



STMD Programs



Transformative & Crosscutting Technology Breakthroughs

Pioneering Concepts,
Developing
Innovation
Community

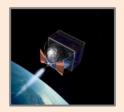
Creating Markets & Growing Innovation Economy



Game Changing
Development Program



Technology Demonstration Missions Program



Small Spacecraft Technologies Program



Space Technology Research Grant Program



NASA Innovative Advanced Concepts (NIAC) Program



Center Innovation Fund Program



Centennial Challenges Prize Program



Small Business Innovation Research & Small Business Technology Transfer (SBIR/STTR) Program



Flight Opportunities Program



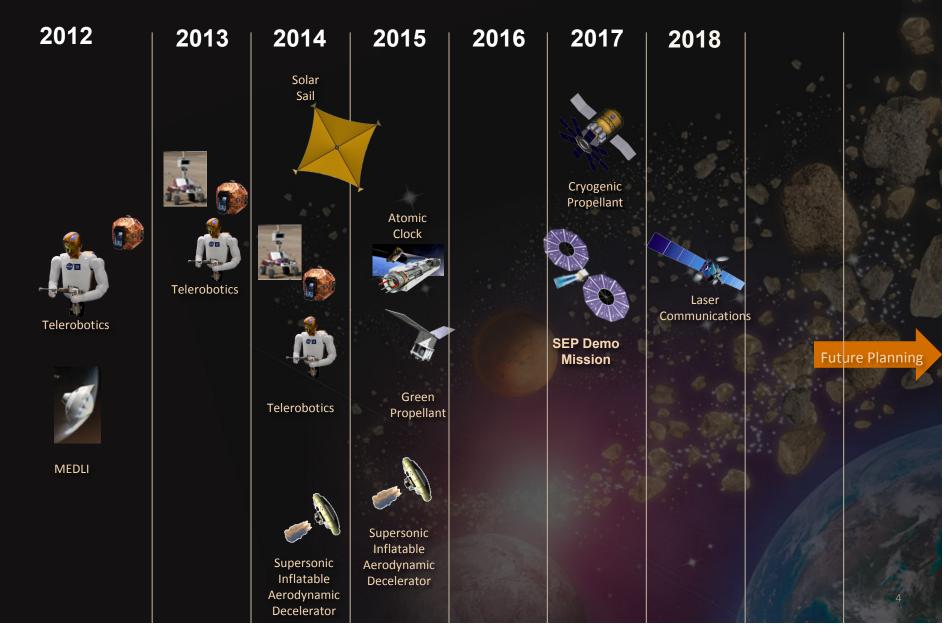
Why TDM



Infusing Revolutionary, Crosscutting Technologies to Benefit NASA and the Nation



Technology Demonstration Missions Major Events & Milestones





TDM Portfolio

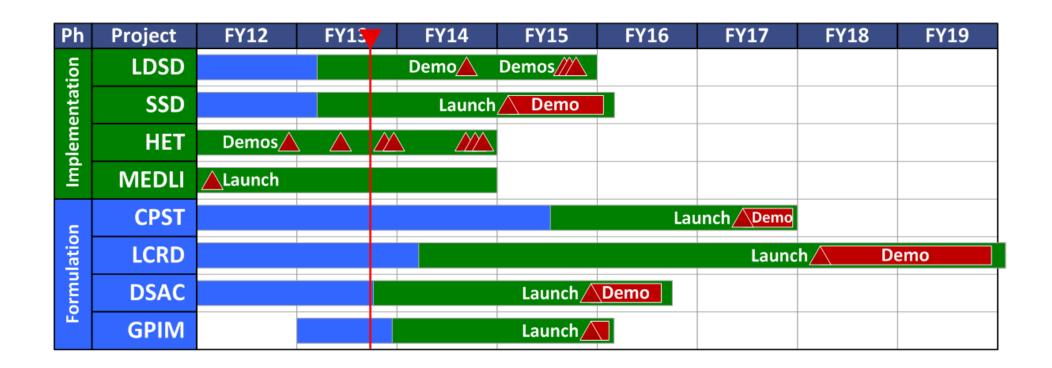


Technology Areas	ETD: TDM				FY 10	FY 11	FY 12	FY 13	FY 14	FY 15	FY 16	FY 17	FY 18	Infu- sion
A	Human Explorat	ion Telerobotics (HET	.)				⑤		≯ ⑦॥	111111111111			111111111	>
* • *	Cryogenic Propellant Storage and Transfer (CPST)					<u>(5)</u>						> 6	>	
*	★ Green Propellant Infusion Mission						⑤			> ⑦Ⅲ			S	
* *	Solar Electric Propulsion*							<u></u>			> ⑦॥		🔖	
Technology Areas	CSTD: TDM				FY 10	FY 11	FY 12	FY 13	FY 14	FY 15	FY 16	FY 17	FY 18	
		personic Decelerators	(LDSD)				<u></u>			≻ 6⊪				S
F	Laser Communic	cations Relay Demon	stration (LCRD)				6						··· >	*
F	Deep Space Ator	mic Clock (DSAC)					<u>(5)</u>					≻ ⑦॥	111111111	
*	Sunjammer Sola	ar Sail Demonstration	(SSD)				<u></u>			> 9∥			111111111	%
Technology Areas (1	TA)	TA.4. Robotics	Á	TA.8. Sci.	Instr./Sei	nsors		TA	.12. Mate	rials/Struc	tures 😘	Infu	sion pa	th to:
TA.1. Launch Propulsi	on 🏨	TA.5. Comm./Navigation	K	TA.9. EDL	_			TA	.13. Grour	nd/Launch			nce	
TA.2. In-Space Propuls	sion 🌋	TA.6. Human Health	+	TA.10. No	anotechol	ogy	1	TA	.14. Thern	nal	#	Ехр	loration	1
TA.3. Space Power/Sto	orage 🔌	TA.7. Human Expl. Dest.		TA.11. M	odeling/S	imulation		Те	chnology l	Readiness	Levels (TR	(L) ①	9	



TDM Portfolio





Green Propellant Infusion Mission (GPIM) Overview



