

OUTLINE

- ♦ Small Satellite Launch Market Projections
- ♦ NASA Small Spacecraft Technology Program Development and Demonstration
- Examples of Other NASA Activities in Small Spacecraft
 Science
 Human Exploration and Operations
- Examples of ESA Small Spacecraft Technology Development and Demonstration
- ◆ Questions and Discussion

"Small Satellites Market Pull"

- NASA State of the Art Report
- SMD-STMD study of how small spacecraft can address Decadal Survey mission needs – starting in Aug-Sept 2014
- Informal assessments of market pull
 - Annual Government Cubesat Forum (NASA, USAF, NRO, NRL, DARPA, ORS, etc.)
 - Smallsat and Cubesat conferences
- Identified civil and commercial markets:
 - Earth remote sensing
 - Space weather and heliophysics
 - Communications
 - NEA exploration (i.e., ARM secondary payloads)
 - In-space subscale testing of technology
 - Education and public outreach
- Launch demand market studies (rideshare, ISS deploy, small launchers)

"Small Satellites Market Pull"



Small can also mean: Affordable, Rapid, Transformative

Market studies of disruptive technologies

Role of STMD versus SMD & HEOMD in push versus pull technology development

REFERENCE:

2014 Nano/Microsatellite Market Assessment

Developed by:

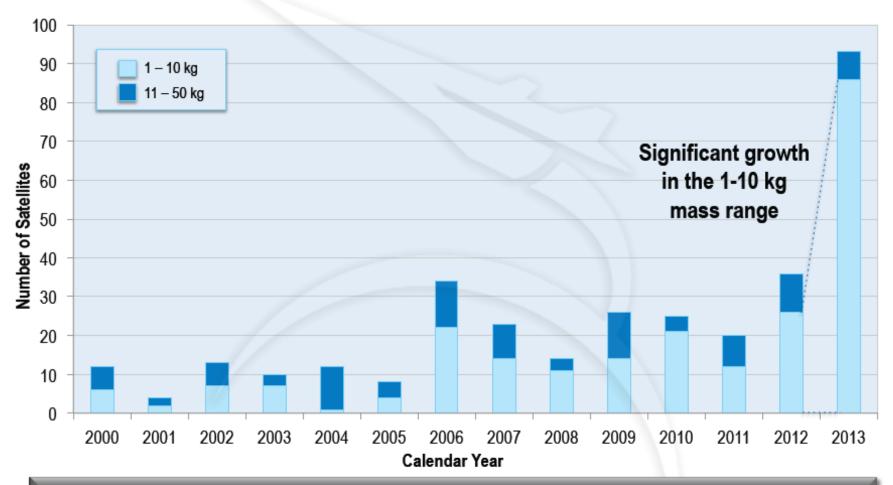
Ms. Elizabeth Buchen

Mr. Dominic DePasquale

Published by:

SpaceWorks Enterprises, Inc. Atlanta, GA

Historical Nano/Microsatellites Launched: 2000 - 2013 (1 - 50 kg)



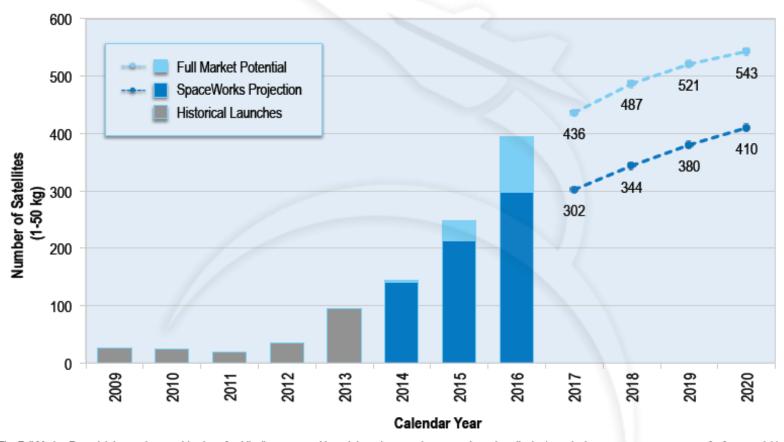
A 330% increase in attempted nanosatellite (1-10 kg) deliveries in 2013, compared to 2012, shows signs of an emerging and growing launch market





Nano/Microsatellite Launch History and Projection (1 - 50 kg)

Projections based on announced and future plans of developers and programs indicate between 2,000 and 2,750 nano/microsatellites will require a launch from 2014 through 2020

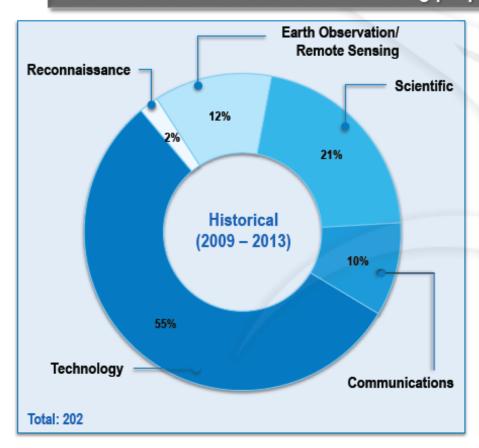


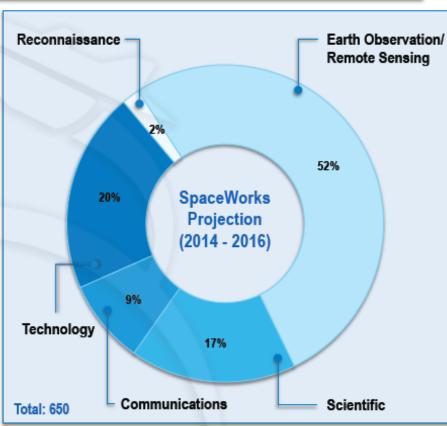
The Full Market Potential dataset is a combination of publically announced launch intentions, market research, and qualitative/quantitative assessments to account for future activities and programs. The SpaceWorks Projection dataset reflects SpaceWorks' expert value judgment on the likely market outcome.



