installations. The roar of jets cut down scientifically is of great important airlined, especially as a "good neighbor" policy. The high-melting point metals, so important to the aircraft of tomorrow when speed many

Iteeted for durability and enterance.
The necessity of using heavy
shields in planes using nuclear
fuel, because of radiation. Is another problem to be overcome.
Scientists are now working on a
light, but effective shield that
will allow for arlequate or increased their or pay loads on air carriers, and yet allow the use of

U. S. NEWS & WORLD REPORT, Oct. 18, 1957

Now Science Has a Way To Cut Noise of Jets

To Cut Noise of Jets

CLEVELAND—Earsplitting noise may cease to be a feature of the jet plane if a new device fulfills its promise.

The noise suppressor was for a B-47 some stat week by the National Advisory Committee for Aeronautics. The device is a sort of corrugated exhaust pipe with a special "ejector" that slows down the flow of exhaust gases.

NACA scientists at the Lewis Flight Propulsion Laboratory said the suppressore and existing the suppressore the series of t

sor cut noise by 13 decibels—or the dif-ference between the noise made by a jet plane and the noise of a propeller-driven Constellation.

ARMY NAVY AIR FORCE JOURNAL October 12, 1957

NACA Scientists Cite Mach 4 Speeds

High-flying turbojet-propelled aircraft traveling it speeds up to four times the velocity of sound—about 2000 mph at altitude—are envisioned by scientists of the National Advisory Committee for Aeronautics' Lewis Flight Propulsion Laboratory at Cleveland, Ohio. At the NACA Triennial Inspection this week Lewis scientists discussed research which indicates that the turbojet has Mach 4 speeds of the NACA Triennial Committee for the NACA Triennial Inspection this week Lewis scientists discussed research which indicates that the turbojet has Mach 4 speeds of the NACA Triennial Committee of the NACA Triennial Committee of the NACA Triennial Inspection that the NACA Triennial Section 10 to 10 t

indicates that the turbojet has Mach 4 speed capabilities.

Formidable obstacles stand in the way of actual design and development of a turbojet capable of nearing the hyper-who have conducted extensive research on aircraft propulsion systems and their components are confident that a Mach 4 turbojet can be built.

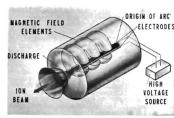
During the 15 years since the turbojet engine was first flown in the United States, significant progress has been made in its development. The United aircraft capable of sustained flight above twice the speed of sound. Even faster airplanes powered by turbojets are in the experimental stage.

A large portion of the Lewis research

planes powered by turbojets are in the experimental stage.

A large portion of the Lewis research effort has been expended on the study of turbojet engine problems. In comparison to the comparison tof the comparison to the comparison to the comparison to the compa

Considerable Mach 4 inlet and exhaust Considerable Mach 4 inlet and exhaust nozzle research is in progress at Lewis, including a special program to determine the best principles for designing inlets having high pressure recovery and low drag over a range of flight speeds up to Mach 4. Several research models devoted to the study of the complex inlet problem were shown during the Inspection.





FIRST NACA study model of an ionic propulsion unit drives a small turbine (right) to show it is producing thrust (about 1/4,200 ib.). Unit is operating inside an evacuated bell iar. Schematic of its operation is shown at left.

How NACA Views Future for Propulsion

By J. S. Butz, Jr.

Cleveland, Ohio-Present problems and future development possibilities of aircraft and space propulsion systems—including Mach 4 turbojets and ionic engines-were discussed here during the Triemial Inspection of the NACA's Lewis Flight Propulsion Laboratory. The ionic engine, a low-thrust powerplant for use in space, is a fundamental investigation and the project is in its infancy. But it has the same long range purpose as all NACA work: to provide basic data which can be applied to actual design problems be the nation's engineers.

Other activity in progress at the Lewis Laboratory:

"Turbojet operation up to Mach 4.

"Hypersonic propulsion (above Mach 5).

- High temperature materials.
 High energy fuel-oxidant combinations for rocket engines.
 High energy fuels for air-breathing
- Atomic powerplants for aircraft.
 Turbojet noise.

Mach 4 Turbojet

NACA believes that the speed capability of turbojets, once thought to be limited to subsonic levels operationally, can be pushed to about four times the speed of sound-about 2,600 mph. at altitude. This speed, NACA believes, may be the ultimate capability of turbojets.

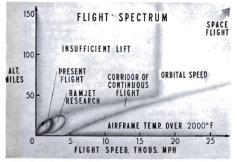
leves, may be the ultimate capability of turbojets.

This Mach 4 engine will differ significantly from today's turbojets. Its inlet and exhaust systems will have to have variable geometry to maintain acceptable efficiency over its whole operating range.

ating range.

The engine's effective thrust can be severely limited by pressure losses in

MACH 4 turbojet model reflects NACA studies. It has variable geometry inlet and exit, three stage compressor, and very short combustion chamber and afterburner.



NACA estimates that aircraft will have to stay in the continuous flight corridor (above) to be aerodynamically and economically efficient. Flight above the corridor requires very large engines and wings. Structural cooling problems are too prominent below the corridor.

AVIATION WEEK, October 21, 1957

AVIATION WEEK, October 21, 1957

be zero. However, this does not mean that the engine can no longer produce Weight Problem The ionic rocket

thrust.