Overview: GPIM is a spaceflight demonstration of a complete propulsion system for spacecraft attitude control and primary propulsion using the "Green Propellant" AF-M315E developed by AFRL as a substitute to Hydrazine

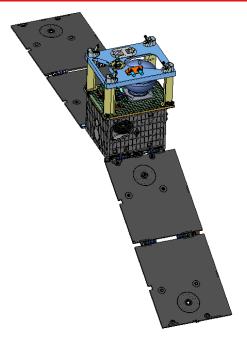
Benefits: Utilizing AFM-315E will significantly reduce the safety restrictions and complexities placed on hydrazine ground operations while substantially increasing performance (50% improvement over hydrazine)

Thrusters: One 22 N and four 1 N thrusters with new high-temperature catalyst technology designed, developed, and tested by Aerojet. Cost comparable to hydrazine thrusters.

Spacecraft bus: Flight-proven Ball Configurable Platform (BCP) 100 bus

Access to space: Secondary payload (ESPA compatible) on DOD Space Test Program launch STP-2 (Falcon Heavy) in Fall 2015

Team members: Ball Aerospace (lead), Aerojet, Air Force Research Laboratory, Space & Missile Systems Center, NASA Glenn Research Center, NASA Kennedy Space Center



GPIM spacecraft



GPIM propulsion system



Propellant Characteristics

- AMF-315E was developed by AFRL in 1998 as an alternative to Hydrazine
 - Focus on reducing toxicity and increasing performance
- Propellant is an ionic salt blend of HAN (Hydroxylammonium Nitrate) solid oxidizer with water and a compatible fuel
- >50% improvement in volumetric performance versus hydrazine

AF-M315E

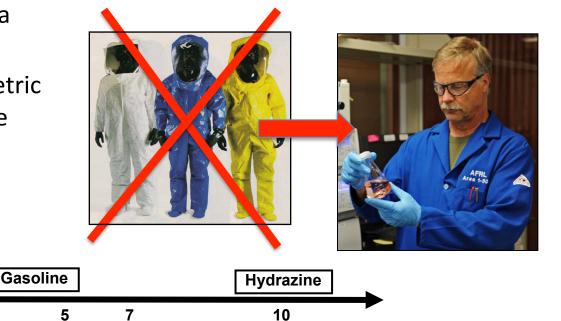
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Toxicity Scale

INH, Ipronid

Water

- Less toxic (LD50) than caffeine
- Negligible vapor toxicity allows propellant loading with typical PPE (no SCAPE or monitoring requirement)



*, INH used in treating tuberculosis; Ipronid is an antidepressant drug



Project Team



Ball Aerospace

- Program Lead PM & PI
- Project System Engineering
- Mission requirements
- · Flight thruster performance verification
- Ground and flight data review
- BCP-100
- AI&T
- Launch and Flight Support

GREEN PROPELLANT INFUSION MISSION

Aerojet Redmond Operations – Co-I

- Green propulsion payload
- 1N and 22N thruster development
- Payload integration
- Ground and flight data review

NASA GRC - Co-I

- Plume modeling
- Thruster independent testing
- Experimental plume diagnostics
- Ground and flight data review