Nano/Microsatellite Trends by Purpose (1 - 50 kg)

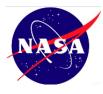
Over half of future nano/microsatellites will be used for Earth observation and remote sensing purposes (compared to 12% from 2009 to 2013)





A smaller proportion of technology development/demonstration nano/microsatellites will be built in the next few years (20% vs. 55% from 2009 to 2013)







Affordable, Rapid, & Transformative Development and Demonstration

Objectives:

- To develop and demonstrate new small spacecraft technologies and capabilities for NASA's missions in science, exploration and space operations
- To promote the small spacecraft approach as a paradigm shift for NASA and the larger space community.

Focused Technology Development Projects in:

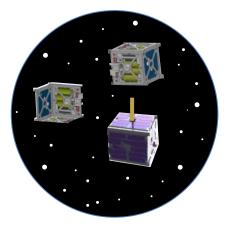
Communications, Avionics, Propulsion, Power, Instruments, Manufacturing, Small Earth Return Vehicle

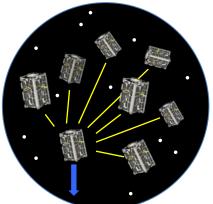
Flight Demonstration Projects in:

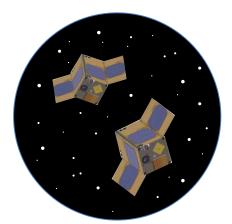
Radio and Laser Communications
Formation Flight and Docking
Low cost satellite buses
Smallsat swarms for space science missions

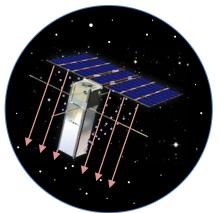
Implemented through:

Directed NASA projects
Contracts with private industry
University-NASA partnerships
Collaboration with SBIR and other programs



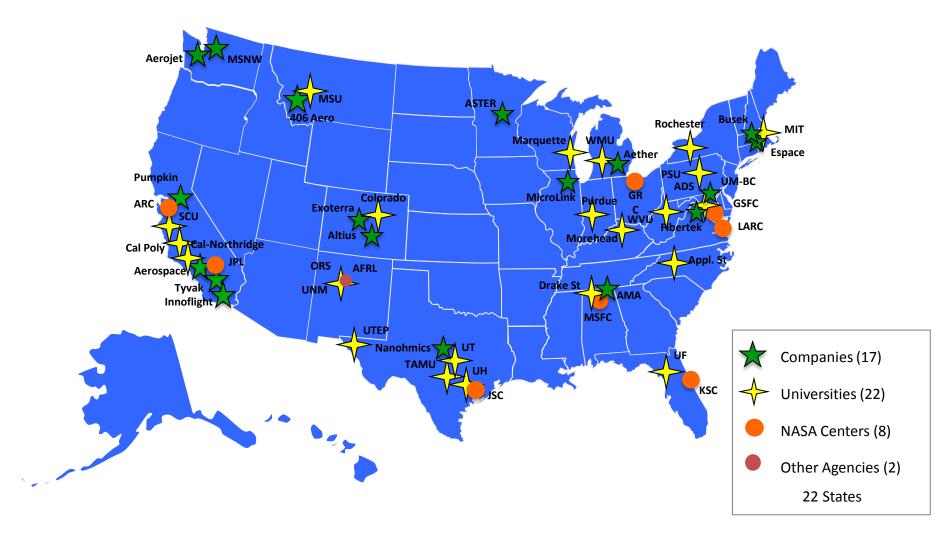






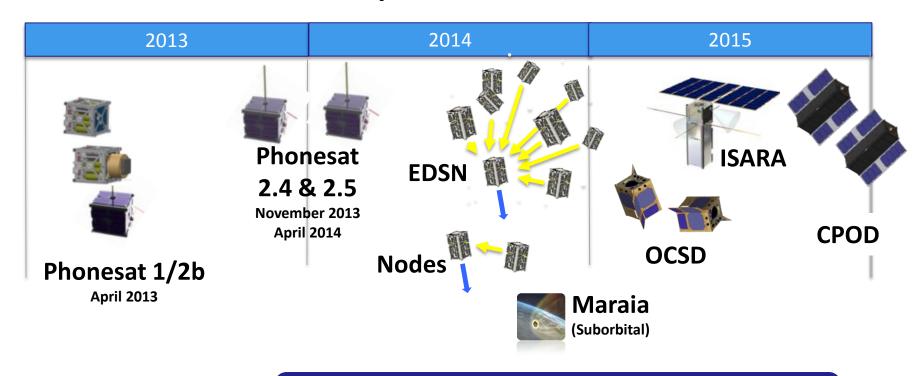


Nationwide Participants and Partners



NASA

Projects: 2013-2015



Smallsat Technology Partnerships
13 projects with NASA-University collaboration

5 Propulsion Technology Projects

EDSN: Edison Demonstration of Smallsat Networks **ISARA**: Integrated Solar Array and Reflectarray Antenna **OCSD**: Optical Communications and Sensor Demonstration **CPOD**: Cubesat Proximity Operations Demonstration