At a flight speed of 6,300 mph, all of the air entering the combustion chamber has been heated to it dissociation temperature. The energy released to the through the bonds holding the atoms of the various gas molecules together. The only as molecules together. The only are to recombined in the dissociation process, which splits the bonds holding the atoms of the various gas molecules together. The only way to recover the fuel energy consumed by dissociation and make it produce usable thrust is to have the atoms recombine into their molecules within the engine. Recombination takes place when the air is expanded and its temperature lowered.

Noxile Geometry

The proper length and geometry for a nozzle with maximum recombination the proper length and geometry for a nozzle with maximum recombination the proper length and geometry for a nozzle with maximum recombination the proper length and geometry for a nozzle with maximum recombination that the proper length and geometry for a nozzle with maximum recombination that the proper length and geometry for a nozzle with maximum recombination that the proper length and geometry for a nozzle with maximum recombination that the proper length and geometry for a nozzle with maximum recombination that the proper length and geometry for a nozzle with maximum recombination that the proper length and geometry for a nozzle with maximum recombination that the proper length and geometry for a nozzle with maximum recombination that the proper length and geometry for a nozzle with such as the proper length and geometry for a construction of the proper length and prope

The proper length and geometry for a nozzle with maximum recombination bencafits are being studied by the NACA. Dissociation losses begin to be a major consideration in ramjet and turbojet design just above Mach 2. They can amount to a 50% loss in thrust at about 5000 mph.

greatly increases travel times in space, but very great velocity changes can be effected and control problems of the space ship's trajectory would be minimized because of the long burning times. Some figures on the low-thrust, long-burning time approach were given in an NACA example. If a 10 ton space whiche were powered by a 10 lb. thrust engine for one month, it would increase its original velocity by over 25,000 mph.

The small ionic unit demonstrated during the Lewis tour is the first one constructed at the laboratory. It produced about 1/4,200 lb. hrust, needed 1,000 volts to start, operated at about 500 volts and 5 amps. It was run in a bell jar evacuated to stimulate an altitude of 80,000 to 100,000 ft. In this way air was used as the fired. In space a

amount to a 50% loss in thrust at about 5,000 mph. Ionic propulsion studies which have referently been initiated by the NACA are intended for use on vehicles which have use on long voyages into space because of their very low fuel consumption. Primary drawbacks are heavy weight and need for electric power. Their operating cycle consists of striking an electric are in a gas. The bark is a second propulation of the propulation of

they would encounter during re-entry. The plasma jet is also a feasible low-thrust, low fuel consumption device for space propulsion.

High Temperature Materials

Progress with all types of propul-sion units is largely waiting on im-proved structural materials. Some goals of metallurgical research are to raise maximum use temperatures of: • Torbojet blade materials from about 1,650F to 2,500F. • Ramjet materials from 2,500F to 3,100F.

 Mamjet materials from 2,500° to 3,000°.
 Nuclear rocket materials to 5,000°.
 Nuclear rocket materials to 5,000°.
 One method of increasing high temperature strength of current nickel and cobalt base alloys is to suspend hardening particles in the material. These particles have roughly the same effect on the strength of the alloy as the dispersion of gravel does in cement. The effect of these high temperature hardening particles is demonstrated by dispersion of gravel does in cement. The effect of these high temperature hardening particles is demonstrated by the results achieved by suspending finely divided stable ceramic-aluminum coside in a nickel matrix. This raises the use temperature of the nickel base alloy from the current figure of about 1.650F to as high as 2.100F.

Use of a refractory powder in aluminum alloys has raised the capabilities of aluminum several hundred degrees. The use of nickel and cobalt base materials is limited, however, by their low melting points. The nickel base alloys are liquid at about 2.600F. Other nictals with higher melting points must be used as base materials for higher temperature alloys. Two metals currently being considered as base materials are columbium with a melting temperature of 4,500F.

with a melting temperature of 4,500F and tungsten which melts at 6,100F

AVIATION WEEK, October 21, 1957

Tungsten allovs can be expected to

Tungsten allovs can be expected to sustain 10,000 psi, stress at temperatures 1,700F higher than nickel alloys. The entire class of high melting point metals have the very serious problem of rapid oxidation. When heated the surface atoms of these metals slip free and combine rapidly with oxygen.

Oxidation rate is greatly increased in an atmosphere of ionized air such as would be found around a high speed missile nose cone or in an atomic engine. Rise in oxidation rate is due to the more active nature of the air atter its molecules have been dissociated into charged particles. These particles want to recombine into more stable molecules and this is done against the metal. Recombination releases heat, further raising the metal temperature and increasing oxidation still more. Metals oxidize about 400 times faster in the presence of dissociated air than in normal air. NACA research in this race to date has been to define the problem, now measures of countering this severe oxidation are being studied.

Another group of materials being investigated at Lewis are ceramics which have very high melting points and good oxidation resistance but are handicapped by brittleness.

One of the most serious effects of brittleness is susceptibility to thermal stresses which are high in many missile and atomic applications.

Careful treatment and polishing of ceramic surfaces have greatly increased their ductility. This work has essentially just begun and practical methods of recating large sections of ceramic are not available.

the air intake system, drag due to air spilling out of the inlet, and improper expansion in the exhaust nozzle. These effects can be minimized at all speeds with variable geometry systems.

The compressor section will have perhaps three rotating stages compared to 12 or 15 for some of today's engines. Ram compression in the inlet will be so high that a small pressure rise across the compressor will be sufficient. Transonic design principles have been clarified and have improved the compressor efficiency in this range. Compressor efficiency in this range conductive to the compressor of the compression of the compressor efficiency in the respective material similar to the turbine blades.