Air Force SMC

- Mission Operations
- Ground Segment Support
- STPSat GSE



Kennedy Space Center – Co-I

- · IMLI for flight experiment
- Green propellant handling, loading processes
- Propellant assay analysis
- Ground and flight data review

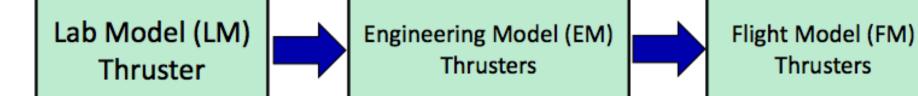
AFRL Edwards - Co-I

- Propellant (contribution)
- Propellant loading cart (contribution)
- Propellant loading
- Ground and flight data review



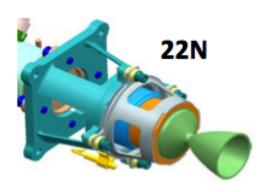
Thruster Development



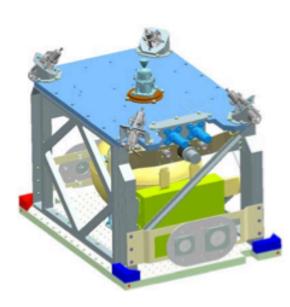




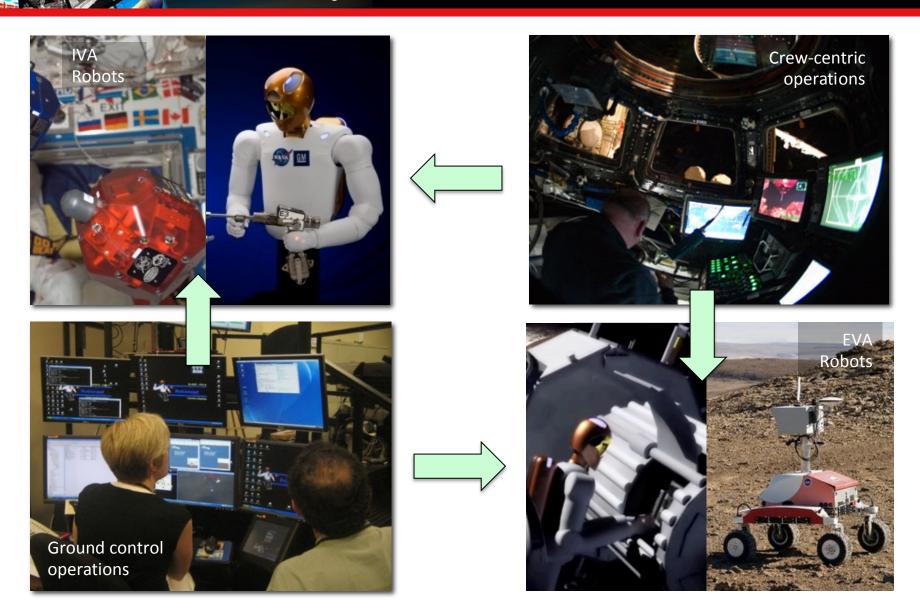
22N







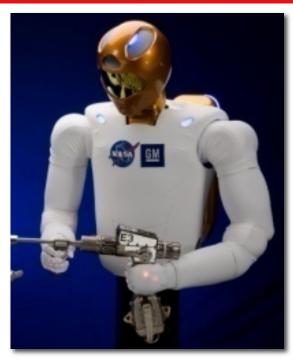
Human Exploration Telerobotics





Telerobotics Systems





Robonaut 2 (R2)

- Humanoid robot (42 DOF, human-scale/ safe)
- Ground control and crew centric operations
- Perform dexterous manipulation tasks



Smart SPHERES

- Free-flying robot (6 axis, cold-gas propulsion)
- Ground control and crew centric operations
- Perform remotely operated mobile sensor tasks



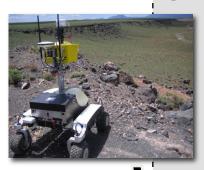
Surface Telerobotics

- Mobile robot on surface (Moon, asteroid, Mars)
- Crew centric operations from inside flight vehicle
- Perform surface activities before/support/after crew

Surface Telerobotics Roadmap



Ground Analogs



Develop systems for crew teleoperation of surface robots

Implement and test multiple conops

Simulate future human mission concepts

ISS Laboratory



Obtain baseline system engineering data

Validate & correlate prior ground simulations via high-fidelity ops sims

Reduce risk (exploration architectures based on inaccurate assumptions)

Beyond LEO



Enable crew to operate telerobots when site, dynamics and distance preclude ground control