SBIR - Deep Space Cubesats

Small Spacecraft Technology "Pathways" Phonesat/Edison Follow-on Demos Precise formation flight with propulsion **EDSN** ISARA comm & radar **Phonesats** remote sensing demos **EDSN Nodes** Laser Communication OCSD follow-on "Inspector" Propulsion capability **Development Projects CPOD** 2014 SBIR Phase 1 2014 SBIR Phase 2 **6U Bus & MicroSEP Deep Space Cubesats** Deep Space Cubesatsx development Deep Space & 2015 SBIR Phase 1 2015 SBIR Phase 2 large ∆V propulsion Topic TBD Topic TBD capability

Maraia suborbital test



Suborbital and orbital demos

Earth Return Payload & EDL testbed



STMD Involvement in Small Spacecraft



Small Spacecraft Technology

- Focused technology development
- Flight demonstration projects

Game Changing Development

- General cross-cutting technology development
- Miniaturized Electrospray Propulsion development projects

SBIR/STTR

- Existing subtopics for small spacecraft technology
- 2014 Select topic: Deep Space Cubesat Technology

Flight Opportunities

- Technology payload development and test opportunities (FY13 NRA, etc.) Suborbital
- Potential orbital flight opportunites

Centennial Challenges

- Prize competitions in formulation

Technology Demonstration Missions

- ESPA-class spacecraft flight demonstrations

NIAC Space Technology Research Grants Center Innovation Fund

General cross-cutting concept & technology development and some small spacecraft projects



Science

Earth Science Smallsat Activities



2013	2	014	2015	2016
M-Cubed2	IPEX	GRIFEX (notional)		
M-Cubed2 (Michigan Multipurpose	IPEX (Intelligent Payload Flight Experiment)	GRIFEX (GEOCAPE RO Flight Experim	IC In-	CYGNSS (EV-2)
MiniSatellite) Validation of on-board new radiation-hard-	Validation of-high data rate cameras (Gbps), and autonomy	Verify performa fast imaging de readout Integra	tector	Cyclone Global Navigation Satellite System 8 LEO nanosats using GPS

Circuit (ROIC) with in-

pixel A-D converter

Led by Jet Propulsion

Task Cost: \$1.9M

Launch: NET 2014

Lab

new radiation-hardby-design FPGA

Led by Jet Propulsion Lab

Task Cost: \$1M

Launch: Fall 2013 on

NRO-L39

algorithms to deliver high value, low-latency products with >20x reduction in instrument raw data rate

Led by Jet Propulsion Lab

Task Cost: \$1M

Launch: Fall 2013 on

NRO-L39

8 LEO nanosats using GPS reflections to monitor ocean surface roughness

Led by University of Michigan

Lifecycle Cost: \$151.7M

Launch: Early CY 2016

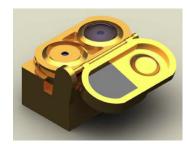
InVEST: In-Space Technology Validation for Earth Science



The InVEST solicitation sought small instruments and instrument subsystems that will advance technology to enable relevant Earth science measurements and targeted the cubesat platform. The first four awards were selected in April 2013.

The *Microwave Radiometer Technology Acceleration (MiRaTA)* cubesat will validate multiple subsystem technologies and demonstrate new sensing modalities that could dramatically enhance the capabilities of future weather and climate sensing architectures - MIT Lincoln Laboratory





The *Radiometer Assessment Using Vertically Aligned Nanotubes (RAVAN)* project will demonstrate a bolometer radiometer that is compact, low cost, and absolutely accurate to NIST traceable standards. RAVAN could lead to affordable cubesat constellations that, in sufficient numbers, might measure Earth's radiative diurnal cycle and absolute energy imbalance to climate accuracies (globally at 0.3 W/m2) for the first time.

- Johns Hopkins Applied Physics Laboratory

The objective of the *Cubesat Flight Demonstration of a Photon Counting Infrared Detector (LMPC)* is to demonstrate in space, a new detector with high quantum efficiency and single photon level response at several important remote sensing wavelength detection bands from 0.9 to 4.0 microns. - The Aerospace Corporation





The *HyperAngular Rainbow Polarimeter (HARP)* will validate a technology required by the Aerosol-Cloud-Ecosystem (ACE) mission concept and prove the capabilities of a highly-accurate, wide-FOV, hyperangle, imaging polarimeter for characterizing aerosol and cloud properties.

