



CFE Services

Rib Stiffened Sandwich Panels using Aluminum, Silicon Carbide, and Beryllium for Lightweight Telescope Structures

B. Catanzaro, November 1, 2016

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Over Arching Goal of SBIR

Reduce the Time and Cost of Deployment of
Lightweight Telescopes

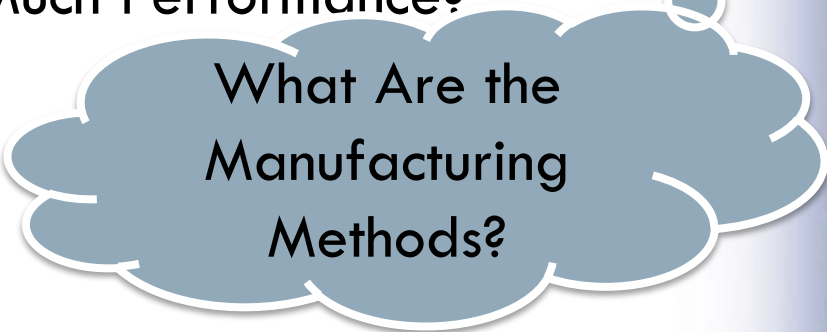
- Prior Presentations – Wavefront Coding
- Stretch Goal – Rapid Fabrication of
Lightweight Structures from Advanced
Materials

Theme...

- How Can I Gain Access to Advanced Materials?
- How Do I Reduce Cost / Schedule without Sacrificing (too) Much Performance?



What
Materials?



What Are the
Manufacturing
Methods?



What Are Some
Cost Drivers?

Which Materials?

	ρ	E	α (20C)	K	Cp
CFRC	1.78	93	0.02	35	800
ULE	2.21	67	0.03	1.31	766
Zerodur	2.53	91	0.05	1.64	821
Super Invar	8.13	148	0.3	10.5	515
Invar 36	8.05	141	1	10.4	515
SiC	3.21	456	2.3	186	680
Pyrex	2.23	63	3.3	1.13	1050
Kovar	8.35	138	5	17	439
Titanium	4.43	215	8.8	7.3	560
Beryllium	1.85	287	11.3	216	1925
SS 304	8.00	193	14.7	16.2	500

Handbook of Optomechanical Engineering, Ahmad
Opto-mechanical Systems Design, Yoder

General Processing Methodologies

CFRC

- Mandrel / Mold
- Unitape Layup
- Cure
- Machine
- Bond

SiC

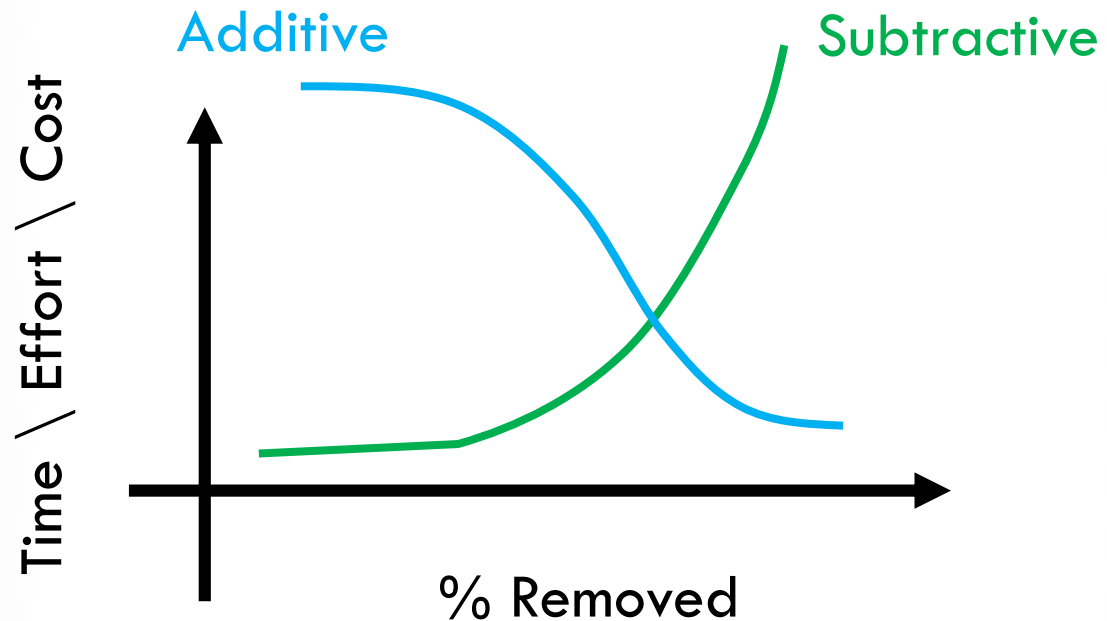
- Mandrel / Mold
- Convert
 - Greenbody + Fire
 - Deposit
 - Infiltrate
- Machine
- Braze / Bond