Enable crew to operate surface robot from orbit when circumstances preclude ground control

HRS, AES

HET

HEOMD missions



Surface Telerobotics













Sunjammer

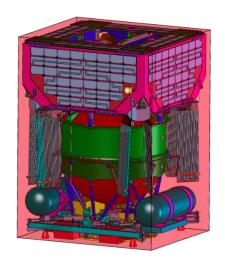


• Demonstration of a mission infusable solar sail co-manifestested

with DSCOVR

Sunjammer mission objectives

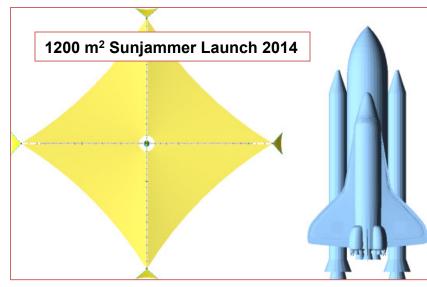
- Deployment
- Attitude Control
- Navigation



Dim.: 28in x 28in x 38in

Mass: 153kg (wet)



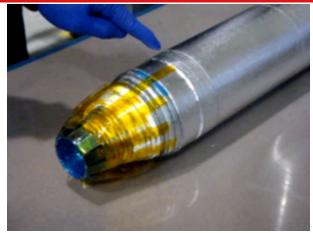




Sunjammer

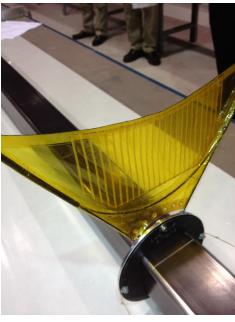


Boom Stowed





Spreader Web with Spreader Lines Stowed



Spreader Web with Spreader Lines Deployed



Full Length Boom Deployed



Sunjammer



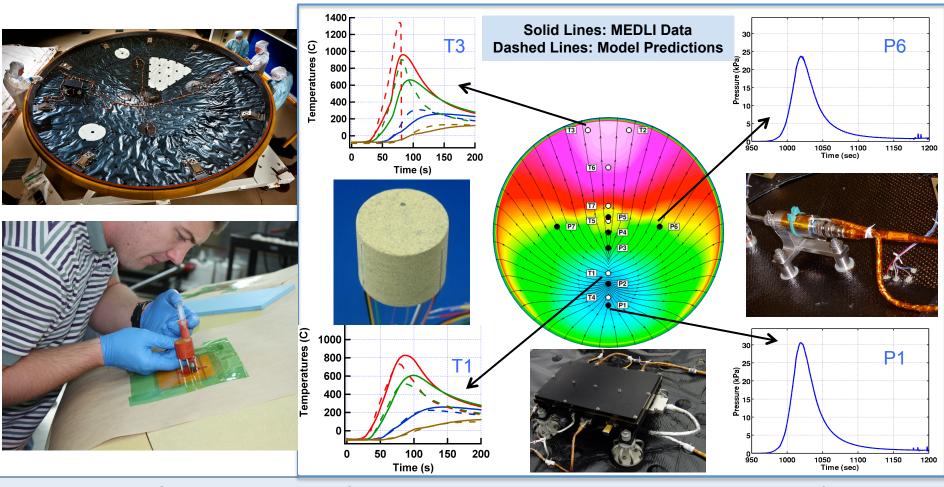


Video - Typical Sail Folding Technique (ISP Sail) ETU Sail Deployment
Development
June 2013



Full Scale ETU Sail Suspended from Manual Deployment Fixture

ISL Entry, Descent and Landing Instrument Flight Data

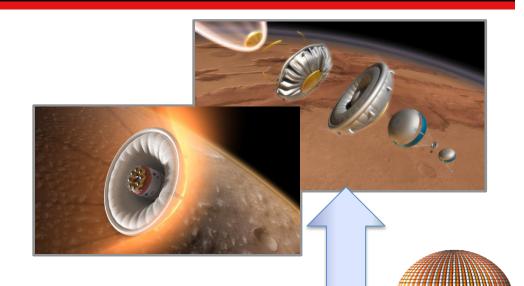


- MEDLI (Mars Science Laboratory (MSL) Entry, Descent and Landing Instrumentation) successfully measured temperature and pressure on MSL's heatshield during Mars entry on August 6, 2012.
- > **MEDLI Thermal data** shows heating predictions were high in some places and low in others; recession less than 0.1"
- > MEDLI Pressure data shows the spacecraft flew as expected, and encountered some winds below 15 km