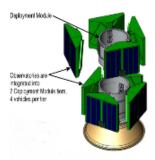
- University of Maryland, Baltimore County

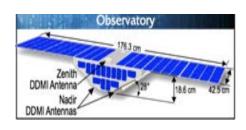
Earth Science Flight Program Opportunities

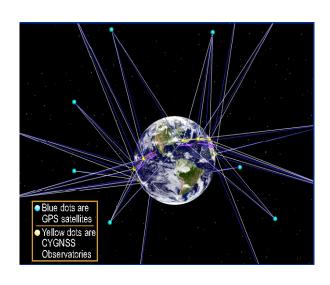


Earth Venture - 2 (EV-M): Complete, self-contained, small missions

- Small sat or stand alone payload for Mission of Opportunity
- > \$150 M total development cost
- Solicited in FY11 and every 4 years
- Class D (5y development)
- EV-2 was first Venture full spacecraft mission call
 - ➤ AO released June 17, 2011
 - Cyclone Global Navigation Satellite System (CYGNSS) selected July 2012
 - > 8 microsats (18 kg ea.) for GPS reflectometry to measure ocean surface winds
 - ➤ LEO; 35° inclination, 500 km altitude
 - NASA provided launch vehicle; sized for a Pegasus







EV Schedule	Туре	Solicitation	Selection	Launch/Delivery
EV-2	Full Orbital	2011	2012	LRD ~2016
EV-4	Full Orbital	2015	2016	LRD ~2021

Heliophysics Involvement in Small Spacecraft



ROSES Solicitations

- An open solicitation to propose cubesats for Heliophysics research has been included in our ROSES NRAs
 - ROSES-12: LCAS solicitation: CeREs selected

CeREs: A Compact Radiation bElt Explorer to study charged Particle dynamics in Geospace.

- Lead: NASA Goddard Space Flight Center
- Partner Organization: Southwest Research Institute
- Science: Study charged particle dynamics; Radiation Belt losses
- Implementation: A 3- year low earth orbiting cubesat, carrying a Compact High Resolution Electron Ion Telescope (CoHeREnT)

Instrument Development

- Annual ROSES solicitations
- Instrument development supporting Heliophysics science (not restricted to small spacecraft)
- Proposals solicited annually
- Approximately 1-4 awards per year
- Several selected for development of compact instrumentation

Previous Heliophysics Small Spacecraft Missions

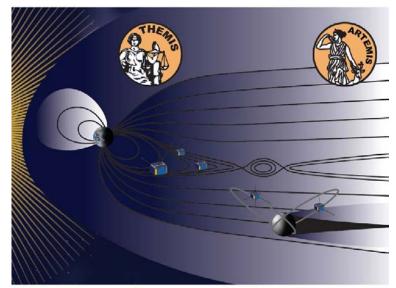


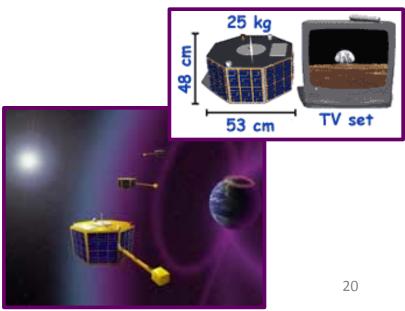
THEMIS: Time History of Events and Macroscale Interactions during Substorms

- Science Goal: to determine what physical process in near-Earth space initiates the violent eruptions of the aurora that occur during substorms in Earth's magnetosphere.
- Midex Explorer Mission: Selected March 2003;
 Launched February 17, 2007
- Constellation of 5 identical spacecraft
- 126 kg each (with propulsion)
- Completed Prime mission; three THEMIS probes operating as an extended mission; two of the 5 s/c repurposed into lunar orbit are operating as the ARTEMIS mission; Acceleration, Reconnection, Turbulence and Electrodynamics of the Moon's Interaction with the Sun"

Space Technology 5 (ST5)

- New Millennium Program
- Technology demonstrations and research on the Earth's magnetosphere
- Constellation of 3 identical spacecraft
- 25 kg each (with propulsion)
- Launched March 22, 2006, completed June 20, 2006





Planetary Science Involvement in Small Spacecraft



SMD Astrobiology Small Payloads Program

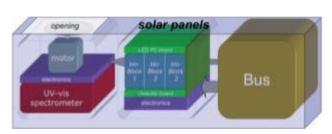
O/OREOS Nanosat Organism/Organic Exposure to Orbital Stresses

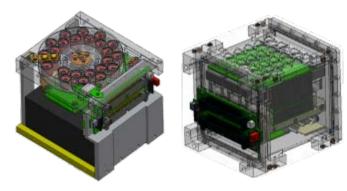
Goal: Astrobiology -- viability of microorganisms and astrobiologically relevant organics over 6-month space exposure.

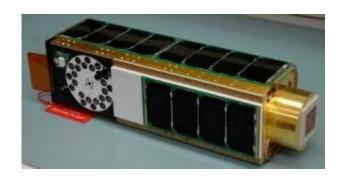
Technology:

- 10x10x30 cm Nanosat Bus (3U)
- UV-Vis Spectrometer (1 cube)
- Biology growth-&-analysis system (1 cube)

Success: Launched Nov. 19, 2010. Full mission success, both payloads, May 23, 2011. Funding, concept to orbital data analysis: \$2.5 million. 4 refereed papers published, 3 in preparation.