Beryllium

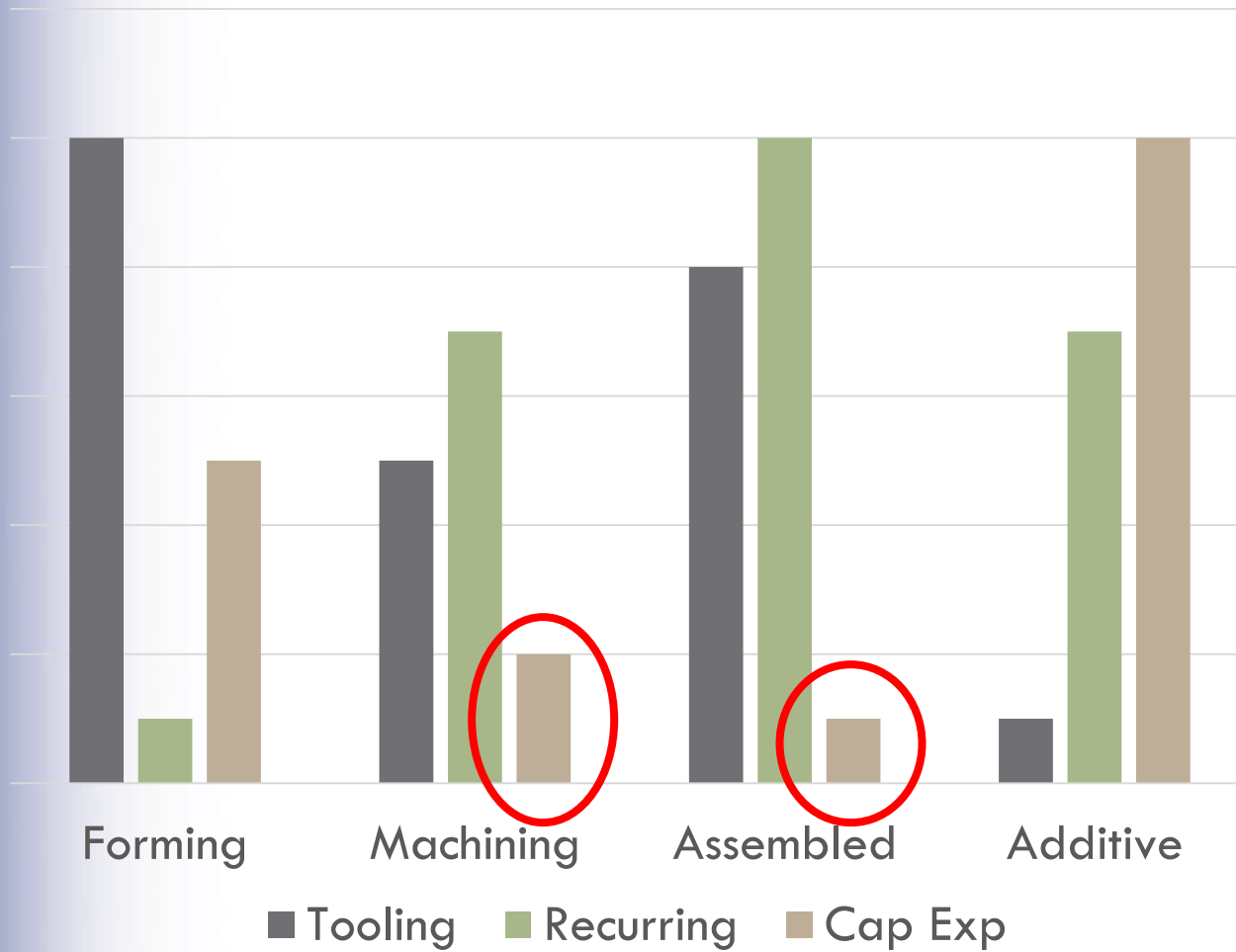
- HIP / Treat
- Machine
- Braze / Bond

Intuitive Trade-Off

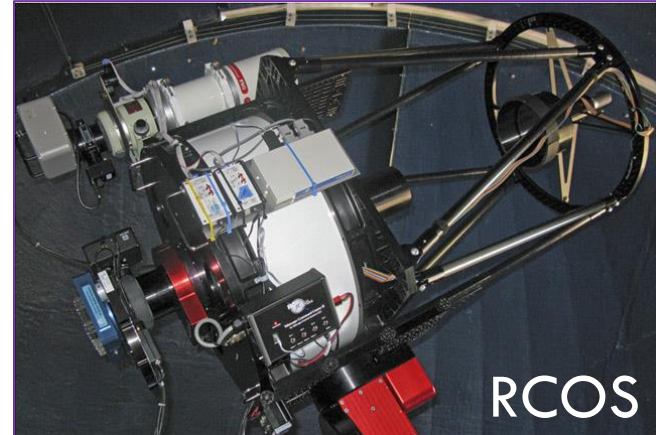
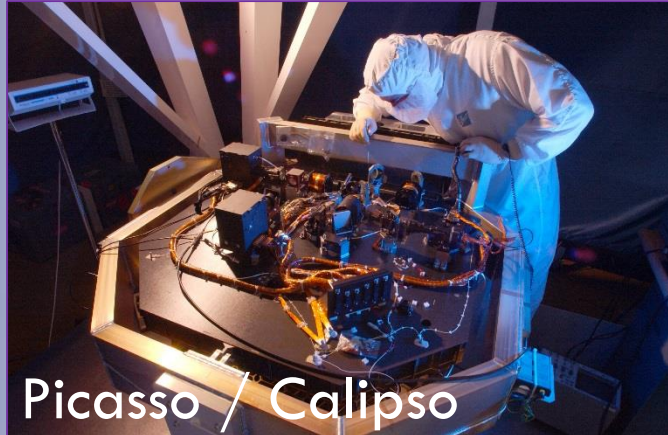
- Lots of Enthusiasm Regarding 3D Printing
 - Can (Should?) Advanced Materials Be 3D Printed?



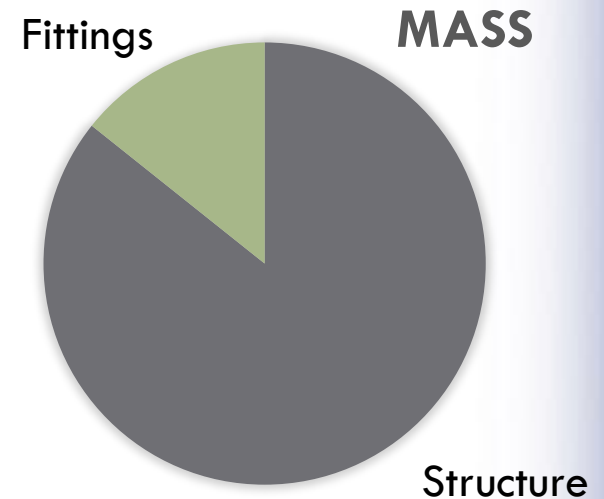
Cost Structure



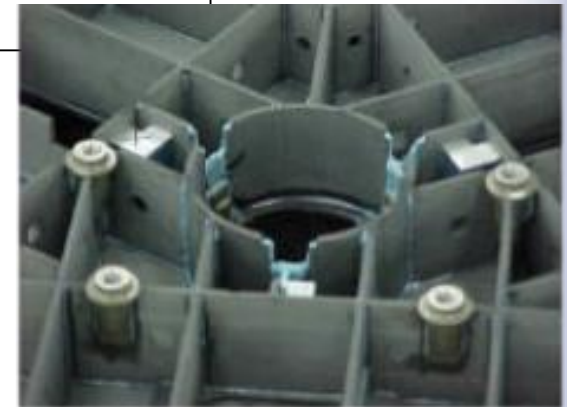
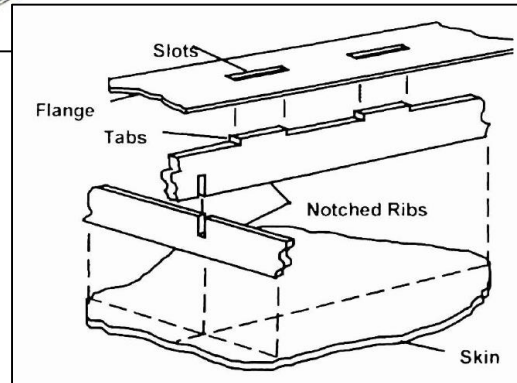
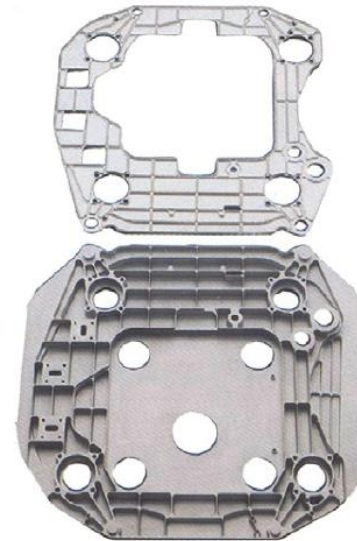
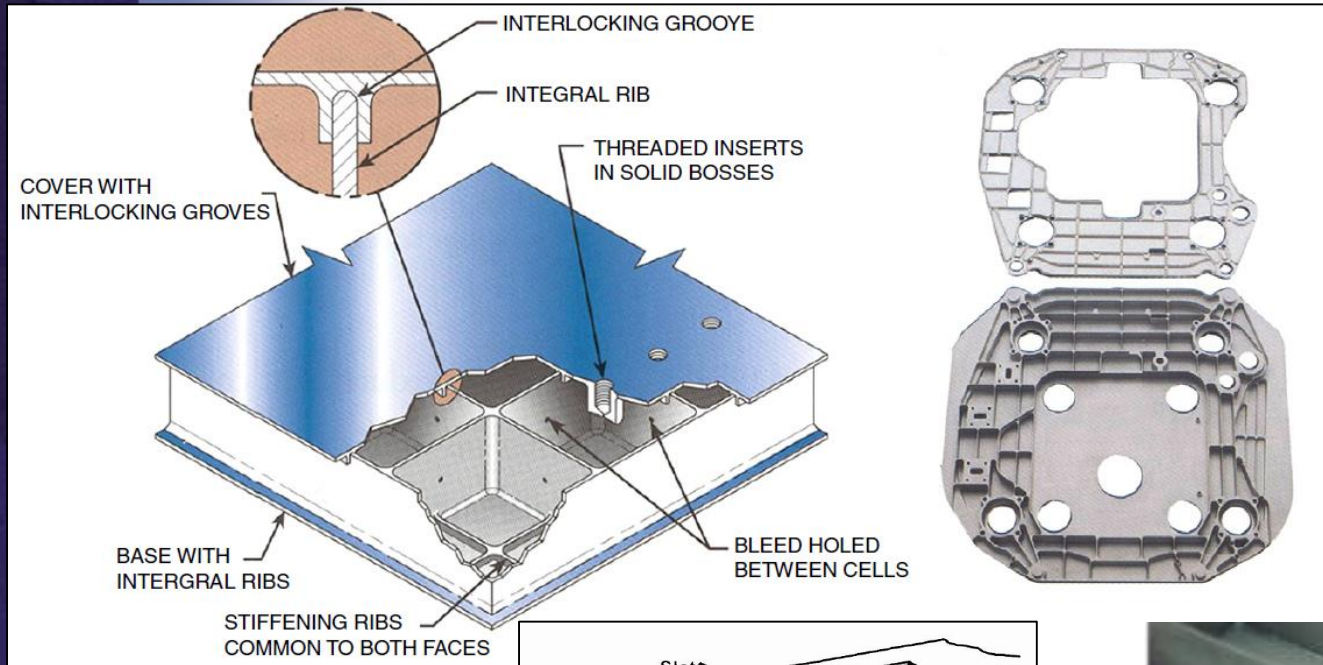
Inspired by Assembled CFRC