Combustion chambers and after-burners will be much shorter than present ones. Improvements in fuels as well as combustion chamber design make this possible.

make this possible.

Hypersonic Propulsion

Hypersonic Propulsion

NACA also is studying hypersonic propulsion, which involves systems for delivering continuous power to vehicles at speeds above Mach 5.

In broad terms the NACA sees such vehicles traveling in the continuous flight corridor illustrated on page 73. This corridor is the region in which very high speed, very high altitude flight is considered accodynamically and economically feasible. Flight in the area above the corridor requires greater wing area, structural weight and larger engines lowering the vehicle's range markedly. Below the corridor structural soundness of the aircraft would be jeopardized by aerodynamic heating.

The only air breathing engine presently considered for use at hypersonic

The only air breathing engine presently considered for use at hypersonic speeds is the ramjet. The NACA is studying its behavior from 2,700 to 4,600 mph. Two problems are paramount in this study. One is the cooling of engine parts, and other is conversion of the energy released by burning fuel into useful thrust.

Ramjet Cooling

Ramjet Cooling problem is illustrated by conditions in the ramjet at a forward speed of 3-4,000 mph. Temperature in the combustion chamber is about 5,000F which is above the melting point of most of today's metals. NACA studies of cooling these internal parts show that the best way is to use fuel and then burn it in the combustor.

The other problem of producing thrust from the burning fuel grows more complicated as velocity increases. Below Mach 2 thrust in a jet engine is generated by increasing the temperature and consequently the velocity of the air passing through the engine.

the air passing through the engine.
NACA estimates that at a flight speed
of 6,300 mph. the temperature rise
that the fuel can impart to the air will

High Energy Propellants

High Energy Propellunts

NACA's rocket research effort is almost completely centered about high energy fuels and evidizers. Experimentation includes the most high reactive and dangerous substances which are potentially the best propellunts.

Practical studies are being made of the difficulties of using hydrogen fluorine combination. This is typical of the work at Lewis Laboratory.

Fluorine is the most powerful oxidizer known and consequently reacts readily with most materials, often violently. One problem in its use is to find material for tanks, valves and lines in the fluorine. Rocket combustion chamber and nozzle construction problems are magnified by use of fluorine.

Hydrogen is most difficult to keep in the liquid state as it boils at +423F. Well insulated tanks and lines are needed to keep the hydrogen supply system working properly.

AVIATION WEEK, October 21, 1957

Jet Engines Capable of a Speed Four Times That of Sound Are Forecast

THE NEW YORK TIMES

By RICHARD WITKIN

Times That of Sound Are Forecast

INTROVED HAAT resistant may be a second to the secon

Ion Power 'Attractive' Prospect, A-Plane Doubtful Bet, NACA Says

well be working on a similar for propulsion system.

The cocasion was the triennial inspection of the National Advisory Committee for Aeronautics' Lewis being in the propulsion Lordon, The powerplant problems, including the special servolvamics of bigs-beed flight propulsion.

NACA scientists and engineers seemed to the control of the National Advisory Committee for Aeronautics' Lewis being made on the nuclear free powerplant problems, including the special servolvamics of bigs-beed flight propulsion.

NACA scientists and engineers to be being made on the nuclear free powerplant problems, including the special servolvamics of bigs-beed flight propulsion.

NACA scientists and engineers to be being made on the nuclear free powerplant problems, including the propulsion.

NACA scientists and engineers to be being made on the nuclear free powerplant problems, including the propulsion to the propulsion.

NACA scientists and engineers to be being made on the nuclear free powerplant problems, including the propulsion to the propul

special aerodynamics of high-speed flight propulsion.

NACA scientists demonstrated their working model of an ionic rocket—an "attractive" propulsion of the control of the

the ionic eordee of the space centify would be cut off, allowing the craft to coast through outer space on its own escape speed of 25,000 miles per hour.

The use of a low, 10-pound thrust engine over a period of a month period of a period of a period over a period of a period over a period of a period over a p

powered aircraft.

The main obstacles are the weight problem of the shielding which must be provided to protect the crew from radiation. Very little payload is left after the weight of needtor and its shielding is allowed for.

is allowed for.

Dr. Silverstein wasn't sure that the country would want a nuclear-powered sucreaft. He said the building of a nuclear aircraft building of a nuclear aircraft cut offense policy-which type of plane is needed airch thou much money the nation is willing to spend to develop and build the plane.

size, with an invisible siteam of the manner of the manner

government's serious interest in the technique.

Outlining some of the MacA's research on engines "capable of operating in a vacuum" beyond the earth's atmosphere, Eugene J. Manganiello described the current Russian astellite and the futuristic "engines," NACA made these disclosures at a tour of its S100 million laboratory here for government, industry and scientific groups:

1. Ramjet test missiles recently lave made flights with light energy fuels which may conclude the serious proposed to the control place and missiles by as much as 40 percent.

The fuels are boron-hydrogen compounds or boranes, which can be made in liquid form: At present they cost several bundred dollars a pound, but production plants now are beyones and missiles recently lave fuels they conclude the production plants now are between such fuels become practical for military aircraft.

2. Research now indicates

when such fuels become practical for military aircraft.

2. Research now indicates that jet engines can be made to power planes at 2600 miles an hour, or four times the speed of sound, whereas it once was thought their limit would be below sonic speed.