Planetary Science Involvement in Small Spacecraft



INSPIRE: Interplanetary NanoSpacecraft Pathfinder In a Relevant Environment

PI: Andrew Klesh, PM: Lauren Halatek, Jet Propulsion Laboratory

Objectives:

- Demonstrate key nano-spacecraft abilities: telecommunications, navigation, command & data handling, relay communications, and deep-space reliability/ fault protection
- Demonstrate science utility with compact payloads (Vector Helium Magnetometer & imager)

 Demonstrate new technologies for low-cost COTS / university components

Overview:

Volume: 3U (10x10x30cm)

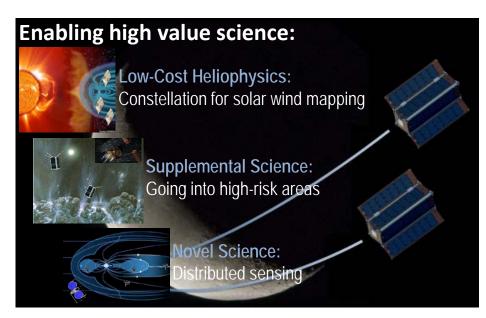
Mass: 4.8 kg

Power Generation: 20 W

• **Data Rate:** 62-64000 bps

ACS: Cold gas (3D printed),
 0.02 deg knowledge, 1-2 deg control

- Radio: DSN Compat X-band/2-way Nav, UHF xlink
- Payloads: Magnetometer, Agile Sci Algorithm, cam



Status:

- Selected for launch by NASA's Cubesat Launch Initiative (2/26) – awaiting manifest
- Schedule: Ready for launch Fall 2014
- Mission Duration: 1-3 months, 1 kbps at max distance
 1.5x10E6 km (DSN 34m dish)

Planetary Science Involvement in Small Spacecraft

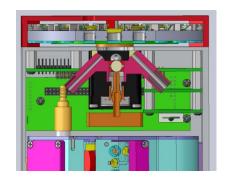


In Development

RASIR: Reactivity Analyzer for Soil, Ices, & Regolith
A 1U Optical Sensor Array to Measure Extraterrestrial Soil Chemistry

Primary Science Objectives

- Measure chemical processes that alter or remove organics from the environment of interest:
- Correlate organic inventory measured by other payload instruments to soil reactivity levels

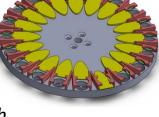


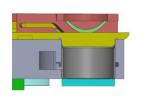
Scientific Lineage: MOx, MOI, MAOS, SEVO

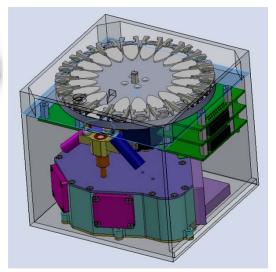
Flight Heritage: O/OREOS SEVO

<u>Technical Approach:</u> Time sequence of **UV-Vis** or fluorescence spectra of **sensing films** in bottom of each sample well

200 – 1000 nm, 1 – 2 nm resolution







Small Instruments for Planetary Science



ROSES Solicitations

Opportunities for the community to propose small instruments for Planetary research can be found in our Research Opportunities in Space and Earth Science (ROSES) NRAs. Science payloads on most future planetary science spacecraft will be limited to small, low-mass, and low power, consumption instruments. Some of these instruments will make excellent candidates for small sat missions.

Instrument Development Programs

 PICASSO - Conducts planetary and astrobiology science instrument feasibility studies, concept formulation, proof of concept instruments, and advanced component technology development (TRL 1-4)

Proposals received annually Approx.~ 10- 12 awards

 MatiSSE- Develops and demonstrates planetary and astrobiology science instruments to the point where they may be proposed in response to an AO (TRL 3-6)

Proposals received bi-annually

Approx. ~ 3-6 awards

Astrophysics Small Spacecraft Missions



Explorer Program:

- The Explorers Program has provided flight opportunities for dozens of astrophysical missions. Seven have been 200 kg or less in mass.
- Most Astrophysics Explorers under 200kg date from the 1960's and 1970's.

Why small satellites have not played a larger role in Astrophysics

- Small instruments typically cannot provide enough photons or enough angular resolution to exceed the capabilities of ground-based instruments.
- In gamma ray astronomy a large mass is required to stop high energy photons.

As technology advances, smaller spacecraft will become more viable.

- 2 cubesat proposals received in 2011 ROSES cycle
- 8 cubesat proposals received in 2012 ROSES cycle.
- The Explorers Program will continue to be an option for funding small spacecraft.



Human Exploration and Operations

HEOMD: Cubesat Launch Initiative

NASA

- Launch opportunity available for cubesats from NASA, other government agencies, universities, other educational institutions, and US non-profit organizations.
- The research should address aspects of science, exploration, technology development, education or operations.
- No funding provided for satellite development but integration and launch is provided at no cost.



- Provide meaningful aerospace and STEM education
- Advance the development of technologies
- Conduct science research

Implementation:

- Annual call for proposals
- Selection by panel including HEOMD, STMD, SMD, Education Office
- Manifesting and integration overseen by NASA LSP
- Referred to as ELaNa missions (Educational Launch of Nanosatellites)
- Flown as auxiliary payloads on previously planned NASA or other government missions.
- Cubesats are deployed using a standard Poly-Picosatellite Orbital Deployer (PPOD)



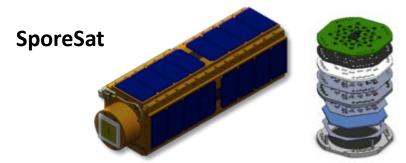
HEOMD: Space Life and Physical Sciences



Current Space Biology Projects in Cooperation with SMD Astrobiology

Small Complete Missions of Opportunity in Astrobiology and Fundamental Space Biology AO (SALMON 2008) – 2 selected projects SporeSat and EcAMSat.