- Tube / Truss
- Panel
 - Honeycomb
 - Rib Stiffened
 - Open Back
 - Closed Back



Rib Stiffened Structure Examples



Krumweide, 1991

Pandurangan, 2007

Catanzaro, 2000

Manufacturing Process Concept

Create Flat Stock

- Practical Thicknesses Likely Known
- Inventoried, Optimize for Manufacturing

Design Structure / Fittings

- Mortise / Tenon
- Metallic Threaded Fittings

Cut Flat Stock

- Waterjet
- Wire EDM

Assemble

- EA9394, 3M 2216 (et al. adhesives)
- Tooling for Precision Fittings

Stiffness / Density Suggests Unique Design

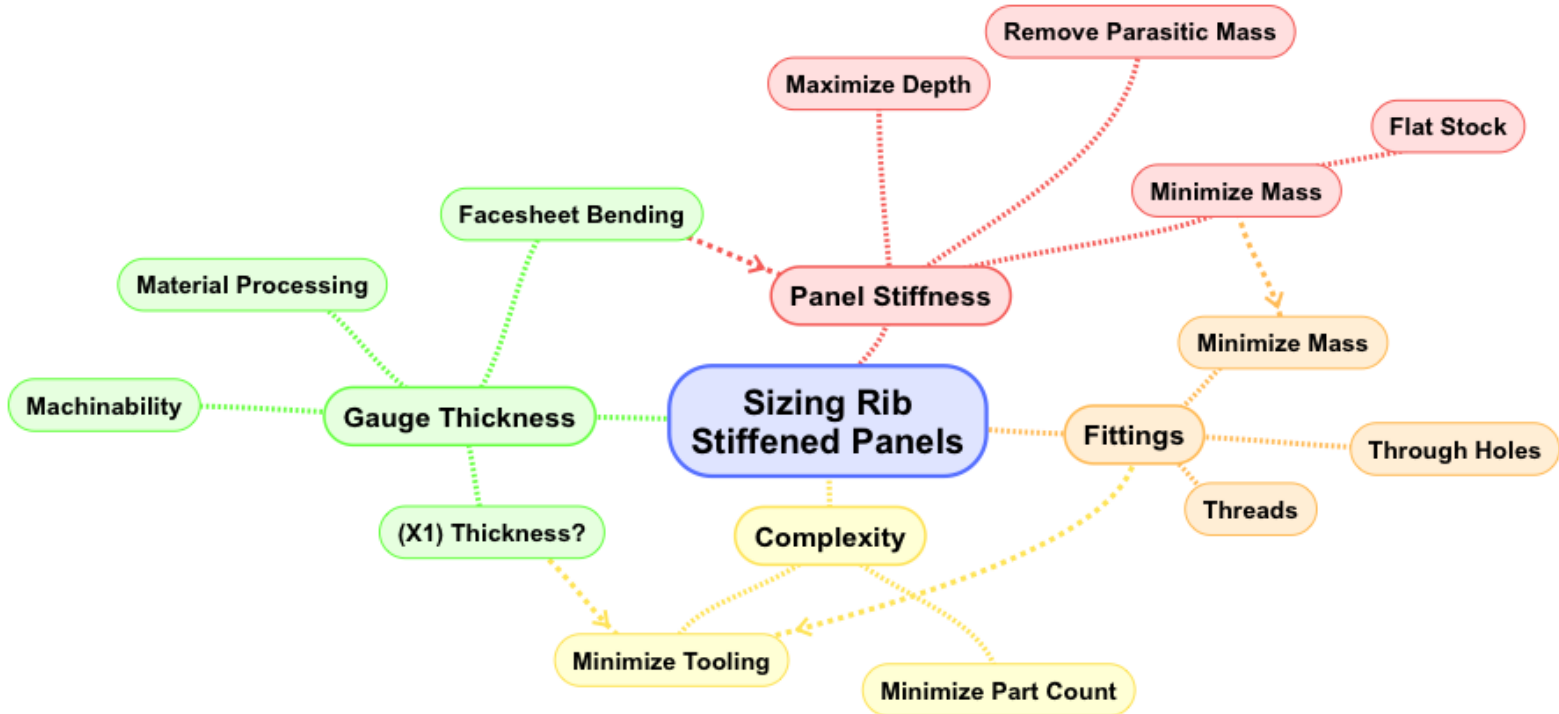
	ρ	E	$(E/\rho)^{1/2}$	α (20C)
CFRC	1.78	93	7.2	0.02
Invar 36	8.05	141	4.2	1
SiC	3.21	456	11.9	2.3
Titanium	4.43	215	7.0	8.8
Beryllium	1.85	287	12.5	11.3

X8 Variation

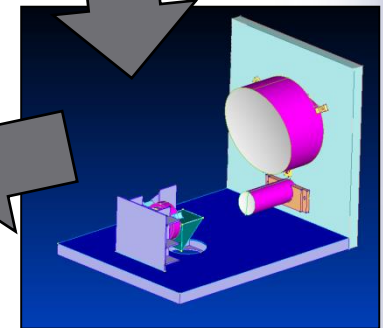
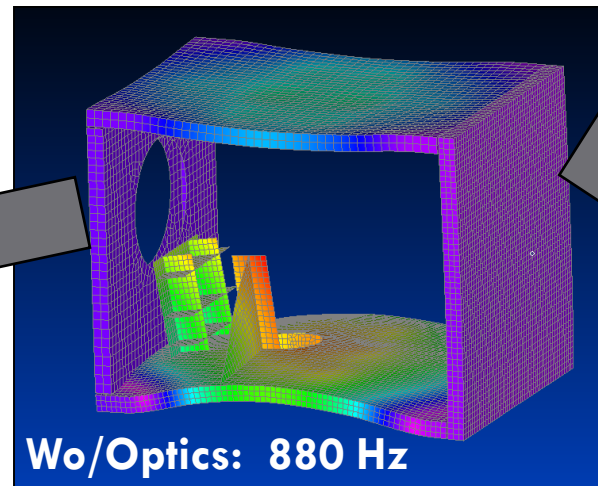
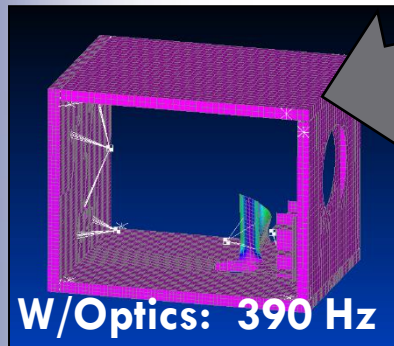
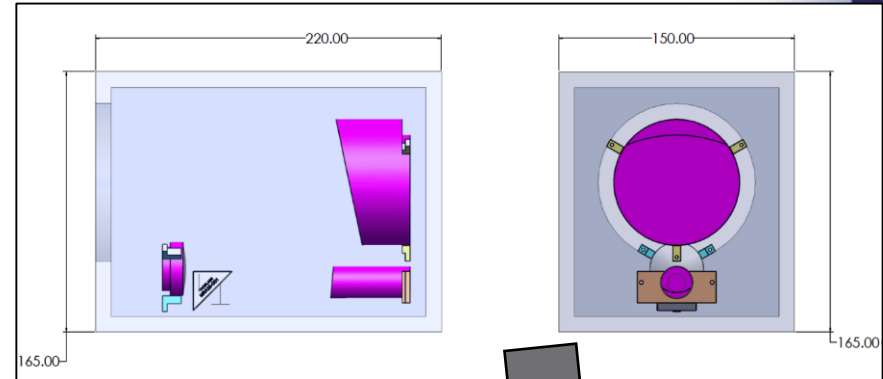
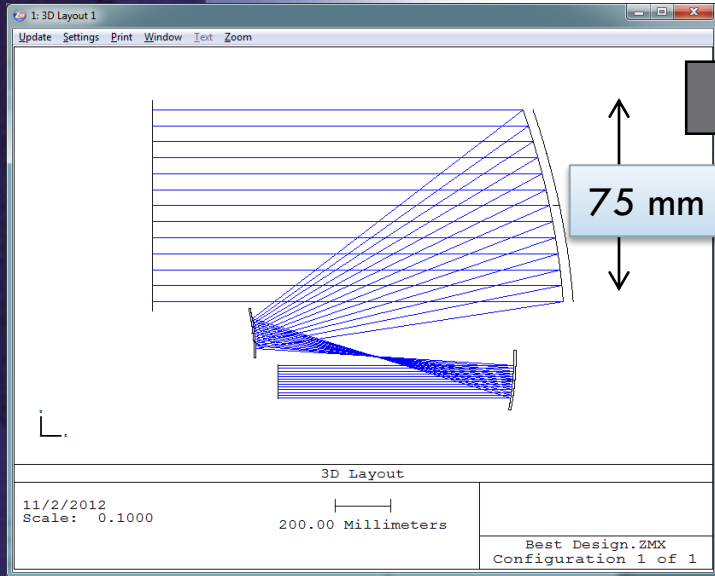
X5 Variation