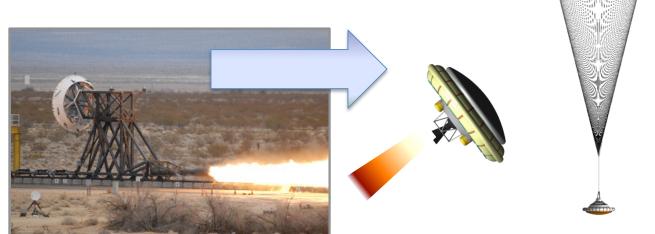
ow Density Supersonic Decelerator



- LDSD is designing and testing EDL systems to enable a new class of planetary entry vehicles
- Improvement over MSL
 - Up to 1mt increase in landed mass
 - 25% increase in elevation
 - 3x reduction in landing ellipse









LDSD Components



SIAD-R (6m) Decelerator SIAD-E (8m) Decelerator

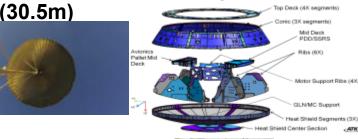
Ballute (4.4m) (Chute Deployment Device) Supersonic Parachute (30.5m)

Core Structure (Including Heatshield)

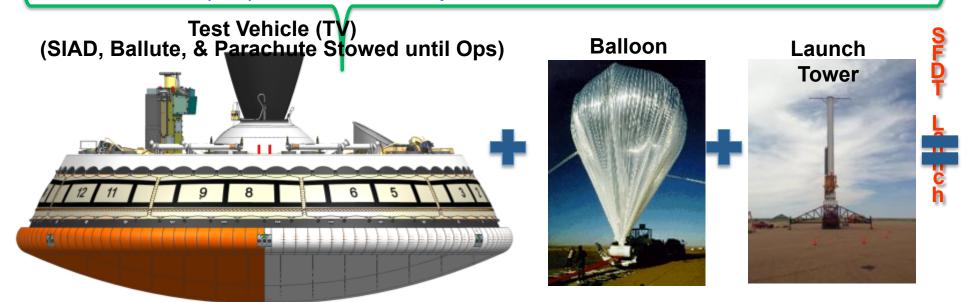








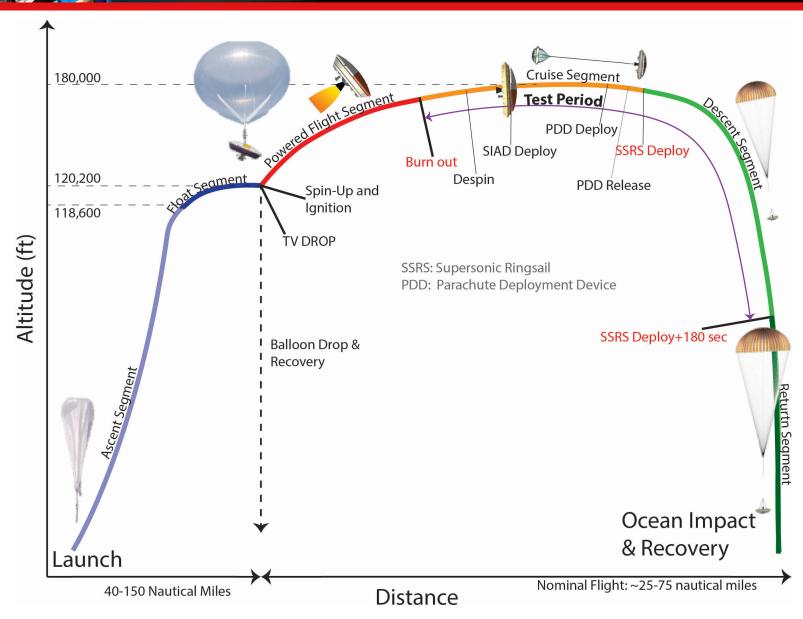
The components <u>above</u> along with the Avionics Pallet, Star 48-Motor, Spin Motors, GLN-MAC (IMU), cameras, etc. comprise the Test Vehicle shown below