In discussing ionic propul sion, the NACA said that ac

sion, the NACA said that ac-celerating lons to high speeds would provide useful energy for satellites, and added: "Development and improved efficiency of such a propulsion device may lead to its applica-tion in flight beyond the at-mosphere."

and members of Congress who participated in the Triennial Inspection of the Lewis Flight Pro-pulsion Laboratory in Cleveland last week most certainly went away more impressed than ever before with the vital importance of giving the utmost support to our research organizations and Shown at this laboratory of the National Advisory Committee for Aeronautics were such promising projects as research into highsuch promising projects as research flowing, energy fuels promising increases in the range of supersonic aircraft and missiles by as much as 40 per cent, new fuel-oxidant combinations which could vastly advance rocket performances, new engine materials which could retain high strength

at the extreme temperatures general in hyper-sonic flight, and many other fields significant to Each of the new fields into which research scientists are now just peeking eventually will be entered and conquered. The nation that first reaps the benefits of such explorations will have a definite advantage. Russia's recent accomplishments leave no doubt of the efficacy of her research which, unlike ours, is unhampered by budgetary

Two years ago Dr. Jerome C. Hunsaker, then Two years ago Dr. Jerome C. Hunsaker, then NACA chairman, reported that "The current trend toward leveling off expenditures for scientific research in aeronautics is forcing hard decisions to slow down or to defer indefinitely research projects essential to the timely development of new weapons." His warning went unheaded. The advice he gave then needs repeating today with emphasis—"It is now wise to accelerate scientific progress. In the long run scient erate scientific progress. In the long run, scien-tific research is the best insurance that there will 'value received' from the country's whole aircraft program."

Confidence in our future and its security depends in great measure upon the support in money and authority which we are willing to give to re-search organizations such as the National Advisory Committee for Aeronautics.

> NEWSWEEK October 21, 1957

Beyord the 'Ultimate'

Beyon'd the 'Ultimate'

For fifteen years jet-propulsion engineers have been squeezing more and more speed out of the turbojet engine. But at Mach. 25 (about 1,960 miles per hour) it was thought that it had reached its ultimate speed. Last week the National Advisory Committee for Aeronautics anonunced in Cleveland that a still speedier future lies ahead for the turbojet. Their latest studies, said the committee, indicate that the turbojet-with vital changes-is capable of propelling an aircraft four times the speed of sound, or about 2,600 mgh.

The most crucial rearrangement of the turbojet's physique will not be in its hot, whining insides but in its inlet and exhaust nozzles, usually the concern of the engine, will be enlarged. Their size will be variable in flight, so the engine can adapt itself to the wide range of speeds and altitudes a Mach 4 airplane will encounter before reaching "design flight conditions" at 85,000 feet.

N.Y. World Telegram and Sun Oct 8, 1957

lons May Hold Space Travel Key

land this week.

"If a stream of ions, as in an arc jet, can be accelerated to high velocities by use of electric or magnetic fields, a small amount of thrust will be produced." Jons are tiny bits of matter—atoms which have been stripped of some of their

CLEVELAND, Oct. 8.—The electrically charged.
National Advisory Committee of Arman Committee of Arman Committee of Section 1 of Aeronautics bombarded a small pinwheel inside a glass earth's atmosphere a vehicle would require only slight powers of the National Principle.
The same and small committee of the National Complete of the National Committee of the

Turbojets Are Theoretically Capable of Mach 4 Speeds

By J. S. Butz, Jr.

Cleveland, Ohio—Extension of the useful speed of turbojet engines to Mach 4 will provide a much greater latitude in the design of future aircraft than has been expected.

Widening the choice of powerplants will make it easier to optimize 2,600 mph, aircraft for a definite mission re-

mph, aircraft for a definite mission requiring relatively great range, or speed, or some other specific performance.

National Advisory Committee for Aeronautics scientists at the Lewis Flight Propulsion Laboratory here believe that current aerodynamic and combustion knowledge indicates that Mach 4 turbojets are feasible. They point out, however, that practical construction of such an engine will require a major research and development effort mainly because of the high temperature performance required of engine components. Mach 2.5 is about the upper limit for present-day turbine engines.

Types of Powerplants

If the Mach 4 turbojet is developed, the powerplants available at that speed will also include turbojets, ramjets, dual-cycle engines, rockets and various combinations of these. The turbojet would

cycle engmes, rockets and various combinations of these. The turbojet would retain the advantage it has today. It would be the smallest, least complicated engine installation capable of delivering continuous power to an aircraft through its entire flight. Each of the other engine types will undoubtedly be superior in this speed range for certain missions. NACA sees Mach 4 as the probable ultimate speed of the turbojet. Ramjets will have rather clear advantages among air-breathing engines above that speed even though they cannot function during landing, takeoff and initial climb. Overall problem of selecting a power-plant involves many interdependent factors concerning the aerodynamic and structural design of the aircraft as well as the engine. Since it is impossible to separate these considerations, engine studies at Lewis Laboratory include the mutual interaction of the power-plant installation on the airframe.

Hypothetical Cose

Hypothetical Case

Hypothetical Case
A generalized example of such a study
was presented during a recent Triennial
Inspection of the Lewis Laboratory.
Thrust requirements of a Mach 4 turbojet over its whole speed range were determined by considering the drag of a
hypothetical aircraft representing the
most advanced aerodynamic and structural thinking. The aircraft was designed
for a speed of Mach 4 at about 85,000
fr. Adequate thrust was stated to be
35% more than the drag, at every speed.

This insured good acceleration throughout the speed range. With a smaller
thrust margin, excessive amounts of
fuel would be consumed during acceleration and the aircraft's range would
be seriously reduced.