- How Could These Result in Similar Designs?

Optimize Sizing



Develop Guidelines with Reference Design



- Three Mirror Telescope Studied Previously

Practical Sizing of Structure

- Frequency Goal > 500 Hz (System)
- Mass
 - Attempt Best Lightweighting of Panels
 - Overall Mass Driven by Existing Optics
- Conclusion
 - Panels Should be Approx. $f_0 = 1000$ Hz
 - Facesheet Approx. Thick = 1 mm

Is This Optimal or
Just a Guess?

Rib Density vs. Depth

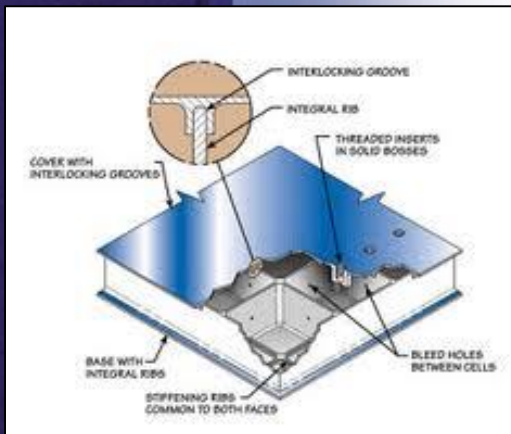
- Assume a Square Rib Pattern
 - Ribs Likely Need to be As Stiff as Facesheet
 - Stiffness Tied to Moment of Inertia

$$I = \frac{1}{12} b \cdot h^3$$

$$\frac{1}{12} t_{rib} \cdot h^3 = I_{rib} = 2 \cdot I_{cell} = 2 \cdot \frac{1}{12} R \cdot t_{face}^3$$

$$t_{rib} \cdot h^3 = 2R \cdot t_{face}^3 \rightarrow R = \frac{t_{rib}}{2} \cdot \left(\frac{h}{t_{face}} \right)^3$$

$$\text{for } t_{rib} = t_{face} \rightarrow R = \frac{h^3}{2t^2}$$



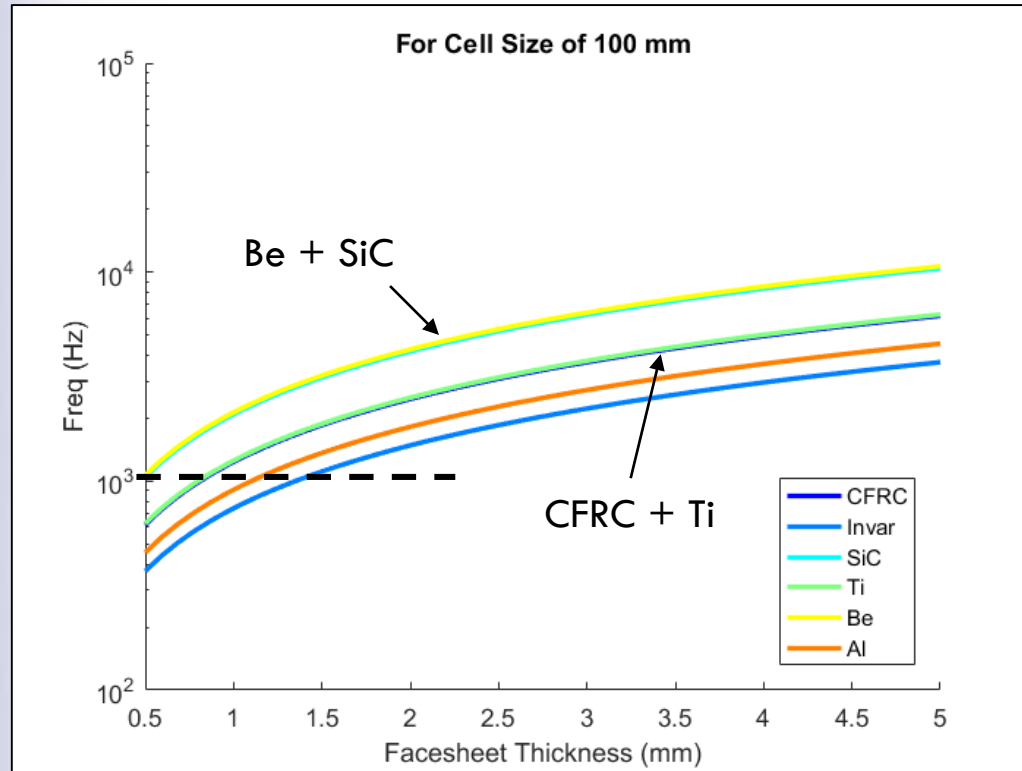
Need to Pick Gauge Thickness

- Gauge Thickness affects Frequency
 - Effect on Rib Stiffness Minimal
 - Effect on Facesheet Thickness Dramatic
- Imagine Each Cell is a Plate with Fixed Edges

$$f_{cell} = \frac{\pi}{1.5} \sqrt{\frac{D}{\gamma} \left(\frac{3}{a^4} + \frac{2}{a^2 b^2} + \frac{3}{b^4} \right)}, \gamma = \frac{A}{m} = \rho \cdot t_{facesheet}$$

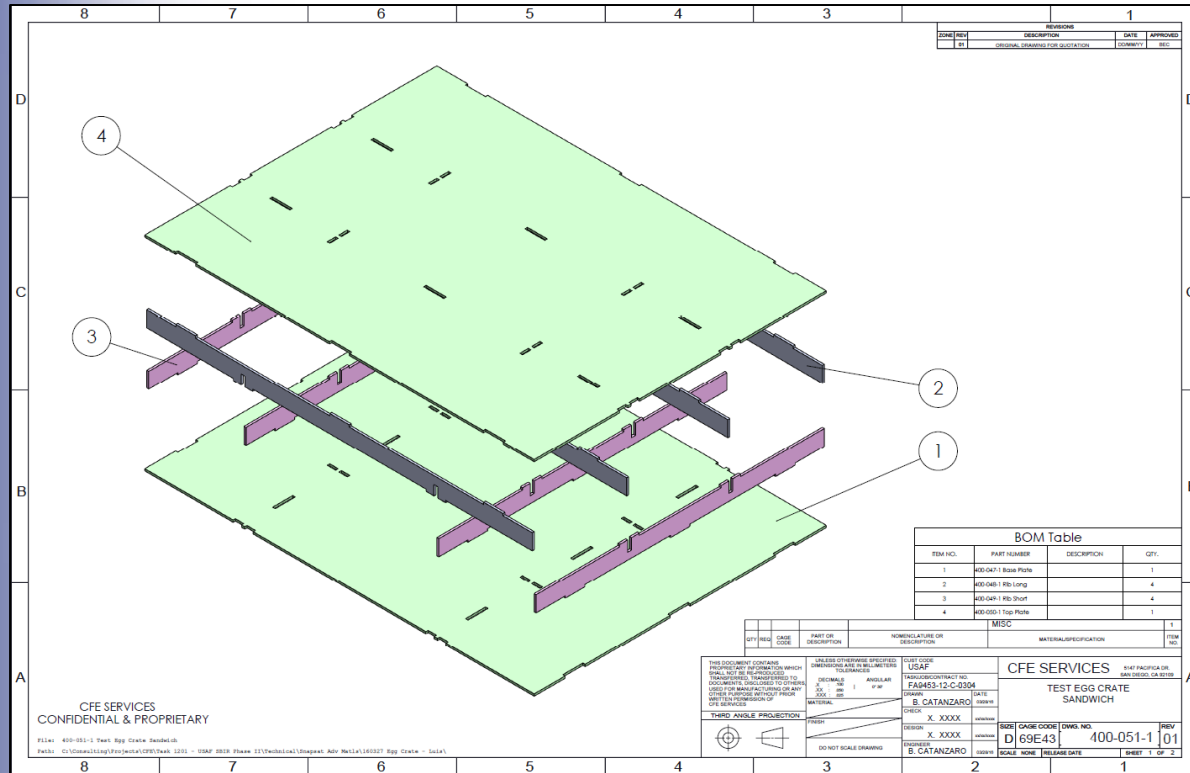
$$a = b = R \rightarrow f_{cell} = \frac{2\pi}{1.5R^2} \sqrt{2 \frac{D}{\gamma}}$$