Future solicitations via standard annual Space Biology NRA; plan to select one project per year for development.



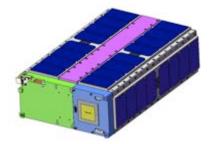
Objective:

To gain a deeper knowledge of the mechanism of cell gravity sensing by studying the activation of plant gravity sensing and electro-physical signaling in a single-cell model system (*Ceratopteris richardii*) using a "lab-on-a-chip" microsensor technology platform.

PI: Amani Salim, Purdue University Engineering: Ames Research Center

EcAMSat

Escherichia coli
Antimicrobial Satellite



Objective:

To determine how microgravity alters antibiotic resistance of uropathogenic *Escherichia coli* (UPEC), including the role of a critical resistance gene that indicates a marked increase in UPEC antibiotic resistance

PI: A.C. Matin, Stanford Engineering: Ames Research Center

ISS and SLS - Small Spacecraft Activity



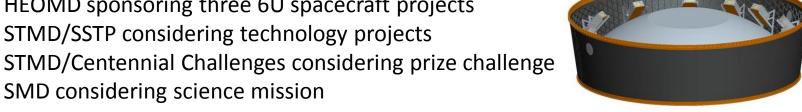
ISS Smallsat Deployment Opportunities

- Deployment of cubesats first began in 2013 via the JEM airlock
- Nanoracks now providing system for larger number of cubesat deployments
- Space Station Integrated Kinetic Launcher for Orbital Payload Systems (SSIKLOPS) will provide for deployment of larger smallsats



EM-1 and other SLS Secondary Payload Opportunities

- Planning for multiple 6U secondary payloads on EM-1 and subsequent flights – with deployment beyond LEO
- HEOMD sponsoring three 6U spacecraft projects
- STMD/Centennial Challenges considering prize challenge



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HEOMD Payloads under development for EM-1 Opportunity 6U Cubesats



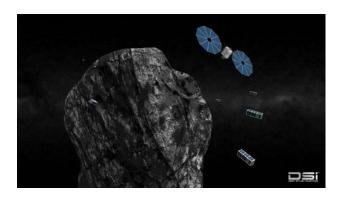
Payload NASA Centers	Strategic Knowledge Gaps Addressed	Mission Concept
BioSentinel ARC/JSC	Human health/performance in high-radiation space environments • Fundamental effects on biological systems of ionizing radiation in space environments	Study radiation-induced DNA damage of live organisms in cis-lunar space; correlate with measurements on ISS and Earth
Lunar Flashlight JPL/MSFC	 Lunar resource potential Quantity and distribution of water and other volatiles in lunar cold traps 	Locate ice deposits in the Moon's permanently shadowed craters
Near Earth Asteroid (NEA) Scout MSFC/JPL/LaRC/ JSC/GSFC	Human NEA mission target identification • NEA size, rotation state (rate/pole position) How to work on and interact with NEA surface • NEA surface mechanical/geotechnical properties	Flyby/rendezvous and characterize one or more NEAs that are candidates for a human mission

Asteroid Redirect Mission Partnerships for Secondary Payloads

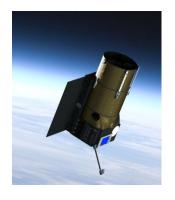




The Planetary Society: Transport extremophiles through deep space and return them to Earth to test panspermia theory.



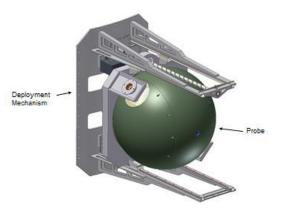
Deep Space Industries: Assessing low-cost commercial spacecraft for compatibility with the ARV and SLS to address Strategic Knowledge Gaps.



Planetary Resources: Adapting small low-cost commercial spacecraft to enhance ARM.



Applied Physics Lab: Hopper dropped on asteroid surface to measure elemental composition of asteroid regolith at multiple locations.



Honeybee Robotics: Multiple small kinetic impactors deployed from ARV to characterize regolith properties.

REFERENCE:

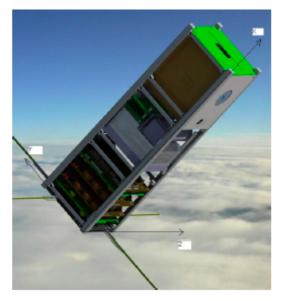
European Space Agency Cubesat Missions for In-Orbit Technology Demonstration

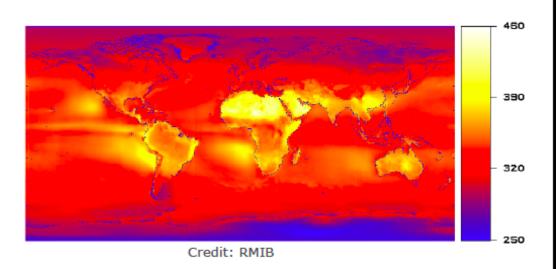
Roger Walker,
Directorate of Technology and Quality Management
ESA/ESTEC
29 May 2014