Main value of this example study was
to show the importance of the engine's
inlet and exhaust nozzle design. The
conclusion is that such systems must
have variable geometry if the engine is
to deliver adequate thrust during the
whole flight. Even if the basic engine
is capable of delivering much more
thrust than the design requirement, it
can be rendered inadequate by a poor
installation. Thrust tosses of the installation are primarily from pressure losses
and the conditions and shock or overexpansion losses in the exhaust nozzle.

Example study is illustrated on pp.
86-87. Upper figure shows a fixed

geometry installation meeting the design conditions of Mach 4 speed at 85,000 ft. The thrust-available curve of 000 ft. The thrust-available curve of this installation superimposed on the thrust-required curve of the aircraft shows that it would be impossible for the airplane to ever accelerate to Mach 4 because thrust is inadequate at the intermediate speeds.

Inadequate Thrust Cause

This inadequate thrust around Mach 2 is primarily due to two things:

• High drag caused by air spilling around inlet. Engine requires about 66% less air at Mach 2 than it does at Mach 4. Excess air is deflected around

inlet.

Overexpansion of air in exhaust noz-zle, causing strong shock waves which slow down exit velocity. Exit area of nozzle, which is correct for Mach 4, is too large for Mach 2.

The next figure shows that using a variable geometry nozzle will still not correct the thrust deficiency around Mach 2, although it will bring subsonic thrust up to the prescribed 35% increase over aircraft drag.

When variable geometry is added to the inlet system the engine and its installation are able to produce the necessary thrust over the whole speed range, as shown in the lower figure.

Variable geometry mechanism of this inlet system would have to include means of varing the angle of the ramps

means of varying the angle of the ramps ahead of the duct as well as an efficient by-pass to bleed excess air in the duct to the outside airstream.

Inlet Function

% OF

Inlet system shown in this example is but one of those currently being investigated by NACA, other research agencies and aircraft manufacturers. Function of these inlet systems is to slow and compress air entering the air-craft. Air should enter the turbojet's compressors at less than half the speed of sound.

of sound.

Ideal inlet system accomplishes this diffusion with a minimum loss of total pressure in the air stream and with minimum drag. Both of these losses serve to decrease the effective thrust

of the engine.

There are generally two methods for slowing a supersonic air stream. One is an external diffusion system and the previous example was an external system. Both class the previous example was an external system. previous example was an external sys-tem. Both slow the supersonic flow to a low supersonic Mach number by con-tracting the air stream and then diffus-ing it to subsonic speed by means of a normal shock wave.

External system generally uses a series of ramps for cones if the inlet is circu-larly which create oblique shock waves to slow the air; then the normal shock is located across the mouth of the inlet



INSTALLATION above is typical of inlet and exhaust systems for Mach 4 turbojets. It has fixed geometry designed for Mach 4, but so dissipates ~ ver at the intermediate speeds that typical aircraft would not be able to exceed Mach 1.5 (..ght).





Large nozzle expansion ratio is not needed below M sufficient to make the installation acceptable (right).



FURTHER ADDITION of a variable geometry inlet gives the installation adequate thr over the entire speed range. Inlet shown here is of the external compression type. Num of possible solutions to the variable geometry problem are being investigated.

AVIATION WEEK, October 28, 1957



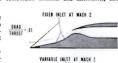
NACA scientist holds model of Mach 4 turbojet. Model has cone type variable geometry

some current engines. This is possible because pressure rise per stage has been raised and because much of the com-pression is accomplished in the inlet duct instead of the compressor. Better fuels as well as improved com-bustion design will greatly shorten com-bustion chambers and afterburner.

Temperature Comparison

NACA predictions of the tempera-ture within the Mach 4 turbojet illumi-nate some of the development problems facing the engine. For comparison, the temperatures in a typical Mach 2 en-gine will be given first. • Compressor inlet—240F • Entrance to combustor—875F • Turbine inlet—1,650F • Turbine outlet—1,000F • Discharge exhaust—3,000F

- Discharge exhaust—3,000F
 On the Mach + turbojet these same
- temperatures will usually be:
 Compressor inlet—1.250F
 Entrance to combustor—1.330F



TOP sketch shows Mach 4 inlet in Mach 2 flow. Air spills around inlet causing high drag. If spillage air is exhausted behind the inlet (below) drag is reduced.

AVIATION WEEK, October 28, 1957

• Turbine inlct-1,450F • Turbine outlet-1,350F • Discharge exhaust-3,300F

Temperature Difficulties

Temperature Difficulties
Most serious temperature problems
will be encountered in the compressor
and the main engine bearings and scals.
The compressor will require the use
of the most advanced allows now used
in turbines to withstand the combination of high temperature and high stress.
The turbine on the Mach 4 engine is
operated below the turbine temperature of the Mach 2 engine because the
stresses in the turbine have been raised
to match the aerodynamic capabilities
of the compressor.
Environmental temperature of bear-

to match the aerodynamic capabilities of the compressor.

Environmental temperature of bearings and sals in the Mach 4 engine will be above present limits of such materials because of the heating of the ram air as well as temperatures within the evele.