Gauge Thickness vs. Material



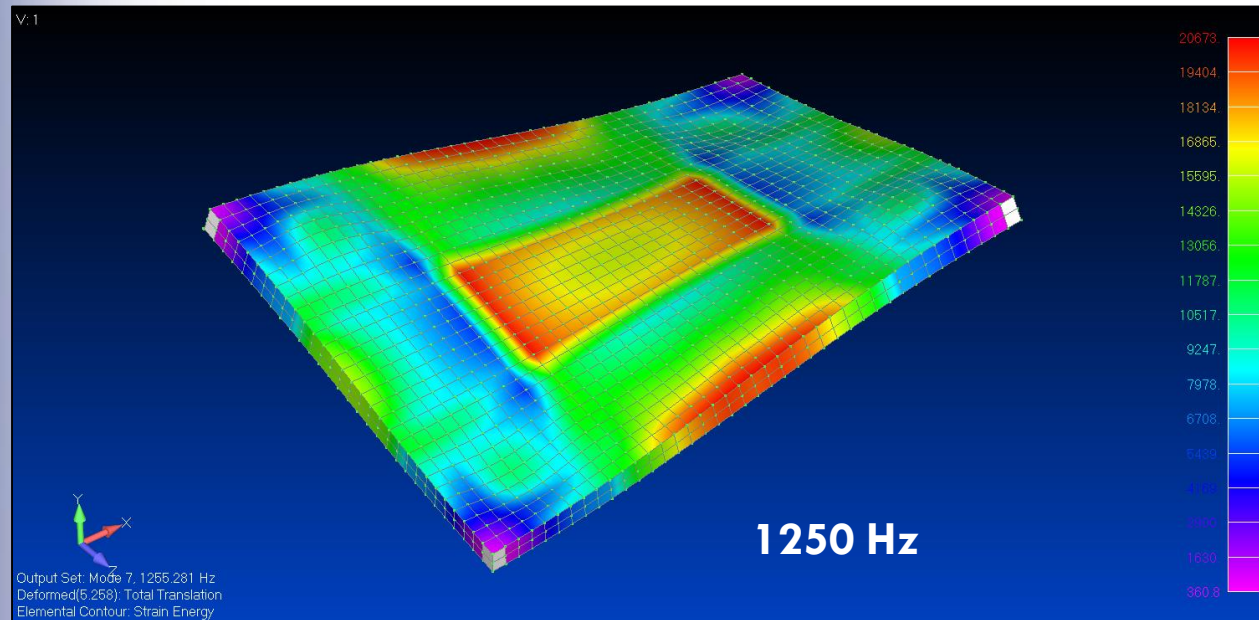
- Aluminum to 1 kHz at 1 mm
- Be + SiC at 1mm Exceed 1kHz... too thick!

Sandwich Panel Reference



- Size: 150 mm x 200 mm
- Goal: $f = 1000$ Hz with Aluminum
- Optimize Design for Adv. Materials

Aluminum Design



$$f_{panel} = \frac{\pi}{2} \sqrt{\frac{D}{\gamma} \left(\frac{2.08}{a^2 b^2} \right)}$$
$$f' = f_{panel} \sqrt{\frac{m}{m + m_{parasitic}}}$$

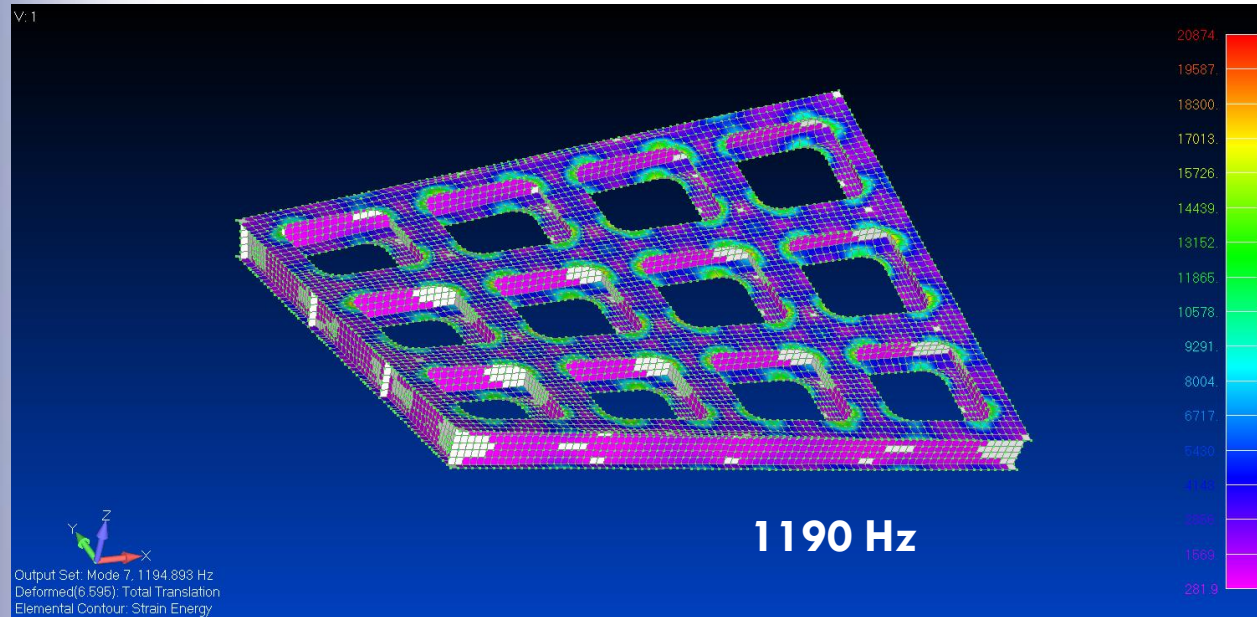
→ $f = 1300 \text{ Hz}$

FEA Results

Material	Free-Free 1 st Freq.	Mass
Aluminum	1 250 Hz	0.19 kg
Silicon Carbide	3130 Hz	0.23 kg
Beryllium	2880 Hz	0.13 kg
Invar	1070 Hz	0.58 kg

- Sandwich Structure Far Exceeds Goals for Advanced Materials
- Remove Unnecessary Mass

Lightweighted Sandwich (SiC)