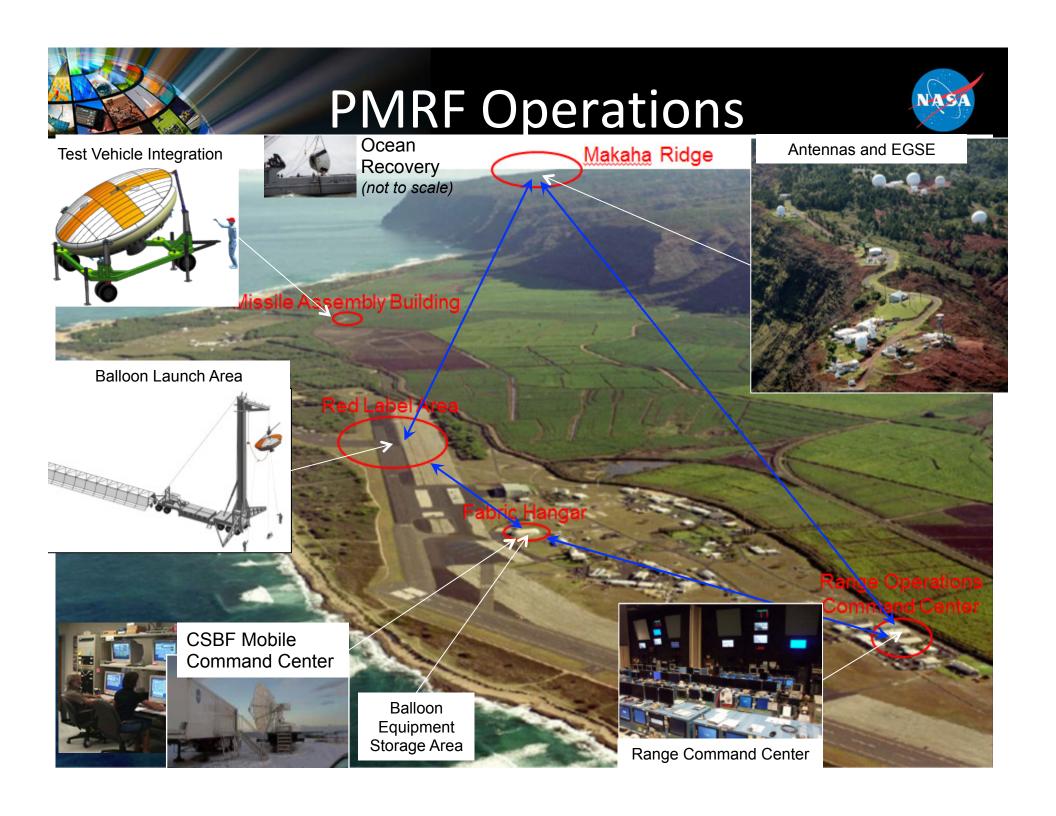


(Not to Scale)

Launch & Trajectory Overview







DSD Supersonic Flight Dynamics Test (SFDT) Test Start: **Test Vehicle** Test Vehicle Landing & **Rocket Powered Decelerator Parachute** Water Recovery Footprint **Splashdown Test Vehicle** Flight: aimed NE **Deployed Deployed Separates** from Balloon Ground Track Shown Balloon Flight **Termination Point Balloon Water Landing** Drop Alt.:120,000 ftand Recovery Footprint **Balloon Climb-out Ground Track**