NACA studies indicate that the ramiest will be definitely surpers to the burs.

jet will be definitely superior to the tur-bojet above Mach 4. This conclusion is bojet above Mach 4. This conclusion is based on the fact that around Mach 3 the turbine begins to extract more pressure from the cycle than the compressor puts in. This detrimental effect of the turbine-compressor unit on the thrust increases with speed. The ramjet obviously doesn't have this handleap and is more efficient above Mach 3. In terms of a typical military missions

and is more efficient above Mach 3. In terms of a typical military mission however, the turbojet is still competitive up to Mach 4. Using ramjets, the air-craft cruising at Mach 4 can be much smaller than an aircraft with similar range using turbojets. However, the smaller ramjet-powered plane would meed some sort of boost arrangement to take off. The weight of this booster would bring the total weight of the would bring the total weight of the ramjet aircraft up to that of a turbojet version, if rockets were used for the

m the ideal case. Air velocity becomes subsonic across this shock and enters the duct at less than Mach. 1.

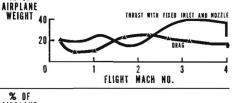
Internal System

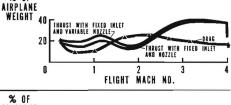
Internal System

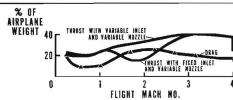
With the internal system, air enters the duct at supersonic velocity. The duct contracts to a throat where normal shock occurs and the stream becomes subsonic. Duct then expands to slow the subsonic stream.

Trouble arises with the external system when the inlet is too large and more air is taken into the duct than the engine needs. The excess air then spills around the inlet causing a large drag. If the inlet area is too small, the engine sucks the normal shock at the inlet back down the duct causing a decrease in flow density and an increase in pressure loss. Very accurate and sensitive controls are needed to keep the shock pattern and inlet area adjusted so that neither of these two excessive losses will occur.

Trouble with internal systems Trouble with internal systems centers around making the normal shock wave start at the inlet and move down the duct to the threat. This movement of the normal shock depends on the ratio of the inlet area to the threat area. The pressure recovery also depends on this area ratio. Unfortunately, a minimum pressure loss and high duct efficiency demands such a small throat area in relation to the inlet area that the normal shock won't move down the area in relation to the inlet area that the normal shock won't move down the duet to the threat. It stands out in front of the duet, creating a large drag. If the threat is opened up to allow the normal shock to move back to the threat, then a loss in total pressure must be accepted. This type of internal system is also very sensitive to back pressure. If the engine surges or in any way creates a sudden back pressure, the shock may pop out of the duet. Starting the shock back up the duet requires the proper combination of Mach num-







AVIATION WEEK October 28, 1957

ber, back pressure, etc. Sometimes an unstable flow is set up that is difficult to stop.

Corrections Devised

These problems of internal diffusers have been largely corrected. The forward or supersonic portion of the diffusers have been perforated, making the duets much easier to start and much less susceptible to back pressure. Variable area throats make the internal duets capable of operating over a wide Mach number range. Both internal and external services appeared to present a control of the property of the proper

a wide Mach number range. Both in-ternal and external systems are being developed and have their enthusiasts. In addition to having a more elabo-rate form of variable inlet and exhaust nozzle than current turbojets, the Mach 4 engine will also have a different de-sign in almost every major component. The compressor will have about three stages instead of 12 to 15 that are on

AMERICAN AVIATION

NACA studies new ways of cutting jet noise



SPECIAL EXHAUST SHAPES developed by NACA attenuate noise level of jet engines Experimental shape shown brought the 121-decibel level of J47 to 107 decibels

Experimental shape shown brought the Ways to attenuate jet engine noise are being intensively studied by National Advisory Committee for Aeronautics engineers at the Lewis Flight Propulsion Laboratory in Cleveland. Originally, it was suspected that the "big" noise of jet engines comes from the combustion of fuels at tremendous rates, and the windage of the whirling compressor blades. It was found, though, that the largest noise found, though, that the largest noise it is timites turbulently with the atmosphere. Fluctuating pressure waves are created, which result in a broad band of acoustic waves.

are created, which result in a broad band of acoustic waves. One way to decrease the noise is to lessen turbulent mixing of the jet stream, or the size and strength of the eddies. Various nozzle shapes have been studied which give substantial re-ductions in the noise level. However, the design changes needed usually bring locreased sirplane drag, engine weight

170

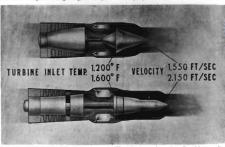
and decreased engine performance.

and decreased engine performance.

Another approach to the problem is oreduce the velocity of the exhaust in the problem of th

more compact.

If a special nozzle shape were combined with an engine having a relatively slow-speed jet exhaust, MACA engineers (sel that the target 15-decibel drop would be obtained. This 15-decibel figure is based on an average jet noise of 116 decibels as compared with propeller noise of 101 decibes.



LOWER EXHAUST VELOCITIES can significantly lessen jet noises. Resulting smaller turbine inlet temperature brings about a more compact engine design.



WIND TUNNEL TEST: Scientist checks an experimental aircraft model in a supersonic wind tunnel at Lewis Flight Propulsion Laboratory in Cleveland. Air stream can be whipped up to 31/2 times the speed of sound.

Heat Blocks Way To Faster Aircraft

Engineers are making slow but steady progress toward a 2600 mph airplane.

The primary problem: Finding a suitable high-temp material. ALERICAN AVIATION, OCTOBER 21, 1957

NACA reveals details of Mach 4 turbojet

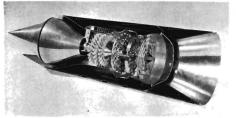
SUPERSONIC turbojet engines capable of sustaining flight at Mach 4 are within our reach, according to engineers at the Lewis Flight Propulsion neers at the Lewis Flight Propulsion mittee for Aeronaulics. A cuttoway view of a model of such an engine, which could power aircraft cruising at 2,600 miles an hour and 85,000 ft. altitude, was displayed at NACA's triennial inspection of the Lewis facilities.