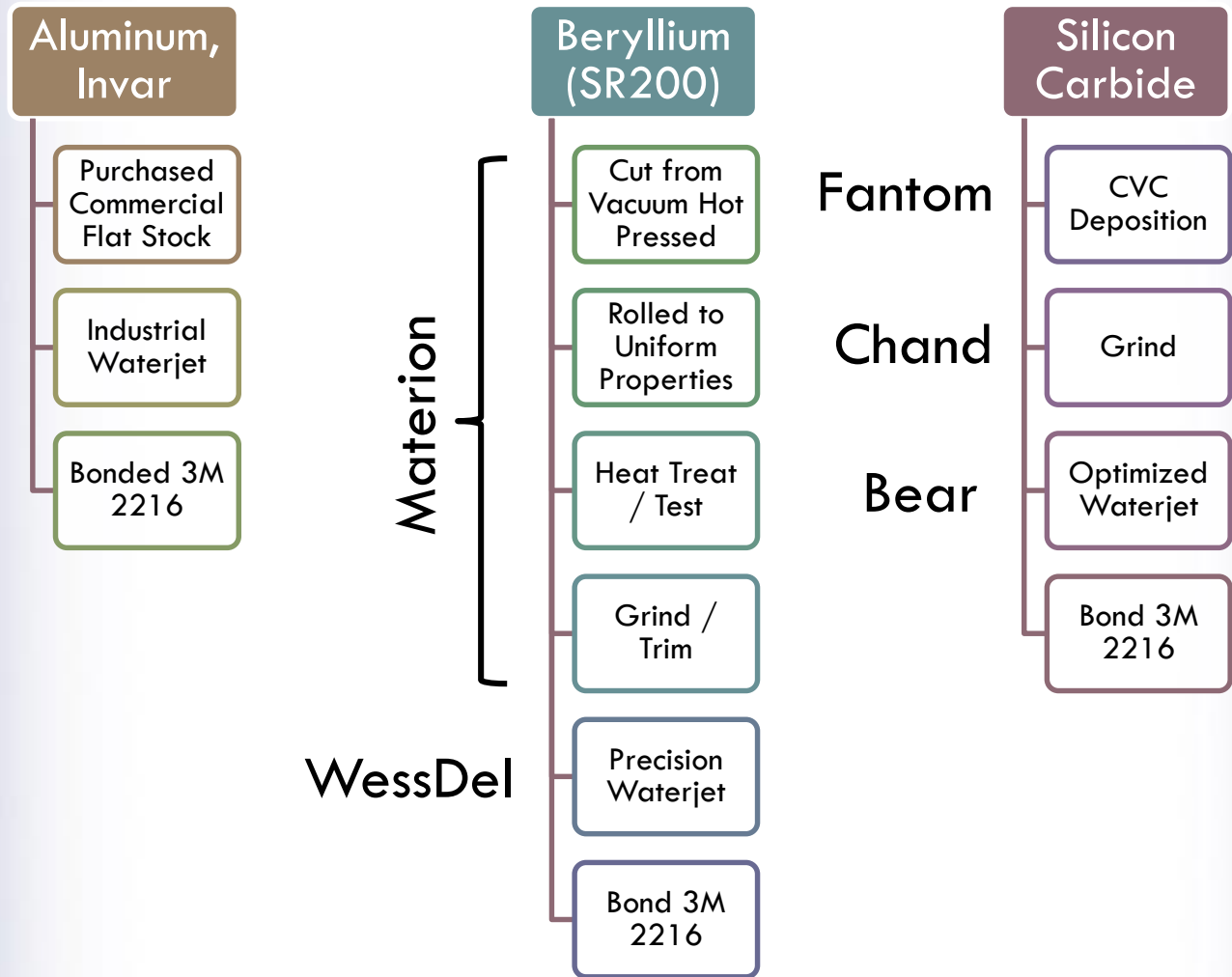
- Similar Performance with Be

FEA Results

Material	Free-Free 1st Freq.	Mass
Aluminum	500 Hz	0.13 kg
Silicon Carbide	1190 Hz	0.15 kg
Beryllium	1050 Hz	0.089 kg
Invar	420 Hz	0.39 kg

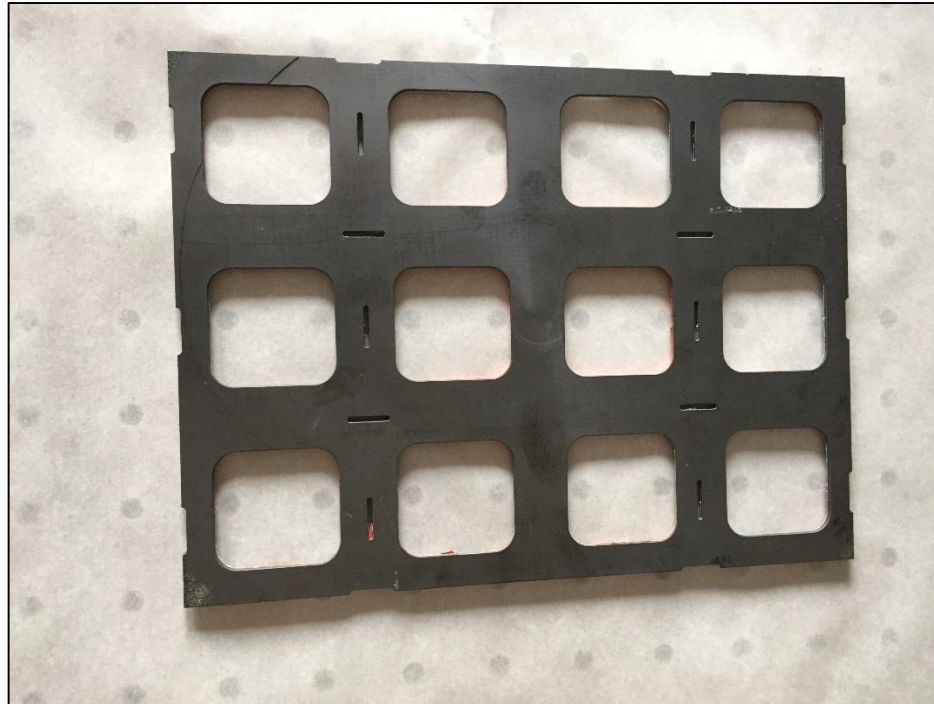
Manufacturing Panels

CalWaterjet

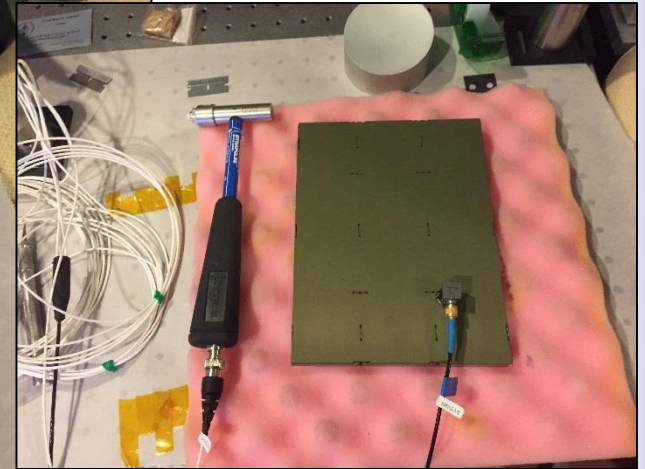
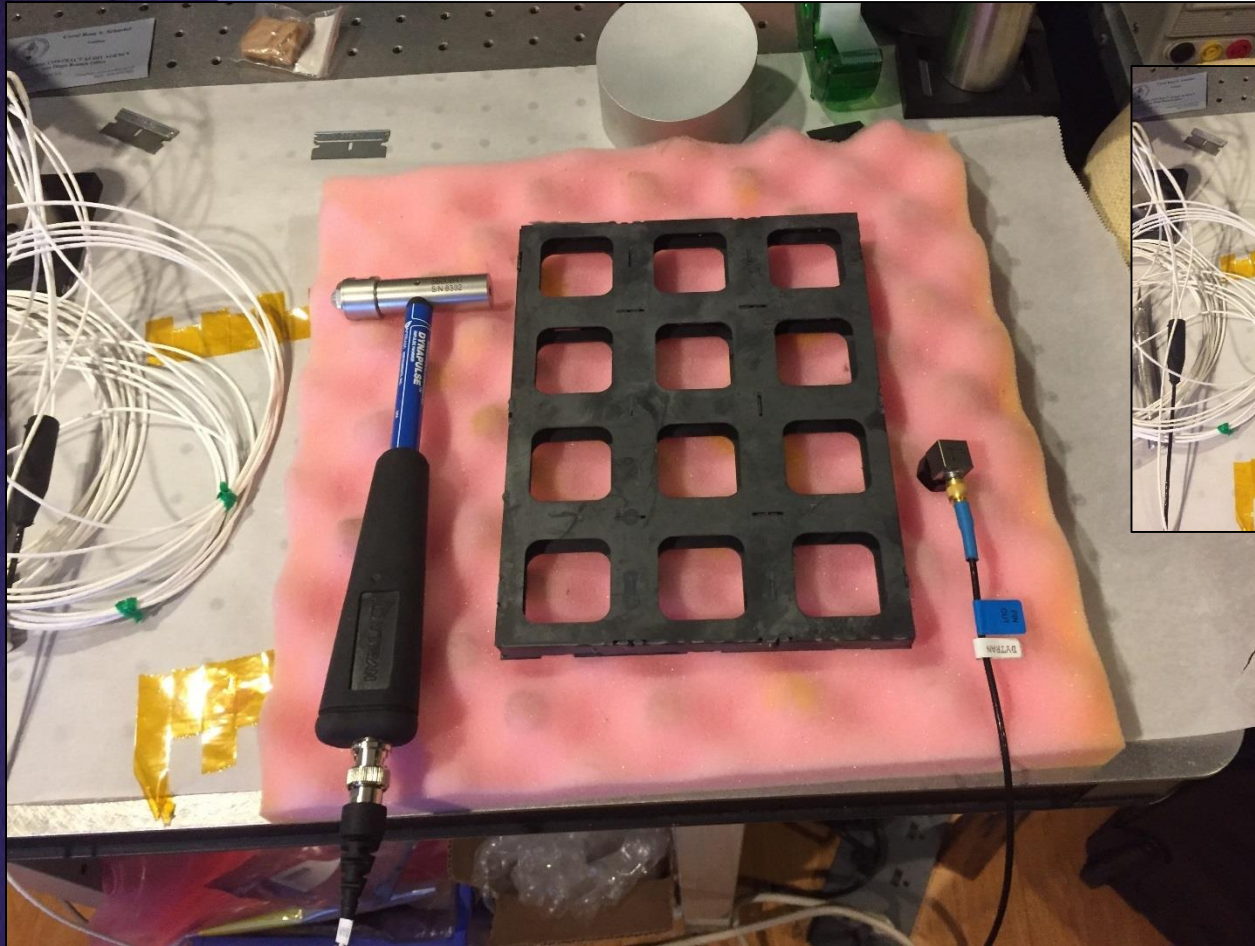