Balloon Launch

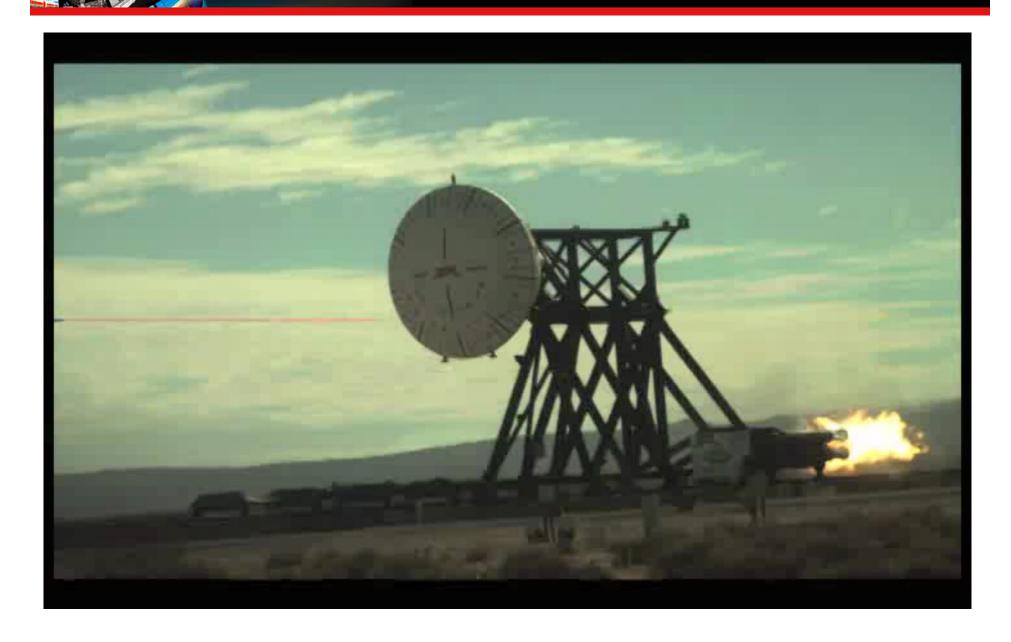


Component Milestones



LDSD Component	Status	Next Milestone Event or Phase	SFDT #s* 1 - Jun 2014 (-R) 2 - Jun 2015 (-R) 3 - Jul 2015 (-R) 4 - Aug 2015 (-E)
SIAD-R	Completed all Fab and Proof tests Development including Rocket Sled tests at China Lake, and Rapid Inflation Tests; Flight Units in Fab	Flight SIAD-R will be integrated to test Vehicle during I&T at JPL Jan - Mar 2014	1 - 3
SIAD-E	Final stages of Designing and beginning Fab	SIAD Development Verification Tests using Rocket Sled at China Lake Jan - Mar 2014	4 (fly on last Balloon flight)
Ballute	Design completed; Fab of the first test ballute has started.	Structural & Inflation Sept - Oct 2013	1 - 4
Parachute	Final stages of Design and Development Testing; Fab of first two test parachutes (Disk Sail and Ring Sail) has started and one if nearing completion.	Conduct next round of Parachute Verification Tests at China Lake Sept - Dec 2013	1 - 4
Core Structure	In Fabrication	Complete fab and Testing NLT Dec 2013 for I&T at JPL Jan - Mar 2014	1 - 4
Launch Tower	Fabrication and Check-Out Completed; Shipped to Ft. Summer	Integrated with Small Balloon and TV Simulator and tested at Ft. Summer, NM in Aug 2013	1 - 4

Rocket Sled Test at China Lake (Nov 2012)



Rocket Sled Test at China Lake (Nov 2012)







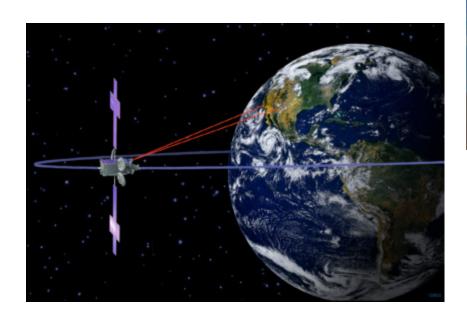
Build Charts



Laser Communications Relay Demonstration



Project Summary: NASA's first, long-duration optical communications mission. The project will help mature concepts and deliver technologies applicable to both near-Earth and deep-space communication network missions.



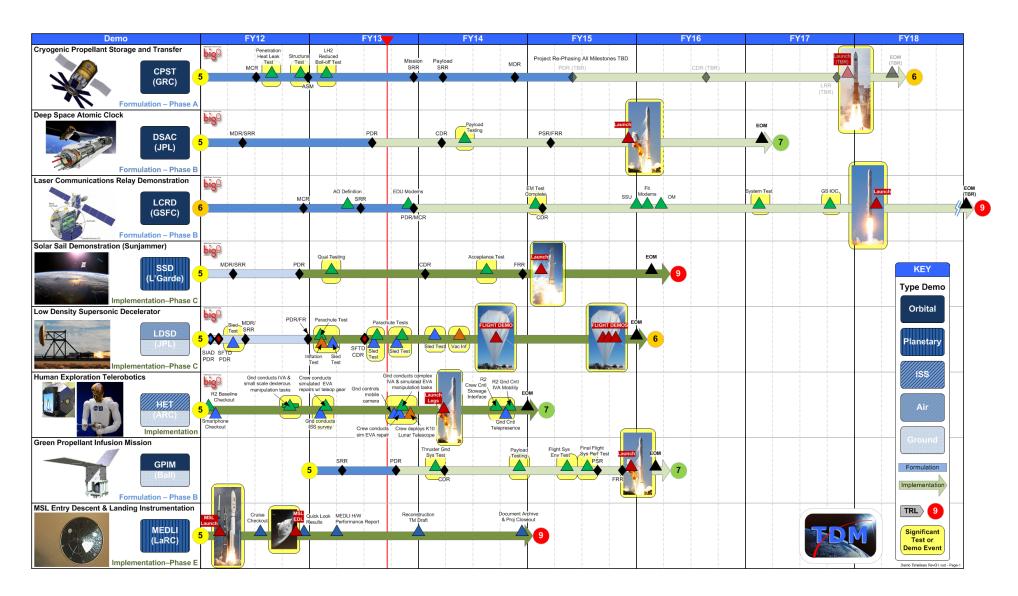






Major Milestones







FY2014 Big Nine



Human

Missions

Science

Missions



Laser Communications





Cryogenic Propellant Storage & Transfer



Deep Space Atomic Clock





Large-Scale Solar Sail



TDM

Low Density Supersonic Decelerators



Increases space-based broadband, delivering data rates 10-to-100 times faster than today's systems, addressing Ammunummunummunummun the demands of future missions.