NACA engineers say the Mach 4 turbojet is the ultimate capability for this type engine, But even this capability could not be obtained, they say, unless the engine is equipped with variances.

Here, the afterburner is the thrust-generating element. At Mach 4, then, the turbojet engine approximates a ramjet engine. Above Mach 4, the pure ramjet engine has the field to itself in-sofar as air-breathing powerplants are concerned.

sofar as air-breathing powerplants are concerned.

NACA engineers are confident that the proposed powerplant can be built. They say that if the engine is operated at its designed speed, it will have a thermal efficiency of about 42%. This compares with the 28% efficiency of a Mach 2 engine, and the 33% efficiency characteristic of steam-electric stations.



CUTAWAY VIEW of model of possible. Mach 4 turbojet engine shows three-stage compressor and small combustor chambers, which result in a highly compact powerplant. The efficiency is the result of an extremely high pressure in the engine cycle. The expansion ratio, based upon internal and ambient pressures, is over 100.

showing designs for aircraft which will go 2600 mph hour. Yet they

are pinned down by the heat barrier where progress is still measured in

Some of its awesome problems

were outlined last week at a trien-nial inspection of the National Ad-

visory Committee for Aeronautics' \$100 million Lewis Flight Propul-

years since the first supersonic flight by Capt. Charles Yeager in

the Bell X-1, designers have had a net gain of only 300°F to work

with, spokesmen said. They cur-rently have turbine blading which

will withstand 1650° without los-ing necessary strength. This is up from 1350° in 1947. But in the next 10 years they are hoping for

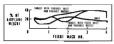
1850° to 2500° materials for gas

-In the 10

sion Laboratory.

Progress Review -

OCTOBER 21, 1957



AIRPLANE DRAG characteristics call for variable inlet and exhaust nozzles to achieve speeds over Mach 1.4.

Mach I because of over-expansion of the exhaust gases. Here, the efficiency of the exhaust pases. Here, the efficiency of the 97% efficiency for a variable exhaust nozzle.

Although the aerodynamic combustion principles are being satisfactorily worked out, severe practical problems exist because of the high temperature in which the engine components must operate. The inlet temperature is pegged at 1,240 F, which makes the properation of the properation of the properation of the compressor is so high that it will need alloys currently being used in turbine blades.

m turbine blades.

Development of engine bearings
and seals comprise other design problems that must be met by the fabricators of this engine, according to
NACA.



TYPICAL ENGINE POD is slung under wing and next to fuselage. Note large exhaust nozzle needed for Mach 4 operation. Air inlet ramp is at right.

turbines. For ramjets they need 2500-3100° and for a nuclear powered rocket, 5000° material. Aircraft and rocket engineers in Cleveland aren't bashful about

Possibilities are nickel with sintered aluminum alloys, columbium and tungsten. Ceramics hold some Lewis research men have been able to impart a measure of ductility to ceramics by eliminating surface imperfections by water polishing. Ductile ceramics developed so far are magnesium chloride and sodium chloride.

Fuel Problems - The materials problem is the most pressing at Lewis, but there are others. With high energy fuels already tested, engineers hope to extend the range of supersonic aircraft and missiles by 40 pct. Liquid fuels containing boron are the most promising so far, but excessive deposition cuts efficiency.

Atomic energy powered aircraft seems still a long way off. The weight of pilot shielding necessary is still the top problem. Engineers are chipping away at it through better shaping, split shielding and searching for new materials. De-composition of the metal through radioactivity is still another hurdle

 Mechanical supposed teleshapes, etc.).
 Eagine design.
 Climb procedures.
 This NACA work is devoted solely to decreasing noise during takeoff and much of it involves the study and extension of ideas originating with various with the control of th much of it involves the study and exten-sion of ideas originating with various manufacturers and research groups here and abroad.

Main Drawback

Main Vrewback
Many types of mechanical supressors have been tried and their main drawback has been that they reduce engine thrust while cutting noise. To the airlines this could mean a revenue loss through reduced passenger loads.
However, the NACA has a nozzle-cjector combination that reduces noise to the required level without the customary thrust losses. The only drawback of this installation is that the cjecture will increase the draw of the aircraft.

back of this installation is that the ejec-ter will increase the drag of the aircraft and cut its range down.

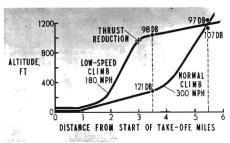
The whole project is in the experi-mental stage and efforts are now being made to make the ejector both retract-

able and light.

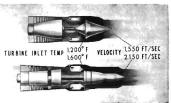
Current study of this nozzle-ejector has given the NACA considerable hope that a workable answer to the noise situation will be reached through a relatively simple mechanical device. Much

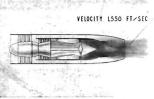


CORRUGATED nozzle in combination with annular sheet metal ejector is most promising noise suppressor being studied by the NACA. Ejector eliminates thrust loss due to the corrugated nozzle, but it has a large drag which would lower aircraft range.



TAKEOFF and climbing procedures especially tailored for each airport and its surrounding communities can do much to lower jet noise in populated areas. Typical climb procedure and the noise it creates on the ground (in decibels) is compared to a proposed low-speed.