Challenges in Waterjet

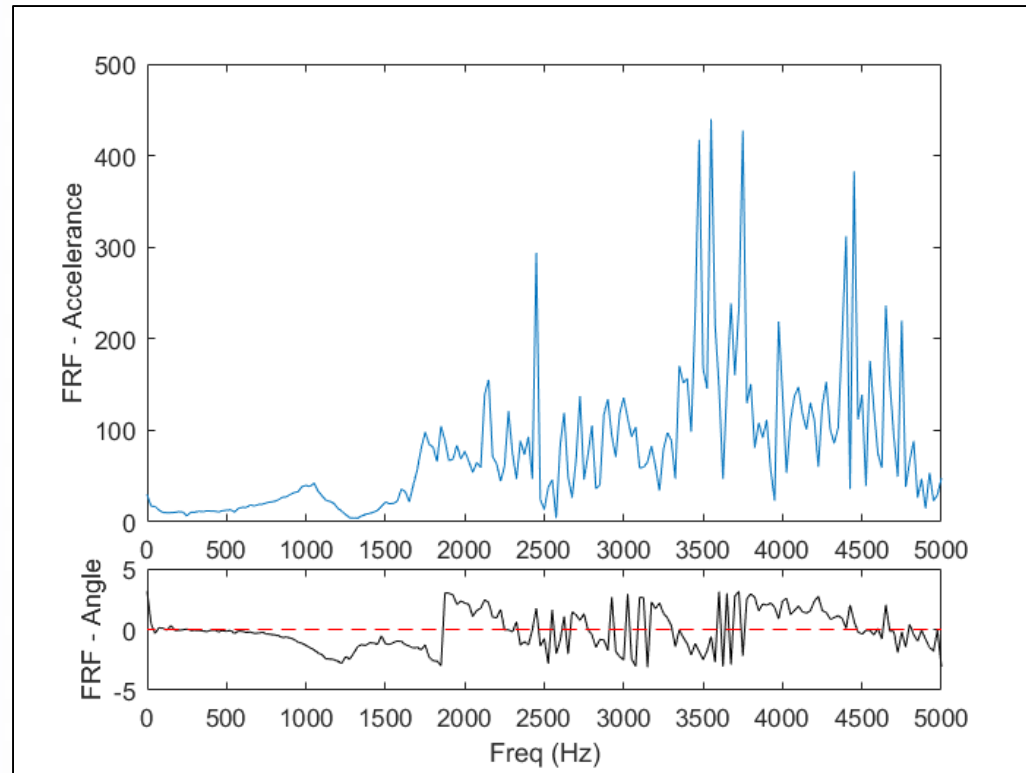
- Silicon Carbide Robust to Grinding
- Waterjet Processing Left Cracks