SIMBA IOD Mission



- Contractor: Royal Meteorological Institute and KU Leuven (Belgium)
- Sun-Earth radiometric science demonstrator:
 - Measure the Essential Climate Variables of Total Solar Irradiance, Earth Radiation Budget and Sun-Earth radiation imbalance
 - Payload: absolute cavity radiometer (RMI), 3-axis ADCS with star tracker (KUL)
 - Platform: 3U CubeSat (ISIS)
 - Heritage: Sova-P instrument on CNES Picard mission; Diarad on SOHO
 - Launch on QB50 flight in 2016 to 380x700 km altitude, 98° inclination





CubeSat IOD Missions | Roger Walker | ESA/ESTEC | 29/05/2013 | TEC | Slide 5

European Space Agency

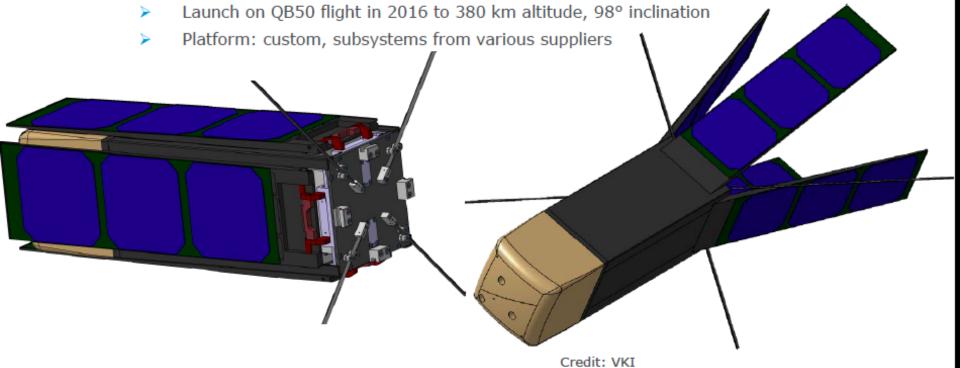
ESA UNCLASSIFIED - For Official Use

Status: PDR in June 2014

QARMAN IOD Mission



- Contractor: Von Karman Institute (Belgium)
- Atmospheric re-entry demonstrator
 - New heat shield materials -> ablation, plasma field measurements
 - Aerodynamic drag augmentation and passive attitude stabilisation system
 - > Telemetry data relay system during re-entry via the Iridium constellation



CubeSat IOD Missions | Roger Walker | ESA/ESTEC | 29/05/2013 | TEC | Slide 6

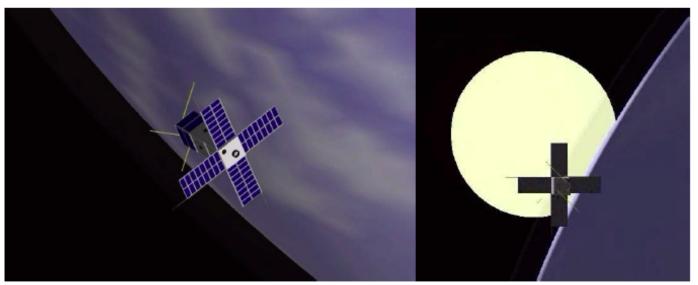
European Space Agency

Status: PDR in June 2014

PICASSO IOD Mission



- Belgian Institute of Space Aeronomy, Royal Observatory Belgium, VTT Finland
- Atmospheric chemistry science demonstrator
 - Stratospheric Ozone distribution -> limb sounding of solar disk with multi-spectral imager
 - Mesospheric Temperature profile -> multi-spectral imager (VTT)
 - Electron density in the ionosphere -> multi-Needle Langmuir Probe
 - Earth Radiation Budget -> micro-bollimetric oscillation system
 - Platform: 3U CubeSat (Clyde Space)
 - Launch on QB50 flight in 2016 to 380x700 km altitude, 98° inclination



Credit: BISA

CubeSat IOD Missions | Roger Walker | ESA/ESTEC | 29/05/2013 | TEC | Slide 7

European Space Agency

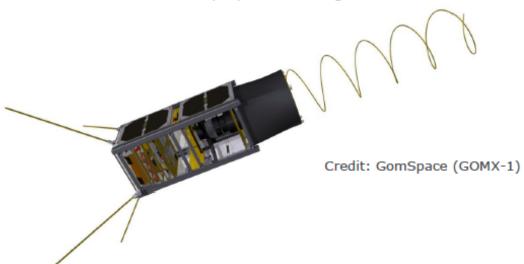
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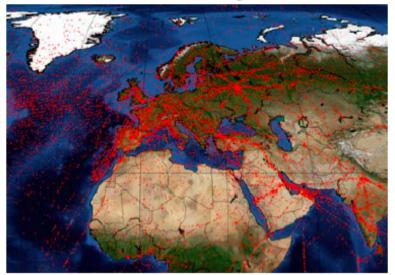
Status: PDR in November 2014

GOMX-3 IOD Mission



- Contractor: GomSpace (Denmark)
- 3U CubeSat telecommunications payload demonstrator
 - Improved Detection/de-coding of ADS-B signals broadcast by aircraft
 - Characterisation of Spot beams broadcast by GEO telecom satellites
 - Primary Payload: L-band Reconfigurable Software Defined Radio receiver
 - Additional payload: 3-axis ADCS with IR horizon sensors for nadir pointing accuracy of ~1° (day/night operation)
 - Launch to ISS via GomSpace/NanoRacks in July 2015
 - Deployment during Short Duration Mission of ESA astronaut Andreas Mogensen