JET ENGINE noise can be reduced by lowering turbine temperature and exhaust velocity. Using new design techniques, low-temperature, low-noise engines can equal the power and efficiency of today's high-temperature jets. More efficient high-temperature engines could also be built with the new design methods. Exhaust velocity and noise of by-pass engine (right) is reduced by low energy by-pass flow. By-pass engine has about same noise level as the low temperature engine (above, left).

AVIATION WEEK, November 4, 1957



EXPERIMENTAL noise suppressor is mounted on a test engine at the NACA Lewis Labora-tory. Nozzle design lowers noise level effectively but also reduces engine thrust.

climbing procedure for a jet transport and its associated noise. It also shows one of the recommended flights paths resulting from NACA test.

Noise Reduction Technique

Noise Reduction Technique

The normal technique is to take off in the jet transport, hold it down until it acceleres to about 300 kt. and then begin the man the property of t

areas.

There is the further possibility of combining all three of these general approaches to the noise problem to achieve a single, economical solution. Cost, thrust losses and weight are all Cost, intust iosses and weight are all factors of the greatest importance when considering devices and procedures to be used by commercial airlines. While effective noise reduction methods may be presently available they require much more development before they will be acceptable to the airlines.

- Increasing downstream mixing of the jet to attenuate the strength of the eddies. Increasing contact area of the jet blast with the outside air to reduce velocity
- Many nozzle shapes have been studied. One is the "organ pipe" type which passes the engine exhaust through a number of small pipes. An-other nozzle uses deeply folded convolutions or corrugations to spread the exhaust. Parallel slots have also been

exhaust Parillel slots have also been tried on nozels.

All of these types can produce the required jet noise reduction of 15 decibels to bring a four jet transport passing directly overhead at 100 ft. with full power to the noise level of a plane with four piston engines under the same conditions.

They are all faced with the same serious disadvantage however, they build up back pressure on the engine and slow its exhaust velocits, causing a thrust loss.

thrust loss.

Ejector Reduces Loss

Ejector Reduces Loss
This back pressure and thrust loss
can be eliminated while achieving the
required sound reduction by surrounding a corrugated nozzle with an ejector,
according to the NACA. The ejector is
a form of jet pump. High velocity engine exhaust air passing through the
ejector pulls low speed secondary air
around the jet blast. This reduces pressure in the jet stream and gives the
velocity shear gradient surrounding it

Quiet vs. Efficiency

Quiet vs. Efficiency

Main arguing point against such low
temperature engines is the fact that the
improved design techniques could be
used to produce a more efficient engine
with no reduction in noise. The low
temperature engine is also slightly
larger then its noisier counterpart and
would thereby tend to increase aircraft
drag in some installations.

One current type of turbojet, which
is as efficient and as quiet as the low
temperature engine, is the by-pass
engine. Noise reduction on this powerplant is achieved in much the same

engine. Noise reduction on this power-plant is achieved in much the same manner as the nozzle-ejector combina-tion except that the mixing of the low energy secondary flow with the high speed jet exhaust takes place within the engine. The air which is ducted around the combustion chamber, tur-bing and part of the sevents takes. around the combustion chamber, tur-bine and part of the compressor on the by-pass engine is mixed with the main jet in the tall pipe. This gives the resultant jet exhaust a lower velocity and less noise. Velocity is especially reduced in the shear area between the jet and the outside air.

The third method that the NACA considering for method that the PACA

is considering for reducing noise is by varying the aircraft flight technique. Chart on page 73 shows the normal

AVIATION WEEK, November 4, 1957

Chicago Baily Oribune

Scientists Predict Ionic Beam Engine May Carry Man on Flights in Space

said.

Manganiello spoke at the opening of the inspection of the the Spoke in the 150 million dollar laboratory, there. The inspection is ethocked were three years to acquaint the aircraft industry and flight scientists with latest engline developments.

Abe Silversein, associate dearth grayity pull, it will be director of the laboratory, told quire a smaller "putt-putt"

BY LLOYD NORMAN

Cleveland, Oct. 7 — Man reporters that he could not lengine to keep it flying in should be able to leap 1,000 forecast when the United frictionless space. This enmiles into outer space in a States could attempt manned gine. Silverstein said, any rockef powered ship and stay space flight. He said this aclake years to develop. It will

unthere for months with the seemplishment, would are.

pound thrust ionic engine could boost a 10 ton vehicle already traveling at gravity recket might weigh more than an hour to 50,000 miles an hour for a month, experts said.

They estimated that such a rocket might weigh more than 10,000 tons, us in g current some content of the content of th

miles into outer space in a rockef powered ship and states could attempt manned prockef powered ship and state paper flight. He said this act up there for months with the felp of "ionic beam" engines, government-flight research scientists said today. Eugenel, Manganiello, assistant director of the Lewis flight propulsion laborator, the said the Russian launch ingist propulsion laborator, of the national advisory committee for aeronautics, soit that "sustained statellite" flight mould be the next state that soil that soviet aviation research was abead of the American flight mould be the next shallite and rise and rise at the launching of an earth satellite. He said the Audited States Vanguard earth statellite. He said the United States Vanguard earth be United States in the State has the more development and engineering complexities that the United States in the New Freezench who where for the said the United States and the Could States has the more development and engineering complexities that the United States has the more development and engineering complexities that the United States has the more development and engineering complexities that the United States has the more development in the United States in the State in the Earth satellite in the said the arth satellite problems in the United States has the more development and engineering complexities that the United States has the more development and engineering complexities that the United States has the more development and engineering complexities that the United States has the more development and engineering complexities that the United States has the more development and engineering complexities that the United States has the more develo