Stiffness Testing



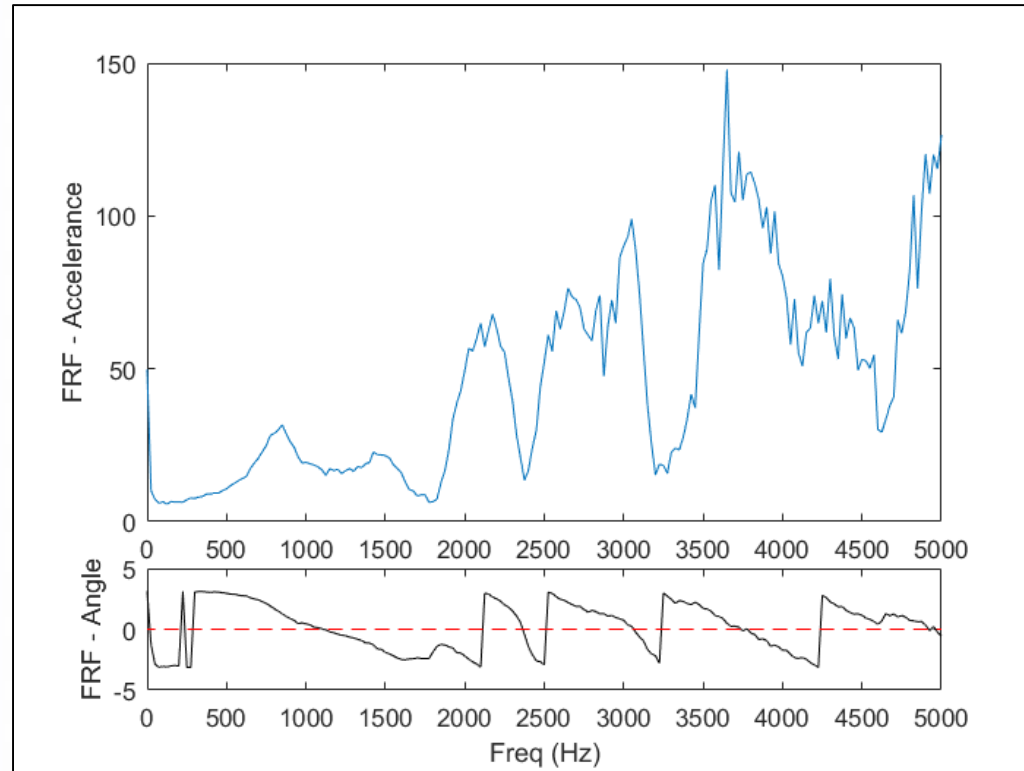
Frequency Response Function



Aluminum Sandwich: 1 kHz



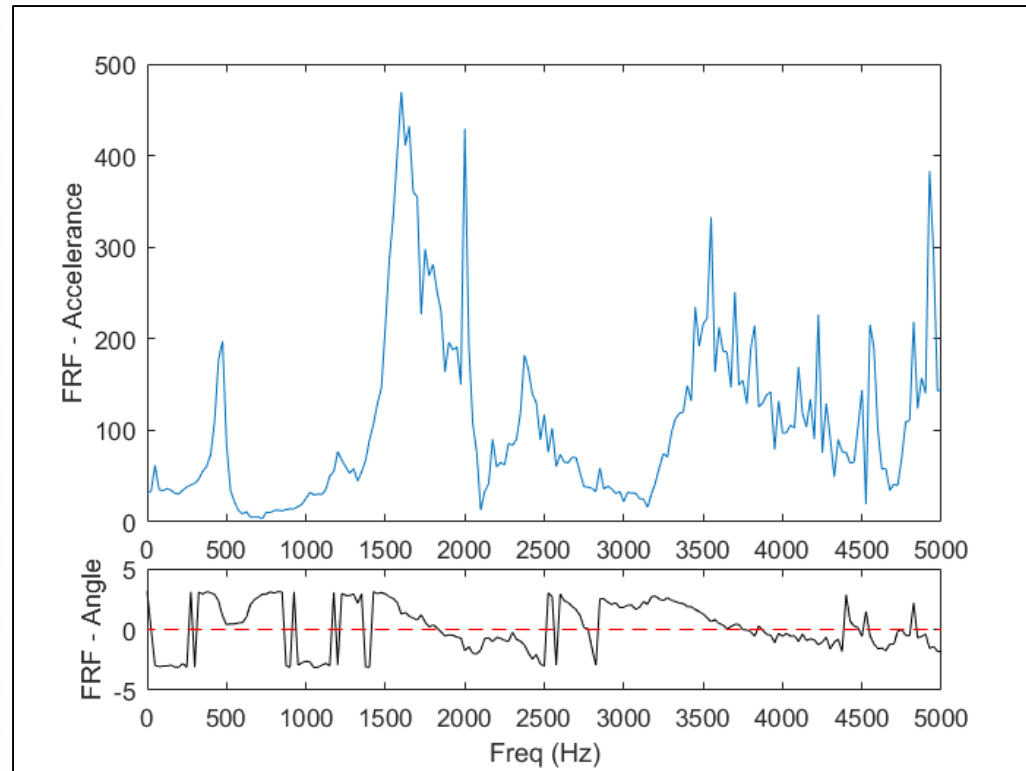
FRF / Invar Sandwich



Invar: <math>< 900 \text{ Hz}</math>



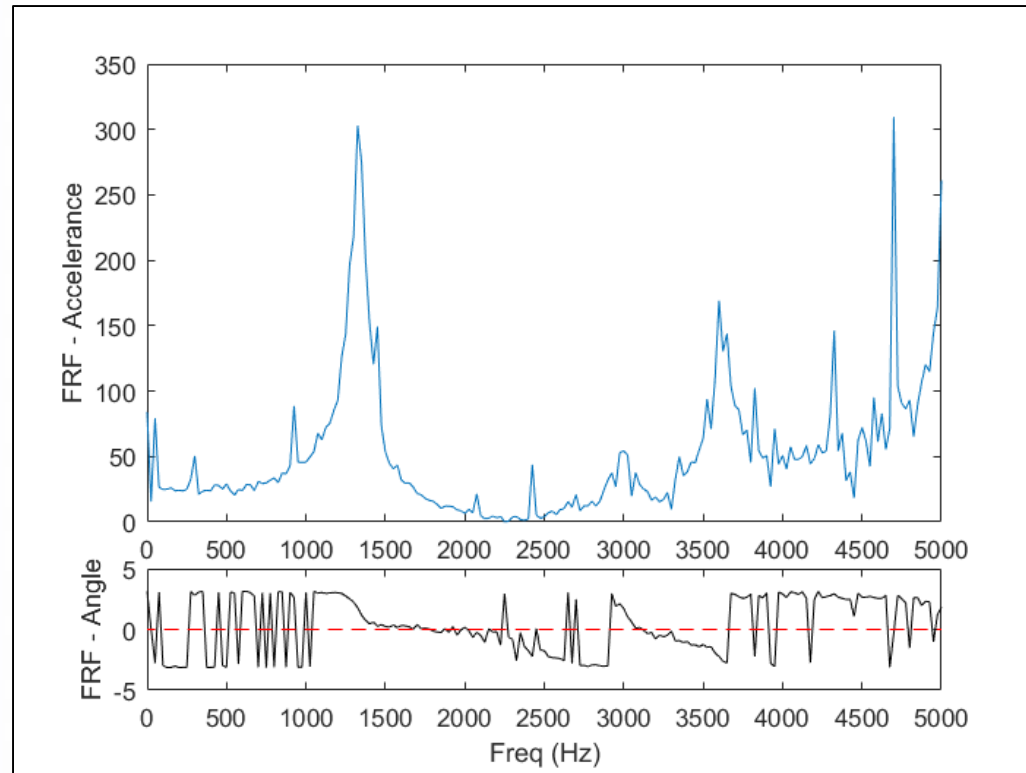
FRF / Aluminum Pane



Aluminum Pane: <500 Hz



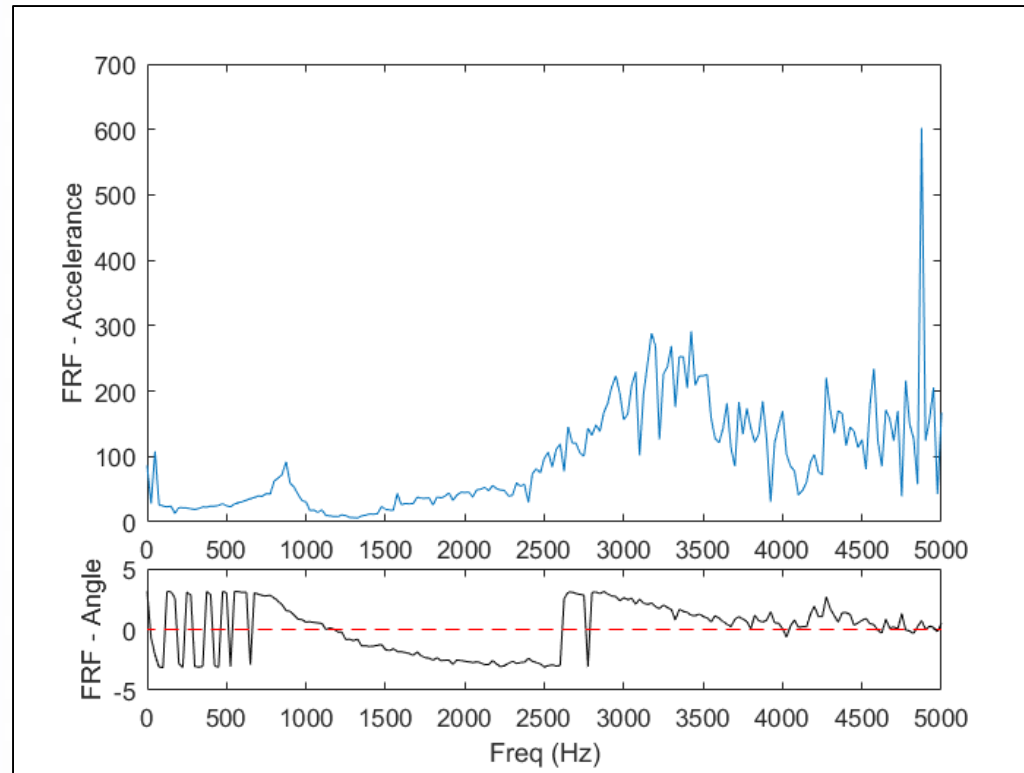
FRF / Beryllium Pane



Be Pane: 1300 Hz



FRF / Silicon Carbide Pane



SiC: 900 Hz*

*With Cracks



Lessons Learned

- Not Always Easy To Achieve Geometry that Optimizes Specific Stiffness
 - Facesheets for Panels must be Thin
 - Thickness Affects
 - Robustness, Grinding Costs
- Beryllium
 - Material and Processing Straightforward
- Silicon Carbide
 - Process Steps Known but Maturing
 - Material Much More Robust than Advised
- Costs... Just About the Same

Acknowledgements

- Materion
 - Rob Michel / Acquired Material Instantly
- WessDel
 - Rich Bernard / Outstanding Waterjet Processing
- Fantom
 - Fred Styer / Provided Material Instantly
- Bear Machining
 - Blair Stultz / Developed Waterjet Process