Better fuel handling technology will improve spacecraft fuel economy. Required for Cryogenic Propulsion Stage (Space Launch System - SLS - upper-stage).

This tiny atomic clock is 10-times more accurate than today's ground-based navigation systems, enabling precise, in-space navigation.

TDM

Green Propellants

This solar sail has an area 7 times larger than ever flown in space, enabling propellant free propulsion and next generation space weather systems.

> Demonstrates new parachutes and inflatable braking systems at supersonic velocities enabling precise landing of large payloads on planetary surfaces.

9own Develops and demonstrating green thus provides an alternative to highly corrosive and toxic hydrazine; consequently expanding the capabilities of small spacecraft systems.

Zannununununun capable of remotely operating robots to assist in future exploration; maturing new robots capable of assisting humans in routine and tedious work.

Ammunumin -Develops large-scale solar array panels and deployment mechanisms. Critical step on the development path to a high-power solar electric propulsion system.

Demonstrating large composite, light weight fuel tanks that can reduce the mass and cost of the next generation SLS.



Solar Electric Propulsion



Composite Cryotank











Robonaut Roadmap

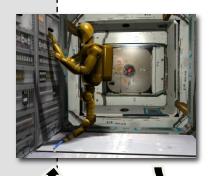
JSC Laboratory



ISS Laboratory



ISS IVA



ISS EVA



Develop dexterous robot suitable for human tasks

Explicitly human safe motions

Fixed base and mobile robot prototypes

Verify 0g operations

Characterize robot performance parameters in LEO

Develop capabilities to perform useful IVA work

Fixed base robot

Assist IVA crew

Routine maintenance

Interior cleaning

Equipment calibration

Remote science

Mobile robot

Improve EVA efficiency through worksite prep & tear-down / stow

Expand EVA capabilities while reducing risks to the crew

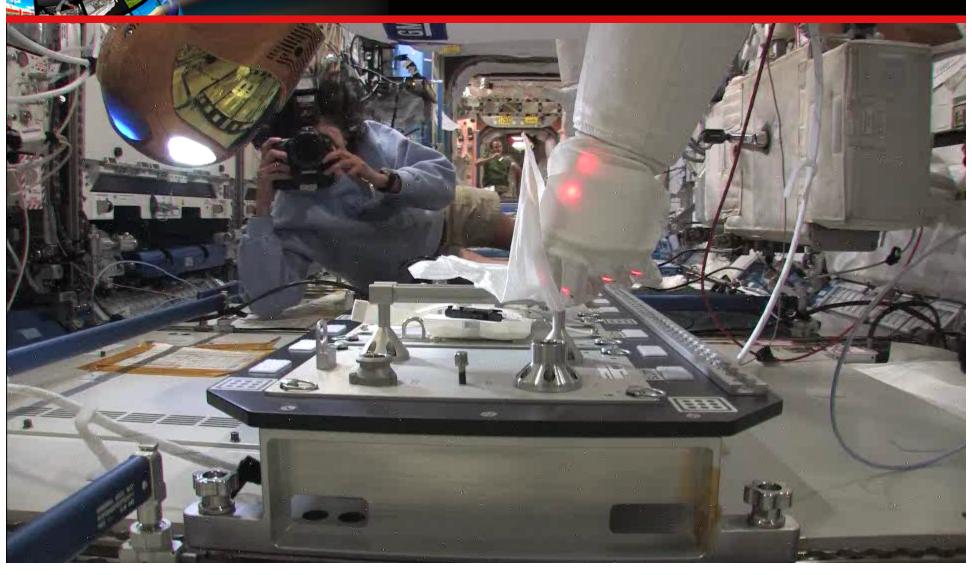
HRS, RSAA

HET

ISS Program



Robonaut 2







ARC Laboratory



ISS Laboratory



ISS IVA



ISS EVA



Enable SPHERES to be used as a teleoperated robot

Upgrade existing SPHERES platform

Telerobotic freeflyer prototypes : Demonstrate IVA & EVA tasks using "Smart SPHERES"

Demonstrate ground and crew control with "Smart SPHERES" Video surveys

RFID inventory

Dosimetry readings

Mobile ground support (camera, procedure prompt, etc.) for crew Routine inspections

Emergency inspections

HRS, DARPA HET

ISS Program



Smart SPHERES