European Space Agency

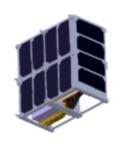
CubeSat IOD Missions | Roger Walker | ESA/ESTEC | 29/05/2013 | TEC | Slide 8

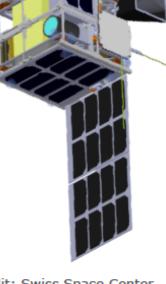
Status: CDR in July 2014

Active Debris Removal IOD Mission



- Contractor: Swiss Space Center, EPFL
- Objective:
 - IOD and operational validation of critical ADR technologies at <u>sub-scale</u> for future use on full-scale missions to remove large debris objects
- Mission concept #1:
 - Rendezvous sensors (Flash Lidar, VIS & IR cameras)
 - Motion reconstruction of uncooperative target
- Mission concept #2:
 - Net capture system dynamics & target interaction
 - Coupled two-body tether dynamics and control
- System concept:
 - 8U Chaser satellite + 4U Target satellite
 - Coupled together and launched in 12U deployment system
 - Low velocity mutual separation after LEOP
 - Close proximity ops with 6 DoF chaser around "passive" target with settable attitude rates





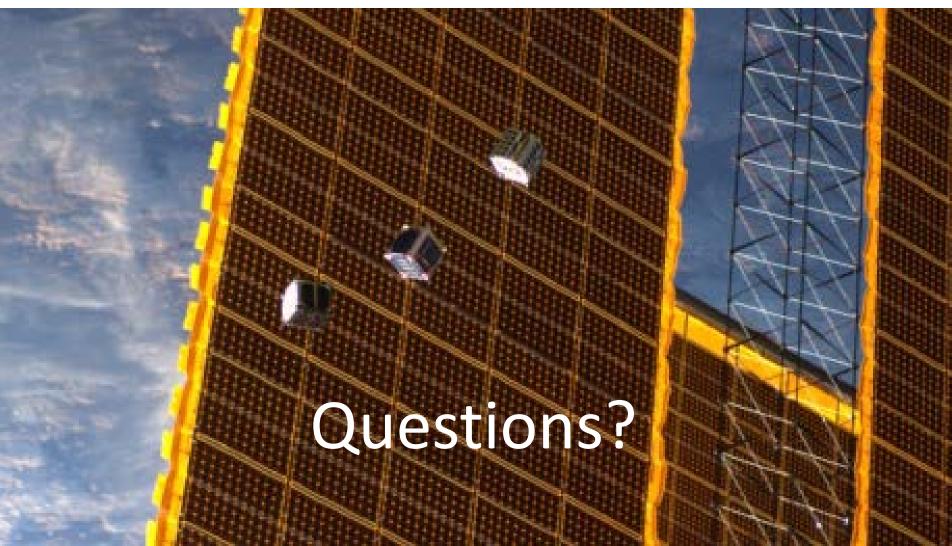
Credit: Swiss Space Center

European Space Agency

CubeSat IOD Missions | Roger Walker | ESA/ESTEC | 29/05/2013 | TEC | Slide 10

Small Spacecraft Technology



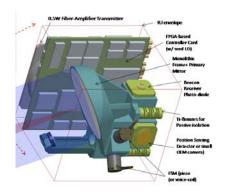


www.nasa.gov/smallsats

BACKUP

SBIR – Deep Space Cubesat Technology 2014 Phase 1 Projects





COMMUNICATIONS AND NAVIGATION

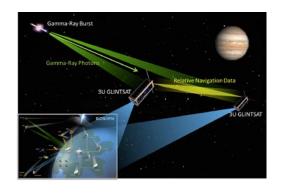
1U Cubesat Lasercom Terminal for Deep Space Communication

Fibertek, Inc. - Herndon, VA



Deep Space Cubesat Regenerative Ranging Transponder

Innoflight, Inc. - San Diego, CA



Deep Space Cubesat Gamma-ray Navigation Technology Demonstration

ASTER Labs, Inc. - Shoreview, MN

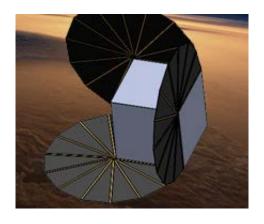
SBIR – Deep Space Cubesat Technology 2014 Phase 1 Projects



POWER GENERATION



High Power Betavoltaic Technology
MicroLink Devices, Inc. - Niles, IL



Deployable Solar Energy Generators for Deep Space Cubesats *Nanohmics, Inc. - Austin, TX*

SBIR – Deep Space Cubesat Technology

2014 Phase 1 Projects



PROPULSION



Cubesat Ambipolar Thruster for LEO and Deep Space Missions

Aether Industries, LLC - Ann Arbor, MI



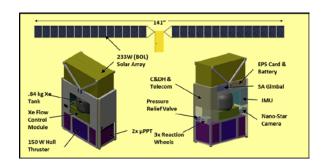
Multi-Purpose Interplanetary Deployable Aerocapture System

Altius Space Machines, Inc. - Louisville, CO

SBIR – Deep Space Cubesat Technology 2014 Phase 1 Projects



DEEP SPACE BUS



Solar Electric Propulsion Cubesat Bus for Deep Space Missions

ExoTerra Resource LLC - Lone Tree, CO



LunarCube for Deep Space Missions

Busek Company Inc. - Natick, MA